

NASA-CR-159675 19790025087 NASA CR-159675 R79-912997-39

# DEVELOPMENT OF SIAION MATERIALS

G.K. Layden

# UNITED TECHNOLOGIES RESEARCH CENTER

prepared for

# NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

NASA LEWIS RESEARCH CENTER

Contract NAS 3-19712

FINAL REPORT

September 1979

LIBRARY GOPY

901 ×3 19/9

LANGLEY RESEARCH CENTER LIBRARY, NASA HAMPTON, VIRGINIA

		•
		£
		•
		. :
		•
		٥

			* !	
1. Report No.	2. Government Acc	ession No.	3. Recipient's Catal	og No.
NASA CR-159675				
4. Title and Subtitle			5. Report Date	
Development of SiAlON Materials			September 1979	
personal of Bilitary interiors		•	6. Performing Organ	nization Code
7. Author(s)			8. Performing Organ	ization Report No.
G. K. Layden			R79-912997-39	•
			10. Work Unit No.	
9. Performing Organization Name and Address				
United Technologies Reserach Cen East Hartford, Connecticut 0610		•	11. Contract or Gran	t No.
Last Martiord, Connecticut 0010	•		NAS3-19712	
			13. Type of Report a	and Period Covered
12. Sponsoring Agency Name and Address			Contractor Rep	
National Aeronautics and Space A	dministration		June 1975, Sep	
Lewis Research Center			14. Sponsoring Agend	cy Code
Cleveland, Ohio 99135 15. Supplementary Notes				
Final Report	Matariala and St	matuma Diadada.		
Project Manager, T. P. Herbell, NASA Lewis Research Center, Clev		ructures Division		
16. Abstract				<del></del>
major phase was either Si <sub>3</sub> N <sub>4</sub> or were incorporated to promote liq sequently retained in boundaries properties of the bodies. Syste earth element) Zr-Si-Al-O-N, Y-S  Room temperature and 1370°C modu in terms of phase relationships	uid phase sinter between $Si_3N_4$ g ms investigated in Ee-O-N, and $R_1$ .	ing. Glass and/or crains which largely most extensively inc-R <sub>2</sub> -Si-O-N.  1370 C creep, and o	rystalline phase determined the p luded R-Si-Al-O- xidation behavio	were con- hysical N (R = rare r are discussed
17. Key Words (Suggested by Author(s))		18. Distribution Statement	t	
SiAlON, Silicon Nitride, Nitrogen Ceramics, Gas Turbine M	aterials	Unclassified - U	nlimited	
Sintering, Ceramic Processing	,	İ		
		7.	N79-3.	3258#
19. Security Classif. (of this report)	20. Security Classif. (	of this page)	21. No. of Pages	22. Price*
Unclassified	Unclassified		137	

			·
			,

#### TABLE OF CONTENTS

		<u>Page</u>
SECTION	I - SUMMARY	1
SECTION	II - INTRODUCTION	4
SECTION	III - EXPERIMENTAL PROCEDURES	7
A. B.	Sample Preparation	7 11
SECTION	IV - RESULTS AND DISCUSSION	13
A. B. C. D. E.	The System Zr-Si-A1-O-N  System Y-Si-A1-O-N  Properties of Bodies in the Y-Si-A1-O-N System  System Y-Si-Be-O-N  R-Si-O-N Systems	13 20 28 56 68
REFERENC	ES	90
APPENDIX	A - COMPOSITIONS INVESTIGATED IN THE YSIAION SYSTEM, AND TEST RESULTS	A-1
APPENDIX	B - BATCH COMPOSITIONS FOR VARIOUS TEST SAMPLES	B-1
APPENDIX	C - FABRICATION AND TEST DATA FOR VARIOUS TEST SAMPLES	C-1
APPENDIX	D - OXIDATION DATA FOR VARIOUS TEST SAMPLES	D-1
APPENDIX	E - SOLID-LIQUID EQUILIBRIA IN THE Y-Si-O-N AND Nd-Si-O-N SYSTEMS	E-1

#### LIST OF TABLES

		Page
TABLE 1 - ANALYSES OF RAW MATERIALS	•	8
TABLE 2 - OXIDE ADDITIVES	•	9
TABLE 3 - COMPATIBILITY TETRAHEDRA INVOLVING Si <sub>3-x</sub> Al <sub>x</sub> O <sub>x</sub> N <sub>4-x</sub> SOLID SOLUTIONS IN THE Zr-Si-Al-O-N SYSTEM	•	19
TABLE 4 - COMPARISON OF PROPOSED β PHASE COMPATIBILITY TETRAHEDRA IN THE Y-Si-Al-O-N SYSTEM	•	26
TABLE 5 - PROPOSED COMPATIBILITY TETRAHEDRA INVOLVING $si_{3-x}^{A1}x^{O}x^{N}4-x$ ( $\beta$ ' SOLID SOLUTIONS) IN THE SYSTEM Y-Si-A1-O-N	•	30
TABLE 6 - PHASE COMPOSITIONS OF TEST SAMPLES IN THE Y-Si-A1-O-N SYSTEM	•	31
TABLE 7 - SUMMARY OF PROPERTY DATA FOR Y-Si-Al-O-N SAMPLES	•	33
TABLE 8 - SOME EQUIVALENT FORMULATIONS, MOLECULAR WEIGHTS, AND DENSITIES FOR SOME OXIDE PHASES		39
TABLE 9 - CALCULATED VOLUME PERCENT PHASES IN 1400°C OXIDE SCALES OF SOME Y-Si-A1-O-N SAMPLES		40
TABLE 10 - PHASE OBSERVED AFTER HEAT TREATMENT OF COMPOSITIONS IN THE SYSTEM BeO-Y203-SiO2		. 59
TABLE 11 - X-RAY DIFFRACTION PATTERNS FOR Y <sub>2</sub> Be <sub>2</sub> SiO7		. 60
TABLE 12 - COMPARISON OF X-RAY DIFFRACTION DATA OF SOME Y-Si-Be-O-N PHASES	• (	. 61
TABLE 13 - PHASE COMPOSITIONS OF SAMPLES IN THE Y-Si-Be-O-N SYSTEM	• (	. 64
TABLE 14 - PARTIAL LIST OF COMPATIBILITY TETRAHEDRA IN THE Y-Si-Be-O-N SYSTEM		. 66
TABLE 15 - REACTIONS OF Si <sub>3</sub> N <sub>4</sub> with Y-Si-O-N and Nd-Si-O-N LIQUIDS.	•	. 71
TABLE 16 - FABRICATION AND PROPERTY DATA FOR SOME R-Si-O-N-BODIES.		. 73

# LIST OF TABLES (Cont'd)

	Page
TABLE 17 - COMPOSITIONS IN THE Y-Ce-Si-O-N SYSTEM	75
TABLE 18 - FABRICATION AND TEST DATA FOR Y-Ce-Si-O-N SAMPLES	76

#### LIST OF FIGURES

			Page
Figure	1 -	Representation of the Quinary System	14
Figure	2 -	System $Si_3N_4$ - $Si_3O_6$ - $Zr_3O_6$ - $Zr_3N_4$ After Rae et al	15
Figure	3 -	System Si <sub>3</sub> N <sub>4</sub> -AIN-Al <sub>2</sub> O <sub>3</sub> -SiO <sub>2</sub> After Jack	16
Figure	4 -	Probable Compatibility Relations (Partial) in the System $Si_3N_4-Zr_3N_4-Al_4N_4-Al_4O_6-Zr_3O_6-Si_3O_6$ at One Atmosphere Pressure of Nitrogen	18
Figure	5 -	${\rm System-Y_2O_3-Al_2O_3-SiO_2\ After\ Bondar\ and\ Galakov\ .\ .\ .\ .}$	21
Figure	6 -	System Si <sub>3</sub> N <sub>4</sub> -SiO <sub>2</sub> -Y <sub>2</sub> O <sub>3</sub> -YN (Composite)	23
Figure	7 -	The System Al <sub>2</sub> O <sub>3</sub> -Y <sub>2</sub> O <sub>3</sub> -YN-AlN	24
Figure	8 -	Phase Assemblages at Different Temperatures in the System $Y_2O_3$ -Al $_2O_3$ -Si $_3N_4$ After Rae et al (ref 20)	25
Figure	9 -	Model of the Y-Si-Al-O-N System	29
Figure	10	- Planar Projection of the $\beta^{\dagger}(0-60)^{-YAG-YS}$ 2 Tetrahedron	34
Figure	11	- 1400°C Isothermal Section of the $\mathrm{A1_{2}0_{3}}\text{-}\mathrm{Y_{2}0_{3}}\text{-}\mathrm{Si0_{2}}$ System	36
Figure	12	- 1400°C Oxidation Rate Constants (mg $^2$ cm $^{-4}$ hr $^{-1}$ ) for Samples in the Y-Si-Al-O-N System, Plotted on the 1400°C Isotherma Section of the Al $_2$ O $_3$ -Y $_2$ O $_3$ -SiO $_2$ Phase Diagram (Symbols	1
		Defined in Table 6)	37
Figure	13	- Oxide Scales on Samples 1078.1 and 1080.2	41
Figure	14	- Oxide Scales on Samples 1105 and 1111.1	42
Figure	15	- $1300^{\circ}$ C Oxidation Rate Constants (mg <sup>2</sup> cm <sup>-4</sup> hr <sup>-1</sup> ) for Sample in the Y-Si-Al-O-N Systems (Symbols Defined in Table 6) .	s . 44
Figure	16	- Polished Cross Section of Sample 1078.2 Oxidized at	. 46

# LIST OF FITURES (Cont'd)

	Page
Figure 17 - Mechanical Property Data vs. Composition for some YSiAlON Samples (Symbols Defined in Table 6)	47
Figure 18 -Fracture Initialed at Metallic Inclusion in $\beta$ ' 2 x $5^{\text{W}}/\text{Y}_2\text{O}_3$ Sample	48
Figure 19 - Fracture Initiated at Subsurface Void in $\beta$ ' 2 + 5 w/o $Y_2O_3$ Sample ( $\sigma$ = 510 MPa)	49
Figure 20 - Sample 1105.10 Fracture Surface ( $\sigma = 341 \text{ MPa}$ )	50
Figure 21 - Unetched and HF Etched Polished Sections of Sample 1105.1	51
Figure 22 - Relief Polished Section of Sample 1115.2	53
Figure 23 - SEM and EDAX Element Maps of Sample of Composition 1115 .	54
Figure 24 - Fracture Surface of Sample 1115.3 ( $\sigma = 327 \text{ MPa}$ )	55
Figure 25 - System Si <sub>3</sub> N <sub>4</sub> -SiO <sub>2</sub> -Be <sub>3</sub> N <sub>2</sub> -BeO at 1760°C, After Huseby et al (ref. 53)	58
Figure 26 - Solid Phase Compatibility Relations in the BeO-Y2O3-SiO2 System	63
Figure 27 - Model of the Y-Si-Be-O-N System	65
Figure 28 - Oxidation Rate Constants (mg <sup>2</sup> cm <sup>-4</sup> hr <sup>-1</sup> ) vs. Composition for Y-Si-Be-O-N Sample	67
Figure 29 - Phase Diagram for the System $Y_2SiO_5 - Y_2Si_3O_3N_4 \dots$	69
Figure 30 - Etched Sections of Samples Containing Si $_3$ N $_4$ and $^{Y}10^{Si}7^{O}23^{N}4$ in Different Proportions fired to 1750°C	72
Figure 31 - Density of Sintered Bodies of Compositions 94Si <sub>3</sub> N <sub>4</sub> 6z(y <sub>2-x</sub> Ce <sub>x</sub> Si <sub>2</sub> O <sub>7</sub> )	77

# LIST OF FIGURES (Cont'd)

<u></u>	age
Figure 32 - 1400°C Oxidation Curves for Compositions 94 m/o Si <sub>3</sub> N <sub>4</sub> + 6 m/o Y <sub>2-x</sub> Ce <sub>x</sub> Si <sub>2</sub> O <sub>7</sub>	79
Figure 33 - 1400°C Parabolic Oxidation Rate Constants for Compositions 94 m/o Si <sub>3</sub> N <sub>4</sub> 6/6 m/o $Y_{2-x}Ce_xSi_2O_7$	80
Figure 34 - Microscopy of 1400°C Oxidized Sample 1140:1	81
Figure 35 - Transversed Polished Section of 1400°C Oxidized Sample 1145:2	82
Figure 36 - Fracture Surface of Sample 1140:9	83
Figure 37 - 1370°C, 69 MPG Creep Curve for Sample 1140:5	85
Figure 38 - 1370°C, 69 MPa Creep Curve for Sample 1145:1	86
Figure 39 - Rim Around Creep Specimen 1145:2	87
Figure 40 - XRD Patterns for Some As fired and Heat Treated Y-Ce-Si-O-N Sample	88

#### R79-912997-39

#### Development of SiAlON Materials

#### SECTION I

#### SUMMARY

The objective of this program was the development of  $\rm Si_3N_4$  based ceramic bodies which can be fabricated to complex shapes by conventional forming and pressureless sintering techniques, and which exhibit properties which make them suitable for high stress applications in gas turbines at temperatures up to  $1370^{\circ}\text{C}$  (2500°F). Target goals were: flexural strength at 25°C of 520 MPa (75 ksi) and at  $1370^{\circ}\text{C}$  of 415 MPa (60 ksi); creep rate at  $1370^{\circ}\text{C}$  and 69 MPa (10 ksi) on the order of  $10^{-5}$  hr<sup>-1</sup>; static air oxidation parabolic rate constant at  $1400^{\circ}\text{C}$  on the order of 0.01 mg<sup>2</sup>cm<sup>-4</sup>hr<sup>-1</sup>.

Work completed during the first 21 months of this program was presented in detail in the interim report NASA CR-135290. The earlier work identified the systems Zr-Si-Al-O-N and Y-Si-Al-O-N, as being promising candidates in terms of oxidation resistance, and mechanical properties, respectively. However, no compositions were produced from either system which presented the desired balance of properties. In the current work, these two systems - particularly the latter were examined in greater depth. Solid phase equilibria were explored, and compatibility tetrahedra delineated. Property data for bodies from various tetrahedra were determined. In the case of the Zr-Si-Al-O-N system, it was concluded that sinterable  $\beta$ ' based ceramic formulations exist only on the "oxide" side of the  $\beta$  homogeniety line. The greatest extent of sinterable formulations lie in the  $\beta'_{10-60}$ -X-ZrN tetrahedron. Solidus temperatures appeared to be low in this tetrahedron, and samples had little strength or creep resistance at 1370°C. A composition on the  $\beta'_{0-10}$ -ZrO<sub>2</sub> plane (94.6 m/o Si<sub>2.7</sub>Al  $_3$ N<sub>3.7</sub> m/o 5.4 ZrO) also exhibited unacceptably high creep at 1370°C. If sinterable formulations with acceptably low creep exist in the Zr-Si-Al-O-N system, they must lie very close to pure Si<sub>3</sub>N<sub>4</sub>.

The Y-Si-Al-O-N system is more complex than the analogous Zr system. Five 3-phase, and eleven 4-phase compatibility tetrahedra exist wherein one phase is either  $\mathrm{Si}_3\mathrm{N}_4$  or  $\mathrm{Si}_3_{-\mathrm{X}}\mathrm{Al}_{-\mathrm{X}}\mathrm{O}_{\mathrm{X}}\mathrm{N}_{4-\mathrm{X}}$  solid solution. Property data for compositions from all of the 3-phase, and two of the 4-phase compatibility tetrahedra were screened for oxidation and mechanical properties. Oxidation properties at a given temperature are discussed by projecting body compositions (in cation equivalent percent) onto the relevant isothermal sections of the  $\mathrm{Y}_2\mathrm{O}_3$ -Al $_2\mathrm{O}_3$  -  $\mathrm{SiO}_2$  phase diagram. The only sinterable bodies with low oxidation rates at  $\mathrm{1400^{\circ}C}$  (on the order of 0.01  $\mathrm{mg}^2\mathrm{cm}^{-4}\mathrm{hr}^{-1}$ ) were compositions whose oxides fell in the primary

phase field of mullite, and which produced scales that were at least 30 volume percent mullite at  $1400^{\circ}\text{C}$ . These compositions fall in regions of the quinery system where the solidus temperature is low (perhaps below  $1500^{\circ}\text{C}$ ) so that  $1370^{\circ}\text{C}$  mechanical properties are poor. No sinterable composition was found that exhibited oxidation and mechanical properties that would be acceptable for high stress applications in oxidizing atmospheres at  $1370^{\circ}\text{C}$ . However, bodies were produced which exhibit properties attractive for stressed application at temperatures of  $1300^{\circ}\text{C}$  and below. One formulation (96.6 m/o Si<sub>2.614</sub>Al<sub>.386</sub>O<sub>.386</sub>N<sub>3.614</sub> + 2.3 m/o Y<sub>3</sub>Al<sub>5</sub>O<sub>12</sub> + 1.0 m/o Si<sub>3</sub>Al<sub>6</sub>N<sub>10</sub>) produced samples which exhibited the following properties:

Property	Value
1300°C oxidation rate	$0.003 \text{ mg}^2 \text{cm}^{-4} \text{hr}^{-1}$
25°C MOR (mean)	518 MPa
1370°C MOR (mean)	340 MPa
1370°C, 69 MPa creep	$6 \times 10^{-5} \text{ hr}^{-1}$

New systems explored during the final phases of the program were the Y-Si-Be-O-N system, and various R-Si-O-N and  $R_1$ - $R_2$ -Si-O-N systems where R represents a rare earth element. Only a limited number of compositions were studied in the Y-Si-Be-O-N system, but these were sufficient to delineate some of the compatibility tetrahedra in the quinery system. In general, oxidation rates for samples from this system were an order of magnitude lower than those for analogous compositions from the Y-Si-Al-O-N system. However, none of the compositions investigated exhibited acceptable creep properties.

Of the R-Si-O-N systems that were studied, dense bodies could not be produced from the yttrium system, probably because the viscous liquid which formed at sintering temperatures (1750°C) could not wet out the bodies. Theoretically, dense bodies could be sintered from the cerium, neodymium, and gadolinium-samarium analog systems, but samples from these systems exhibited excessive rates of oxidation.

Bodies were produced from the  $\rm Si_3N_4-Si0_2-Y_2\_xCe_x0_3$  system which exhibited acceptable oxidation (0.015  $\rm mg^2cm^{-4}hr^{-1}$ ) and 1370°C strength properties (400 MPa). XRD (x-ray diffraction) studies of the bodies disclosed them to be non-equilibrium assemblies with a minor phase constitution of  $\rm Y_2Si_2O_7$  and glass. The 1370°C creep tests of the material in argon atmosphere initiated the recrystallization of the minor phase assemblage to H phase and  $\rm Y_2Si_2O_7$  with a progressive decrease in 1370°C, 69 MPa creep rate to a value of  $\rm 5 \times 10^{-5} \, hr^{-1}$  at the time of rupture, 36 hrs into the test. The

grain boundary transformation, occurring under stress, is presumed to be responsible for the low stress rupture life of the sample.

It appears that the systems of the type  $\mathrm{Si_3N_4-Si0_2-Y_{2-x}R_x0_3}$  have a good potential for producing sinterable bodies for  $1370^{\circ}\mathrm{C}$  application, but development of heat treating cycles to yield equilibrated bodies with low internal stresses is required.

#### SECTION II

#### INTRODUCTION

The gas turbine is considered as a candidate for an alternative engine for automotive and marine use as well as for power generation. Efficient use of fuel can be achieved by high turbine operating temperatures and light rotating parts. These considerations, plus the necessity of using nonstrategic and potentially inexpensive materials, have led to serious consideration of the use of ceramic components. Stringent property requirements have limited the ceramic candidates for rotating parts to  $\text{Si}_3\text{N}_4$  and SiC based materials. While hot pressed  $\text{Si}_3\text{N}_4$  would appear to offer the most attractive properties at the present time, the great expense of machining parts from this material is prohibitive, and alternative materials are sough which can be formed to near net shape by conventional sintering process.

The object of this program is the development of a formulation or formulations for ceramic bodies which can be fabricated to complex shapes by conventional sintering techniques and which exhibit properties which make them suitable for highly stressed applications in a gas turbine. Target property goals for the program are: flexural strength at 25°C of 520 MPa (75 ksi), and at 1370°C of 415 MPa (60 ksi); creep rate at 1370°C and 69 MPa (10 ksi) on the order of  $10^{-5}$  hr<sup>-1</sup>, static air oxidation parabolic rate constant at  $1400^{\circ}$ C on the order of 0.01 mg<sup>2</sup>cm<sup>-4</sup>hr<sup>-1</sup>.

Materials which were investigated contained over about 90 volume percent  $\mathrm{Si}_3\mathrm{N}_4$  or solid solution having the  $\beta\mathrm{Si}_3\mathrm{N}_4$  structure. Such solid solutions are formed by substitution of Al and/or Be for Si, and 0 for N in the  $\mathrm{Si}_3\mathrm{N}_4$  structure (Refs. 1 through 10). Neither pure  $\mathrm{Si}_3\mathrm{N}_4$  bodies nor  $\beta$  solid solutions can be sintered to theoretical density (Refs. 3 and 6).  $\mathrm{Si}_3\mathrm{N}_4$  based bodies can be sintered by choosing compositions which contain some liquid at the firing temperature. This can be accomplished by using compositions in the  $\beta'$ -X phase field in the SiAlON systems (Refs. 11 and 12), or using "sintering aids" – usually additional oxide components which promote liquid formation (Refs. 13 through 20). In either case, the fired bodies contain phases other than  $\beta\mathrm{Si}_3\mathrm{N}_4$  or  $\beta'$  solid solution which to a great extent determine the properties of the bodies, particularly at elevated temperature. The task of identifying acceptable formulations then becomes one of extensive property testing.

Progress toward this goal during the first twenty one months of this program was described in an interim report (Ref. 21), the results of which are briefly summarized below.

Bodies investigated during the first twenty-one months of the program were limited to Si $_{3-x}$ Al 0 N $_{4-x}$  formulations to which were added different weight percentages of various "sintering aids", including AlPO $_4$ , GaPO $_4$ , ZrO $_2$ , ZrC, Cr $_2$ O $_3$ , TiO $_2$ , HfO $_2$ , Y $_2$ O $_3$ , La $_2$ O $_3$ , Er $_2$ O $_3$ , Nd $_2$ O $_3$  and GdSmO $_3$ . Uniformly dense bodies were obtained only with ZrO $_2$ , Y $_2$ O $_3$  (and mixtures of these), and other rare earth oxide additions. The best mechanical properties obtained up to this point were achieved with  $\beta$ ' compositions with low values of x (about 0.5 and below) with additions of Y $_2$ O $_3$ . Bodies of composition Si $_2$ .7Al 3O 3N $_3$ 7 + 2.5 w/o Y $_2$ O $_3$  exhibited room temperature flexural strengths up to about 550 MPa (80,000 psi). In the absence of other more deleterious types of flaws, room temperature strength of these bodies was controlled by the presence of metallic inclusions. Strength at 1370°C was grain boundary limited and reached about 450 MPa (65 ksi). The creep rate for a  $\beta$ ' Si $_2$ .45Al 55O 55N $_3$ .45 - 5 w/o Y $_2$ O $_3$  sample at 1370°C and 69 MPa stress was 6x10-5hr-1.  $\beta$ '-Y $_2$ O $_3$  samples, however, proved to have high oxidation rates at 1400°C (0.3 mg $^2$ cm $^{-4}$ hr-1), although oxidation was quite slow at 1300°C (0.003 mg $^2$ cm $^{-4}$ hr-1).

Bodies formulated as  $\beta'$  + ZrO $_2$  (and  $\beta'$  + ZrO $_2$ -Y $_2$ O $_3$  mixtures) were found to exhibit exceptionally low oxidation rates (virtually zero weight gain of 1400°C) and good room temperature strength. However, the creep rate at 1370°C and 69 MPa stress was on the order of  $4 \times 10^{-3} hr^{-1}$ , making these formulations unusable under high stress at that temperature.

A correlation was shown between the oxidation rate of the ceramic bodies and the minimum liquidus temperatures in the systems containing the oxidation products For compositions where the minimum ternary eutectic temperature in the system containing the oxidation products was above the test temperature (i.e., in the  $\rm SiO_2-AI_2O_3$ , and  $\rm SiO_2-ZrO_2-AI_2O_3$  systems) thin protective oxide scales developed, and the static oxidation rates were low. Bodies whose oxidation products fell in systems with minimum liquidus below test temperature (i.e.,  $\rm SiO_2-AI_2O_3-R_2O_3$  where R = rare earth), exhibited relatively rapid rates of oxidation.

During the earlier phases of this program, described in Ref. 21, it was realized that the formulations investigated ( $\beta'$  + foreign oxide) represented a limited sampling of the possible phase assemblages that could exist in the various systems considered. However, lack of phase equilibrium data for the relevant quinery systems (e.g., Y-Si-Al-O-N, Zr-Si-Al-O-N) was deemed an impediment to a reasonably structured approach to selecting formulations which would yield the desired properties. During the current phase of this program we have carried out compatibility studies in the two above named systems to serve as the framework for understanding and projecting property data. Test samples having compositions lying in various compatibility tetrahedra in the Y-Si-Al-O-N system were evaluated. Also, solid-liquid equilibrium in portions of the Si $_3N_4$ -Y20 $_3$ -Si0 $_2$  system were studied in an attempt to formulate sinterable bodies from this system. Although liquid was present at temperatures at least as low as 1700°C in parts of the system,

this was not effective in producing dense bodies by pressureless sintering. Attempts to find effective agents to promote densification relative to that in the  $\text{Si}_3\text{N}_4-\text{Y}_2\text{O}_3-\text{SiO}_2$  system, without degrading oxidation and/or creep properties too severely, led to the investigation of bodies in the systems Y-Si-Be-O-N, Ce-Si-Be-O-N and R-Y-Si-O-N.

In the report which follows, the chronology of the work reported is ignored in favor of organizing results in terms of systems. Much of the tabular fabrication and test data will be relegated to appendices wherein the chronological sample numbering systems is retained. Recent data on test bodies formulated in the Y-Si-Al-O-N have led to minor revisions of our interpretation of compatible relations in this system, originally reported in ninth quarterly report (Ref. 22). These include: 1) an expansion of the portion of the system investigated to include the higher AlN polytypes reported by Gauckler, et al. (Ref. 1), Jack (Ref. 7), and Land, et al. (Ref. 4), 2) assigning the appropriate homogeneity ranges to the phases X, 15R and 0'  $(Si_2N_2O_{ss})$  rather than treating these as point compounds as was done for simplicity in Ref. 22, and 3) adding one tie line and deleting another. These changes necessitated revision of the assumed phase compositions listed in several tables in earlier reports. This accounts for discrepancies that occur in a few instances between data as it appears here, and as it was originally presented in the quarterly reports. Also, some data from four  $\beta'-Y_2O_3$  samples originally presented in the interim report (Ref. 21) are recast in terms of the compatibility relationships as they are now understood. Similar revisions have been made in drawing the Zr-Si-Al-O-N diagram and in recasting old  $\beta$ '- $ZrO_2$  data in light of these relationships.

#### SECTION III

#### EXPERIMENTAL PROCEDURES

#### A. Sample Preparation

#### 1. Raw Materials

Several different Si $_3$ N $_4$  powders were used during the course of the program. These were: AME high purity 90 percent  $\alpha$  phase -300 mesh powder, KBI high purity 85 percent  $\alpha$  phase -325 mesh powder, Starck high alpha -  $1\mu$  powder, and GTE SN-402 amorphous Si $_3$ N $_4$ . Al $_2$ O $_3$  was Linde A micropolish, and AlN was Atlantic Equipment Engineers (AEE) AL106 99.9 percent -325 mesh. Spectrochemical analyses of impurities in these starting powders, and fast neutron activation analyses for oxygen in the Si $_3$ N $_4$  powders are presented in Table 1. SiO $_2$  powders used were -270 mesh silica glass, Apache 0.03  $\mu$  99.99 + purity, or AEE Si239 99.9 -325 mesh. Other compounds used in formulations are listed in Table 2. Chemical analyses were not performed on these compounds. Unless otherwise stated in subsequent text or tables, samples were prepared using the KBI Si $_3$ N $_4$  and AEE SiO $_2$ .

#### 2. Powder Preparation Techniques

## a) Powders for Compatibility Studies

Phase compatibility studies in the Y-Si-Al-O-N and Zr-Si-Al-O-N systems used the AME  $\mathrm{Si_3N_4}$  powder, Linde A, and -270 mesh silica glass. Initial studies in the  $\mathrm{BeO-Y_2O_3-SiO_2}$  system used  $\mathrm{Be_2SiO_4}$  as a constituent. This was prepared by calcining a mixture of AEE -325 mesh  $\mathrm{SiO_2}$  and beryllium carbonate. Later formulations used BeO obtained from the carbonate as a constituent. Mixtures were lightly ground under acetone in an alumina mortar to a smooth past-like consistency, then allowed to dry fully.

# b) Powders for Test Bar Fabrication

In general, powders for test bar fabrication were prepared in 30 to 100 gram batches. Constituents were weighed out into wide mouth 16 oz. polyethylene jars containing a known weight of either high alumina, or reaction sintered  $\mathrm{Si}_3\mathrm{N}_4$  grinding media, and methanol was then added to cover the charges. In some instances, a measured volume of carbowax-methanol solution was added to the ball mill charges. The jars were rolled at 75 rpm for a predetermined time (typically 18 hrs), then the charges were transferred to wash bottles and dried by spraying onto a heated aluminum plate. The wash bottles were kept agitated during the spraying process, and dried material was frequently scraped from the plate into a

<u>Re</u>	agent	<u>A1</u>	<u>Si</u>	<u>Ca</u>	<u>Co</u>	<u>Cr</u>	<u>Fe</u>	<u>Mg</u>	<u>Mo</u>	<u>Ni</u>	<u>Zn</u>	<u>0*</u>
	AME  KBI  GTE SN402  Starck  AlN	.15	major	< .02	< .05	< .01	.10	< .01	< .01	.02	.02	1.9
	KBI	.05	major	< .02	< .05	< .01	.30	< .01	< .01	< .01	< .02	0.8
Si <sub>3</sub> N <sub>4</sub>	GTE SN402	<.01	major	< .02	< .05	< .01	.01	< .01	< .01	< .01	< .02	2.2
	Starck	.18	major	.01	< .05	< .01	.01	< .01	< .01	< .01	< .02	1.1
	AlN	> 10	.20	<.02	.10	< .01	.07	.40	.02	.05	.02	
	A1 <sub>2</sub> 0 <sub>3</sub>			. <.01								

 $\infty$ 

\_

<sup>\*</sup> Determined by fast neutron activation analyses

# TABLE 2

# OXIDE ADDITIVES .

Alpo <sub>4</sub>	Rocky Mountain Research 99.99% Purity
$z_r^{0}$	Gallard Schlesinger 99.9% Purity
BeO _	Alfa Inorganics Beryllium Carbonate (Calcined)
Y <sub>2</sub> O <sub>3</sub> } GdSmO <sub>3</sub> }	Molycorp oxalates (calcined)
Er <sub>2</sub> 0 <sub>3</sub>	
Er <sub>2</sub> 0 <sub>3</sub> La <sub>2</sub> 0 <sub>3</sub> Nd <sub>2</sub> 0 <sub>3</sub>	Apache Chemicals Inc. 99.9% Purity
Nd <sub>2</sub> 0 <sub>3</sub>	
Sm <sub>2</sub> 0 <sub>3</sub>	American Potash Oxalate (calcined)

surrounding tray. The spray dried material was then passed through either a 20 or 100 mesh screen. The grinding media were washed, thoroughly dried, and weighed after the drying operation, and logs were kept of the media weight. Predictions of the media pick up by the batches were generally factored into the formulations. Exceptions to this practice will be noted where appropriate.

## 3. Cold Pressing of Green Samples

In general, samples for compatibility studies were pressed into 1.5 cm diameter pellets in a hardened steel piston die to a final pressure of 120 MPa. Test bars were pressed to dimensions approximately 0.4 cm x 1 cm x 15 cm in a steel die. These were then cut into three or four shorter bars. These were placed in individual latex bags which were then evacuated and tied off, and isostatically pressed to 275 MPa.

#### 4. Calcining

Samples pressed from batches containing carbowax binder were heated in air from room temperatures to  $600^{\circ}\text{C}$  on a three day linear heating schedule in order to burn out the binder prior to firing in nitrogen.

#### 5. Firing

Sample pellets for compatibility studies in the Y-Si-Al-O-N and Zr-Si-Al-O-N systems were loaded into covered boron nitride crucibles and placed in the graphite susceptor of the vacuum induction unit described in Ref. 21. The furnace chamber was evacuated to about 100 micron pressure, backfilled with nitrogen, reevacuated, and finally filled with nitrogen to one atmosphere of pressure. During the final filling, the heating was started and the furnace brought to the intended temperature in approximately 15 minutes. The temperature was monitored with a commercial W-Re sealed thermocouple. Thermocouples aged with use, and were occasionally calibrated against the melting point of wollastonite (CaSiO<sub>3</sub>, m.p. 1544°C). Unless otherwise stated, samples were maintained at the nominal firing temperature for 1 hour before cooling to room temperature.

Samples for compatibility studies in the  $BeO-Y_2O_3-SiO_2$  system were fired on platinum foil in air in a resistance heated furnace using a variety of heating schedules. Heat treatments will be described later.

In general, test bars were fired in an Astro model 1000-3060-FP-12 furnace equipped with L&N Electromax III controller, and Data Track model 5300 card programmer. The temperature profile was flat over the region of the furnace which contained the samples, and temperature was reproducible to within 5°C at about 1700°C. Samples were placed in covered boron nitride crucibles, inside graphite crucible holders which held the covers in place during evacuation. The furnace

chamber was evacuated and back filled to atmospheric pressure of nitrogen a total of three times, then a slow flow of nitrogen maintained in the furnace during firing. Firing schedules varied and will be presented later.

#### 6. Grinding and Polishing Test Bars

In general, fired bars were waxed to a metal plate and surface ground to a uniform thickness. Samples for mechanical test were finished on the tensile surface through various grits of SiC papers, and some were subsequently polished with Linde A micropolish. The edges of the tensile face were beveled at an angle of 45° for about 0.025 cm. The surface finish will be specified in subsequent tables as 240 grit, Line A, etc. Samples for oxidation testing were finished on all surfaces to the specified finish. Exceptions were samples containing beryllium, which were tested for oxidation in the as-fired condition, and were finished by hand to the specified finish for mechanical testing in order to avoid machine grinding.

#### B. Sample Evaluation

#### 1. Characterization of Fired Samples

Bulk density, specific gravity, and apparent porosity of sintered specimens were determined using procedures of ASTM test C373-5. Microstructures were examined on polished and etched sections of selected test specimens. Phase identification was made on the basis of XRD (X-ray defraction) and metallographic evidence.

### a. Mechanical Testing

Room temperature testing was performed in four point flexure using inner and outer spans of 0.95 cm and 1.90 cm, respectively. Cross head speed was 0.05 cm/min. Tests at  $1370^{\circ}$ C were performed in three point flexure using a span of 1.9 cm and a cross head speed of .05 cm/min. Test atmosphere was argon, and the sample supports and loading nose were tungsten. Fracture surfaces of selected test specimens were examined in the scanning electron microscope (SEM). Creep testing was done in three point flexure at  $1370^{\circ}$ C at a stress level of 69 MPa (10 ksi).

In some instances, test samples were heated in air at  $1370^{\circ}\mathrm{C}$  for extended periods of time prior to testing.

## b. Oxidation Testing

In general, samples were placed on platinum foil so fashioned that the samples were contacted along two lines only. Samples were introduced into a preheated oven

and removed periodically, cooled to room temperature, and the weight recorded. (In some instances, samples were placed across .05 cm dia platinum wire rails that were welded to an irridium-rhodium alloy sheet. This practice was discontinued when it was discovered that rhodium was being transferred to the samples.)

Following the oxidation tests, x-ray diffraction patterns were obtained from oxidized surfaces, and the samples were photographed. In most instances, the samples were then mounted in resin, polished to give a cross sectional view of the oxide scale, and examined metallographically. In selected instances, polished sections were examined by EDAX  $^{\rm R}$  (energy dispersive analysis of X-rays) to ascertain the distribution of elements in the oxide scale.

#### SECTION IV

#### RESULTS AND DISCUSSION

#### A. The System Zr-Si-Al-O-N

#### Background

The five component system was investigated by treating it as a six component reciprocal system, based on the end members  $\mathrm{Si}_3\mathrm{N}_4$ -AlN-" $\mathrm{Zr}_3\mathrm{N}_4$ " and  $\mathrm{SiO}_2$ -Al $_2\mathrm{O}_3$ - $\mathrm{ZrO}_2$ , making up an equilateral prism as shown in Fig. 1. The zirconium nitride " $\mathrm{Zr}_3\mathrm{N}_4$ " was chosen as an end member to make possible the plotting of this member on quadrilateral planes in any tetrahedron with Zr at an apex. The compound  $\mathrm{Zr}_3\mathrm{N}_4$  has been prepared by Juza, et al. (Ref. 23) and was stated to decompose above 700°C into 3ZrN+N.

Prior work on some of the binary subsystems includes that of Aramaki and Roy (Ref. 24) and Welch (Ref. 25) on the system  $Al_20_3$ -Si $0_2$ , Geller and Lang (Ref. 26) and Butterman and Foster (Ref. 27) on the system  $Zr0_2$ -Si $0_2$ , Lejus (Ref. 28) on the system AlN-Al203, Gilles (Ref. 29) and Claussen et al. (Ref. 30) on the system  $ZrN-Zr0_2$ . The ternary subsystem  $Al_20_3$ -Si $0_2$ -Zr $0_2$  has been investigated by Kordyuk and Gul'ko (Ref. 31) and Herold and Smothers (Ref. 32) and, more recently, by Sorrell and Sorrell (Ref. 33). The compatibility relationships drawn in Ref. 32 defining the compatibility triangle  $Al_20_3$ -3Al $_20_3$ .2Si $0_2$ -Zr $0_2$ .Si $0_2$  are at odds with those drawn in Ref. 31 which shows  $Zr0_2$ -Zr $0_2$ .Si $0_2$ -3Al $_20_3$ .Si $0_2$  coexisting. Reference 33 confirms the latter assemblage.

Rae et al. (Ref. 20) examined the system  $\mathrm{Si_3N_4}\mathrm{-Si0_2}\mathrm{-"Zr_3N_4}\mathrm{"-Zr0_2}$  and established a behavior diagram at  $1700^{\circ}\mathrm{C}$  shown in Fig. 2. They also reported one new phase in the Zr-Si-Al-O-N system. It was reported to be close to the composition  $5\mathrm{AlN.3Zr0_2}$  with a cubic unit cell approximately four times that of ZrN. They do do show ZrSiO<sub>4</sub> as a phase existing at this temperature.

The subsystem  $\mathrm{Si}_3\mathrm{N}_4\mathrm{-Si}_2\mathrm{-Al}_2\mathrm{O}_3\mathrm{-Al}\mathrm{N}$  has been investigated by Gauckler et al. (Ref. 5), Land et al. (Ref. 4), Jack (Ref. 7) and Layden (Ref. 6). Figure 3 is a reproduction of the behavior diagram for the system presented by Jack (Ref. 7). There is a basic agreement between the above referenced authors regarding the general aspects of equilibrium between the phases although the compositions assigned to specific new phases (e.g., X, 15R, etc) vary somewhat, as do the homogeniety ranges of solid solution phases.

#### Experimental Results

The phase 5 AlN.3Zr0 $_2$  reported by Rae (Ref. 20) was not confirmed. The only reaction products between components Al $_2$ 0 $_3$ , AlN, "Zr $_3$ N $_4$ " and Zr $_2$ 0 that were

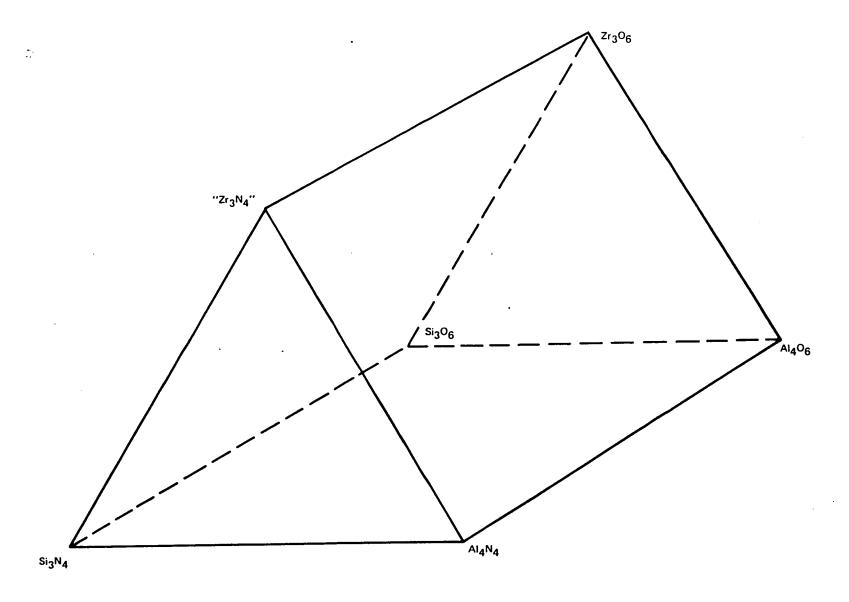


Figure 1. Representation of the Quinary System

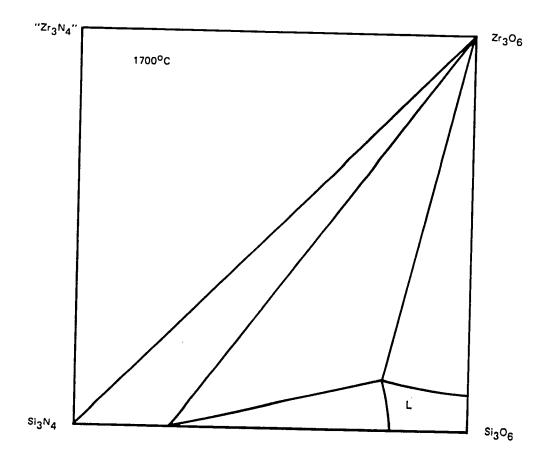


Figure 2. System Si<sub>3</sub>N<sub>4</sub>-Si<sub>3</sub>O<sub>6</sub>-Zr<sub>3</sub>O<sub>6</sub>-Zr<sub>3</sub>N<sub>4</sub> After Rae et al

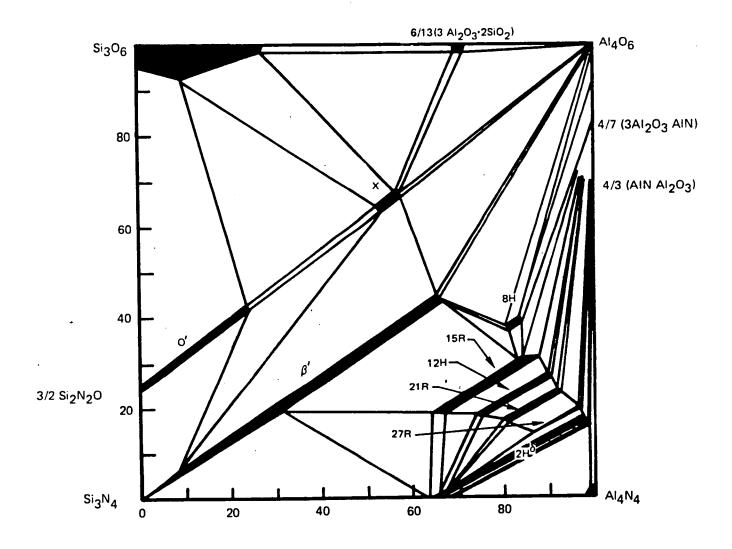


Figure 3. System Si<sub>3</sub>N<sub>4</sub>-AIN-AI<sub>2</sub>O<sub>3</sub>-SiO<sub>2</sub> After Jack

observed were the spinel phase solid solution centered around  ${\rm Al}_3{\rm O}_3{\rm N}$  and a cubic phase between ZrN and ZrO2. Compatibility was observed between "Zr3N4" (ZrN) and  ${\rm Al}_2{\rm O}_3$ ) and between ZrN and the spinel phase. Since the spinel phase is known to be unstable below about 1600°C (Refs. 28, 6) it may be inferred that the observed compatibility with ZrN was metastable at lower temperatures, and that  $ZrN-AlN-Al_20_3$  form a compatibility triangle below about  $1600^{\circ}C$ . Compatibility was established between ZrN and the ternary phases  $\beta$  , 15R and X, and between X and ZrO2. Topographical considerations then necessitated the compatibility of ZrN and the  $2{\tt H}^{\overline{\delta}}$  end member Si3Al6N10 also. On the basis of the  $Si_20-Al_20_3-Zr0_2$  diagram as drawn by Sorrell and Sorrell (Ref. 33), one can infer compatibility between  $Zr0_2.Si0_2$  and X phase, and between  $Zr0_2.Si0_2$ and  $Si_2N_2O$  as well. These joins, however, were not investigated experimentally. The above data, along with the relations shown on Fig. 3 establish the compatibility tetrahedra shown on Fig. 4. Relationships involving the AlN polytypes are not shown except those involving the 15R (treated here as a point, rather than a line compound) and the  $\rm Si_3Al_6N_{10}$   $\rm 2H^{\delta}$  end member. A list of the 3 and 4 phase compatibility tetrahedra shown in Fig. 4 which involve  $\beta$ ' solid solutions as one component is given in Table 3. (Several additional probable 4 phase tetrahedra involving the  $\beta_{60}\, end$  member, not shown on Fig. 4, but presumed to exist on the basis of Fig. 3 are added in parenthesis.)

It can be seen from Fig. 4 that  ${\rm Zr0}_2$  is in equilibrium with  ${\beta}'$  only over the range  ${\beta}_0$  to  ${\beta}_{10}$  (i.e., from 0 to 10 e/o A1). At higher aluminum concentrations in the  ${\beta}'$  solution, small additions of  ${\rm Zr0}_2$  move the composition into the  ${\beta}'$ -15R-ZrN tetrahedron. Of the two  ${\beta}'$ -Zr0 $_2$  compositions (ignoring the small Y203 substitution for Zr0 $_2$ ) previously tested in creep at 1370°C (Ref. 21 pp. 90-93), sample 757 formulated as Si2.45Al.550.55N3.45 + 5 w/o Zr0 $_2$ , falls in the  ${\beta}'$ -X-ZrN tetrahedron (phase composition 92.3 Si2.54Al.460.46N3.54 + 1.9 Si3Al6012N2 + 5.8 ZrN) whereas the formulation of sample 822, Si2.7 Al.30.3N3.7-5 w/o Zr0 $_2$ , correctly describes the equilibrium phase assemblage. Both compositions exhibited high creep rates at 1370°C (4 x 10<sup>-3</sup> and 1.7 x 10<sup>-3</sup>hr<sup>-1</sup> respectively) with the  ${\beta}'$ -Zr0 $_2$  composition (822) exhibiting a rate less than half of that of the  ${\beta}'$ -15R-ZrN composition. Both compositions sintered to near theoretical density at temperatures around 1750°C. During the current reporting period compositions in the  ${\beta}'$ 0-25-Si3Al6N10-ZrN and  ${\beta}'$ 25-60-15R-ZrN tetrahedron were prepared but the bodies did not sinter at 1750°C. In fact, they expanded on firing and exhibited large weight losses.

It can be concluded that sinterable  $\beta'$  based ceramic formulations occur only on the "oxide" side of the  $\beta'$  homogeneity line. The greatest extent of sinterable  $\beta'$  formulations lie in the  $\beta'_{10-60}$ -X-ZrN tetrahedron. Solidus temperatures appear to be low in this tetrahedron, and samples have little strength and creep resistance at 1370°C. At least some compositions lying on the

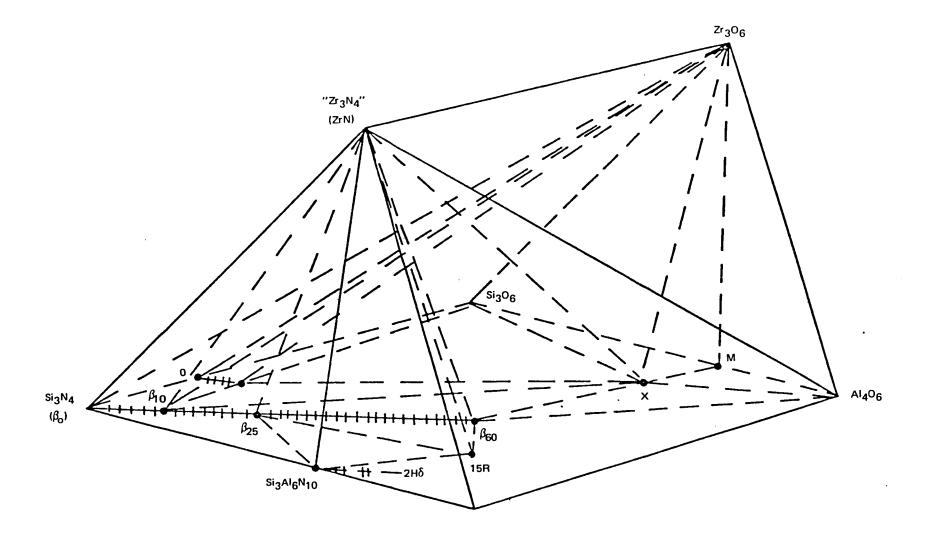


Figure 4. Probable Compatibility Relations (Partial) in the System Si<sub>3</sub>N<sub>4</sub>-Zr<sub>3</sub>N<sub>4</sub>-Al<sub>4</sub>N<sub>4</sub>-Al<sub>4</sub>O<sub>6</sub>-Zr<sub>3</sub>O<sub>6</sub>-Si<sub>3</sub>O<sub>6</sub> at One Atmosphere Pressure of Nitrogen

TABLE 3

# COMPATIBILITY TETRAHEDRA INVOLVING Si3-xA1x0xN4-x SOLID SOLUTIONS IN THE Zr-Si-A1-O-N SYSTEM

# 3 Phase Equilibria

# 4 Phase Equilibria

<sup>2</sup>. 
$$\beta'(0-10)^{-0}(0-10)^{-Zr0}$$
2

3. 
$$\beta'_{10}^{-0}_{10}^{-2}^{-2}_{2}^{-2}$$

5. 
$$\beta'$$
 (10-60) -X-ZrN

6. 
$$\beta'(0-25)^{-ZrN-Si}3^{A1}6^{N}10$$

7. 
$$\beta'_{25}$$
-ZrN-Si<sub>3</sub>A1<sub>6</sub>N<sub>10</sub>-15R

8. 
$$\beta'$$
 (25-60)  $-ZrN-15R$ 

(9. 
$$\beta'_{60}^{-15R-ZrN-8H}$$
)

(10. 
$$\beta'_{60}$$
-ZrN-8H-A1<sub>2</sub>0<sub>3</sub>)

(11. 
$$\beta'_{60}$$
-ZrN-A1<sub>2</sub>0<sub>3</sub>-X)

 $^{\beta'}$ (0-10)  $^{-Zr0}$ 2 plane (and presumably in the  $^{\beta'}$ (0-10)  $^{-0'}$ (0-10)  $^{-Zr0}$ 2 and  $^{\beta}$ 0-10  $^{-Zr0}$ 2  $^{-Si}$ 2 $^{N}$ 20 tetrahedra) also sinter to theoretical density, but the composition tested (Si<sub>2.7</sub>Al.30.3AlN<sub>3.7</sub>-5 w/o ZrO<sub>2</sub>) also exhibited unacceptably high creep rate. If sinterable formulations with acceptably low creep rates are to be found in this system, they must have a  $^{\beta'}$  composition very close to that of pure Si<sub>3</sub>N<sub>4</sub>.

#### B. System Y-Si-Al-O-N

#### 1. Background

As with the Zr-Si-Al-O-N system described above, this system is best treated as a six component reciprocal system which can be visualized by replacing the components "Zr $_3$ N $_4$ " and Zr $_3$ O $_6$  of Fig. with Y $_4$ N $_4$  and Y $_4$ O $_6$  respectively. Before describing experimental work in the system, the limiting binary, ternary and quarternary systems will be reviewed.

# a) Binary Systems Y203-A1203 and Y203-Si02

The Y $_2$ 0 $_3$ -Al $_2$ 0 $_3$  system has been investigated by a number of workers (Refs. 34, 35 and 26). Reference 34 shows the congrunetly melting compounds  $3Y_2$ 0 $_3$ -  $5Al_2$ 0 $_3$  (YAG) and  $2Y_2$ 0 $_3$ .Al $_2$ 0 $_3$  (YAM) with the latter disproportionating below about  $1000^{\circ}$ C. References 35 and 36 do not indicate disproportionation of YAM, and show the additional  $Al_2$ 0 $_3$ .Y $_2$ 0 $_3$  phase, (YAP), having a narrow temperature range of stability between about  $1830^{\circ}$ C and the melting point around  $1875^{\circ}$ C. Reference 35 shows incongruent melting of YAP, while Ref. 36 indicates congruent melting. The system Y $_2$ 0 $_3$ -Si0 $_2$  (Ref. 37) also contains three compounds, only two of which are stable at low temperature. These latter two compounds each exhibit several polymorphic forms; six modifications have been reported for the compound Y $_2$ Si20 $_7$ , and four for Y $_2$ Si0 $_5$ . The stability ranges of the various forms have not been firmly established. The crystal structure of the intermediate compound, Y $_4$ Si30 $_1$ 2, which is reported to be stable from about  $1650^{\circ}$ C to its melting point at  $1950^{\circ}$ C, is the same as that of Y $_3$ Al $_5$ 0 $_1$ 2 (YAG), suggesting the possibility of a solid solution between them at high temperatures.

# b) The Ternary System Y203-A1203-Si02

This system has been studied by Bondar and Galakov (Ref. 38) who delineated the liquidus surface but not the solid phase compatibility. Their diagram has been redrawn in terms of equivalent percent and is presented as Fig. 5. From the position and the direction of falling temperatures of boundary curves between primary crystal fields, one can infer the solid phase compatibility relations at temperatures where the compounds YAlO $_3$  and Y $_4$ Si $_3$ O $_{12}$  are stable. Again, from

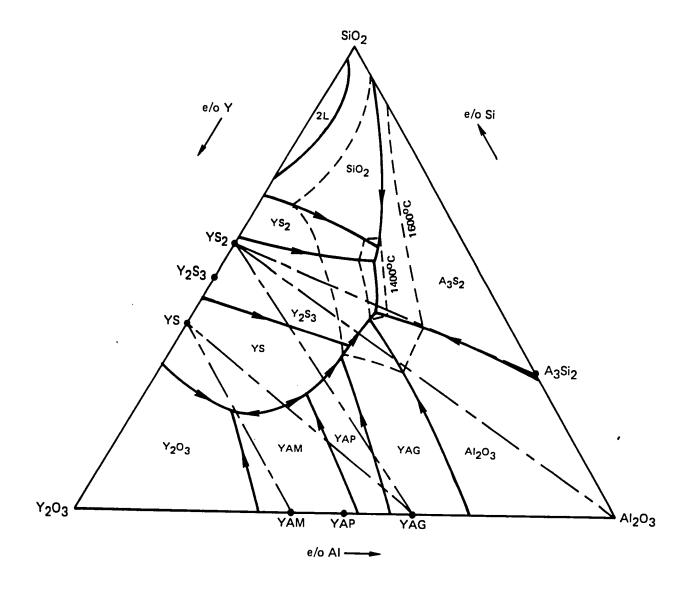


Figure 5 System-Y<sub>2</sub>O<sub>3</sub>-Al<sub>2</sub>O<sub>3</sub>-SiO<sub>2</sub> After Bondar and Galakov

the indicated temperatures on boundary curves, it can be inferred that the lower temperatures of stability of the  $YA10_3$  and  $Y_4Si_30_{12}$  phases are lowered somewhat in the ternary system suggesting some solid solubility in these phases. The extensive liquid region at  $1600^{\circ}$ C is noteworthy, as is the lowest ternary eutectic temperature of  $1350^{\circ}$ C. It can be anticipated that this liquid region would make a substantial incursion into the quinary system under consideration.

- c) The ternary system YN-AlN-Si3N4
- No information on this system was located.
- d) The quaternary system Si3N4-Si02-A1203-A1N

This system was discussed earlier and shown as Fig. 3.

e) The quaternary system Si3N4-Si0-Y203-YN

Three investigations of this system have been reported (Refs. 20, 39 and 40). A composite diagram reflecting a consensus view of the system is shown in Fig. 6. This system will be discussed more fully in Section IVE.

f) The quaternary system Al203-Y203-YN-AlN

This system has been investigated by Rae et al. (Ref. 20) and Holmquist (Ref. 41). Both references agree that the compatibility relations at  $1700^{\circ}$ C are as shown in Fig. 7.

# g) The quinary system

Rae et al. (Ref. 20) have studied compositions on the (nonternary) planes  $A1_20_3-Y_20_3-Si_2N_20$  and  $Si_3N_4-A1_20_3-Y_20_3$ . On the former plane, they report solid solution between the "J" or N-YAM phase  $Y_4Si_20_7N_2$  (see Fig. 6) and  $Y_4A1_20_9$  (YAM), and between "K" or N- $\alpha$ -wollastonite phase  $Y_5i_2N_1$  and  $Y_4A1_2N_2$  Other than these quinary solid solutions, no new crystalline phases were observed in the system. Phase assemblages found by Rae (Ref. 20) on the plane  $Si_3N_4-A1_20_3-Y_20_3$  are shown on Fig. 8.

During the course of our studies in the system, Naik and Tien (Ref. 42) presented the results of their studies of phase compatibility in the volume of the system bounded by phases  $\beta-Al_20_3-Y_20_3$ . Our earlier work in this volume of the system was presented in Ref. 22. The compatibility tetrahedra involving the  $\beta$  solid solution phase assigned by Naik and Tien and by Layden and Holmquist (Ref. 22) are compared in Table 4. These are essentially identical except for

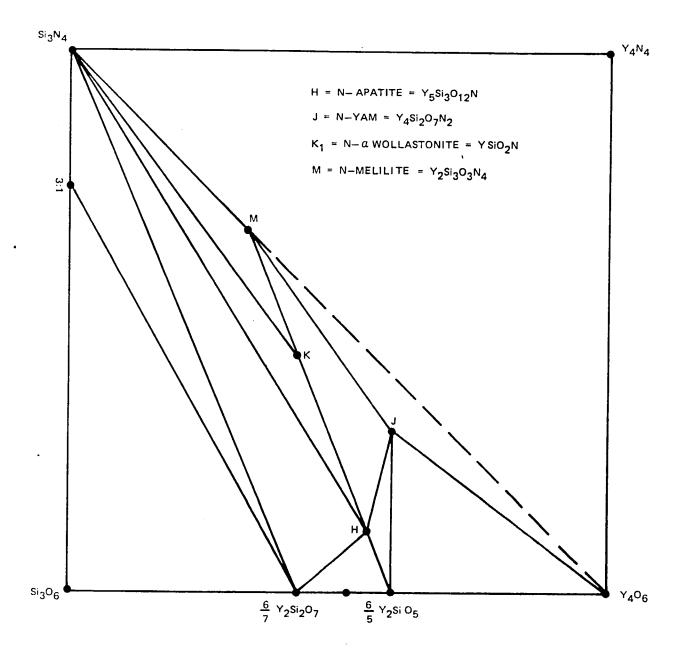


Figure 6 System  $Si_3N_4 - SiO_2 - Y_2O_3 - YN$  (Composite)

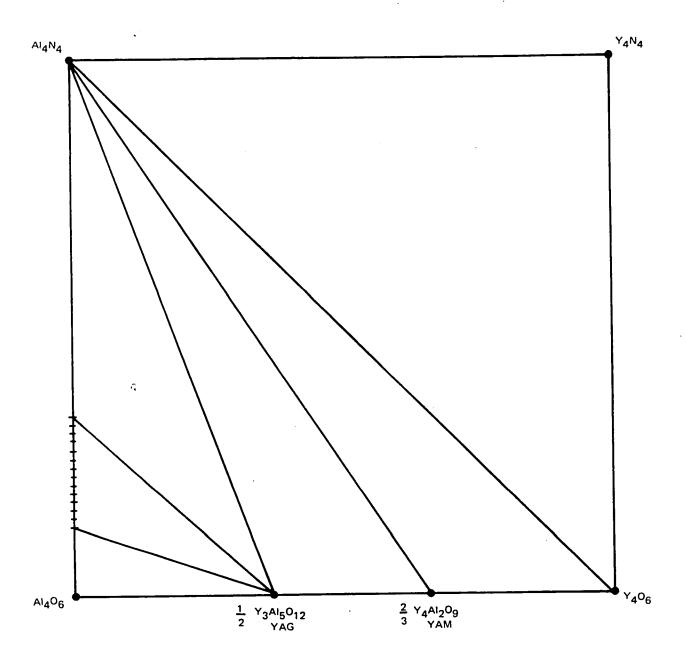
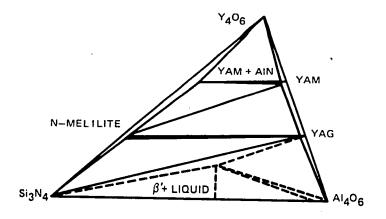
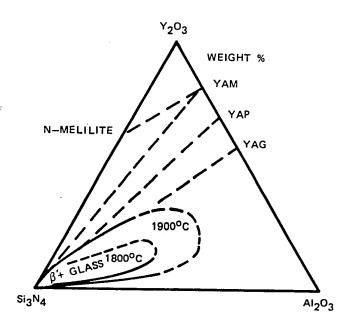


Figure 7 The System  $A\ell_2O_3-Y_2O_3-YN-A\ell N$ 

78-03-67-3



A. 1700°C



B. 1800 AND 1900°C

Figure 8 Phase Assemblages at Different Temperatures in the System  $Y_2O_3 \,-\, \text{Al}_2O_3 \,-\, \text{Si}_3N_4 \,\, \text{After Rae et al (ref 20)}$ 

TABLE 4

# COMPARISON OF PROPOSED β PHASE COMPATIBILITY TETRAHEDRA IN THE Y-Si-Al-O-N SYSTEM

Naik	and	Tien

$$\beta_{0-60}$$
 -  $YS_2$  -  $YAG$ 

$$\beta_{10-60} - YS_2 - X$$

$$\beta_{60} - YS_2 - X - YAG$$

$$\beta_{60} - A1_20_3 - X - YAG$$

$$_{0}^{\beta_{10}} - o_{b} - ys_{2} - x$$

$$\beta_{0-10} - O_{0-b} - YS_2$$

$$\beta_0$$
 - YAG - YS<sub>2</sub> - H

$$\beta_0$$
 - YAG - H - K

$$\beta_0$$
 - YAG - K - M

# Layden and Holmquist

$$\beta_{0-60}$$
 - YAG - X

$$\beta_0 - YS_2 - X - YAG$$

$$\beta_{60} - A1_2O_3 - X - YAG$$

$$\beta_0 - 0_0 - YS_2 - X$$

(ignored solubility limits)

$$\beta_0$$
 - YAG - YS<sub>2</sub> - H

$$\beta_0$$
 - YAG - H - K

$$\beta_0$$
 - YAG - K - M

Code:

$$\beta_{a-b}$$
 - all compositions from a to b e/o Al

$$^{\mathsf{YS}_2}$$
 -  $^{\mathsf{Y}_2}$ Si $_2$ O $_7$ 

$$YAG - Y_3A1_50_{12}$$

H 
$$- Y_{10}Si_6O_{24}N_2$$

$$M - Y_2Si_3O_3N_4$$

$$0 - Si_2N_2O_{(ss)}$$

$$X - Si_3 Al_6 O_{12} N_2$$
 (ss)

the subvolume bounded by the phases  $\beta'-Y_2Si_2O_7-Y_3Al_5O_{12}-X$  within which Naik and Tien established the tie line  $\beta_{60}-Y_2Si_2O_7$  not observed by Layden and Holmquist. After examining the data of Naik and Tien we agree that this tie line should be included.

#### 2. Experimental Investigation

#### a) Representation

While the Zr-Si-Al-O-N system could be depicted with reasonable clarity by the use of a perspective drawing (Fig. 4), this becomes virtually impossible in the case of the Y-Si-Al-O-N system because of the greater complexity of the bounding systems. For this reason a plexiglas model in the shape of an equilateral triangular prism was made. Transparencies of the various bounding systems were produced to the proper scale and fastened to the appropriate faces of the model. Holes were drilled through the model at the locations of the various phases, and thread was used to connect different points in the model in order to delineate various joins.

#### b) Results

Compositions prepared specifically for phase studies, conditions of firing, and tests results, are presented in tabular form in Appendix A. In general, substantial shrinkage during firing was an indication that partial melting had occurred. Nominally identical compositions prepared from different starting materials fired under similar conditions frequently exhibited rather different behavior and resulted in different phase assemblages in the fired bodies. A good example of this is found in data in Appendix A for samples located along the  $Y_3A1_5O_{12}$  - " $Y_3S1_5O_7N_5$ " join. This was particularly the case where substantial melting and bloating of samples were observed. It was also clear that in many instances a substantial amount of glass was retained in the samples at room temperature. As well as the data in Appendix A, phase data obtained from earlier, and subsequent test bars, were also considered in assessing phase compatibility. These data are found in tables in appendices B and C, and will be discussed when the physical property data for the various samples are discussed.

In regard to identification of the various AlN polytypes, we have used the notation of Jack (Ref. 7). However, with the exception of the 15R phase, identification of the X-ray diffraction patterns was made on the basis of X-ray data published by Land et al. (Ref. 4). It is clear that phases designated by Land as  $\theta$ ,  $\eta$ ,  $\epsilon$ , are the phases designated by Jack as 8H, 15R and 2H $^{\delta}$  respectively. It is not clear to which of the 3 remaining polytypes listed by Jack the Land  $\xi$  phase corresponds.

In constructing the model of the system, coexisting crystalline phases revealed by the data in the appendices were assumed to be compatible, and the points representing the compositions of these phases were connected by thread. As well as our own data, that of Rae et al. (Ref. 20) and Naik and Tien (Ref. 42) were used. The resulting model is shown as Fig. 9. Only tie lines relevant to equilibria involving the  $\beta$  solid solution phase have been included. Not shown on the model is the 8H phase which was not unambiguously identified in our samples. However topographical considerations would appear to necessitate a tie line between this phase and  $Y_3Al_5O_{12}$ , thus leading to two additions 4 phase tetrahedra involving  $\beta_{60}$  (SiAl\_2O\_2N\_2) not delineated on the model. Likewise, we have not drawn a tie line between the  $Y_2Si_3O_3N_4$  phase and YN which would be required if in fact no binary compounds occur between either  $Y_2O_3$  are YN on YN and  $Si_3N_4$ . This tie line would delineate the final tetrahedron involving  $Si_3N_4$ .

A list of the three and four phase tetrahdra delineated by the model (plus the three not delineated, but assumed to exist, and listed in parenthesis) are listed in Table 5.

C. Properties of Bodies in the Y-Si-Al-O-N System

#### 1. Primary Data

The formulations for various batches from which test samples were prepared are listed in Appendix B. Tables in Appendix B include information relating to raw materials and processing techniques used in the batch preparations. Unless otherwise stated there, the Si3N4 component was KBI powder. Fabrication, density, and mechanical test data for fired test specimens, are listed in Appendix C and oxidation data are presented in tables or curves in Appendix D. The above named appendices present data for samples from many different phase systems and are arranged chronologically. Data for Y-Si-Al-O-N samples were collected from the various appendices and compositions have been recalculated from the actual batch components (including grinding media pick-up where relevant) in terms of equivalent percent Al, Y, and O, and also in mole percent of the equilibrium phases based on the compatibility relationships shown in Fig. 9 and Table 5. These batch compositional data are presented in Table 6. Also shown in Table 6 are code symbols denoting the equilibrium phase assemblage. A quick glance at Table 6 shows that most of the samples investigated fall into one of the three phase tetrahedra  $\$  which involve broad ranges of  $\beta$  solid solution, as opposed to the four phase tetrahedra which involve unique  $\beta$  compositions. In general, the 4 phase tetrahedra may be considered peripheral to the main area of interest in the system.

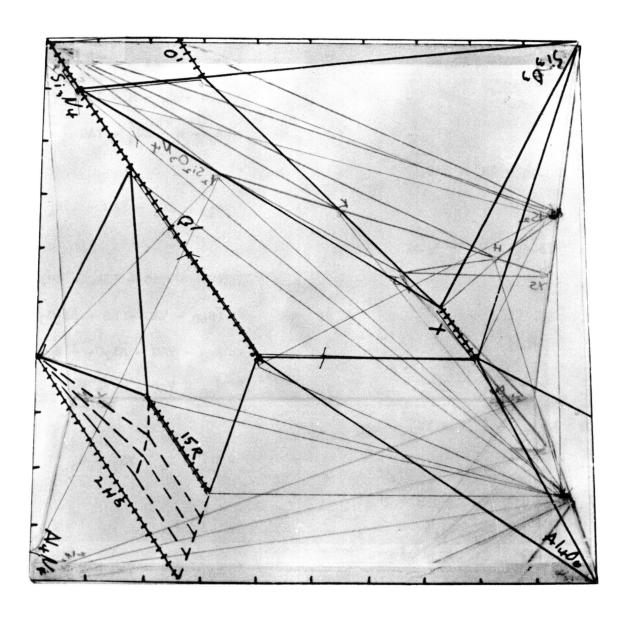


Figure 9 Model of the Y-Si-Al-O-N System

#### TABLE 5

# PROPOSED COMPATIBILITY TETRAHEDRA INVOLVING $si_{3-x}A1_xo_x^N4_{-x}$ (\$' SOLID SOLUTIONS) IN THE SYSTEM Y-Si-A1-O-N

	3 Phase		4 Phase
1.			$(YN - \beta_O - M - 2H_O^{\delta})$
2.			$\beta_{o}$ - M - $2H_{o}^{\delta}$ - YAG
3.	$\beta_{0-25} - 2H_o^{\delta} - YAG$		
4.		•	$\beta_{25} - 2H_0^{\delta} - YAG - 15R_a$
5.	$\beta_{25-60}$ - YAG-15R <sub>a-b</sub>		
6.			$(\beta_{60} - YAG - 15R_b - 8H)$
7.			$(\beta_{60} - YAG - 8H - Al_2O_3)$
8.			$\beta_{60}$ - YAG - Al <sub>2</sub> O <sub>3</sub> - $X_b$
9.	•		$\beta_{60}$ - YAG - $x_b$ - YS <sub>2</sub>
10.	$\beta_{10-60} - X_{a-b} - YS_2$		
11.			$\beta_{10} - x_a - ys_2 - o_b$
12.	$\beta_{0-10} - YS_2 - O_{0-b}$		
13.	$\beta_{0-60} - YS_2 - YAG$		
14.			$\beta_0 - YS_2 - YAG - H$
15.			$\beta_0$ - YAG - H - K
16.			$\beta_0$ - YAG - K - M
YS <sub>2</sub> YAG H K	- all composition from a to b e/o - Y <sub>2</sub> Si <sub>2</sub> O <sub>7</sub> - Y <sub>3</sub> AL <sub>5</sub> O <sub>12</sub> - Y <sub>1</sub> OSi <sub>6</sub> O <sub>2</sub> 4N <sub>2</sub> - YSiO <sub>2</sub> N - Y <sub>2</sub> Si <sub>3</sub> O <sub>3</sub> N <sub>4</sub> - Si <sub>2</sub> N <sub>2</sub> O (ss)	X 15R 2H <sup>S</sup>	- $Si_3A1_6O_{12}N_2$ (ss) - $SiA1_4O_2N_4$ (ss) - $Si_3A1_6N_{10}$ - $Si_3A1_16O_{12}N_{12}$ (ss)

TABLE 6

PHASE COMPOSITIONS OF TEST SAMPLES IN THE Y-Si-A1-O-N SYSTEM

Composition	Equiv	alent Pero	ent				1	Equilibrium	n Phase Co	omposition	(mole perd	ent) (	1)		
Designation (Appendix B)	<u>A1</u>	<u>Y</u>	<u>0</u> §	Symbol		i <sub>3-x</sub> A1 <sub>x</sub> 0 <sub>x</sub> N	1 <sub>4-x</sub>								
					m/o	<u>x</u>	e/o A1	YAG	ΥS 2	$\frac{2H_{\mathbf{o}}^{h}}{}$	<u>15R</u>		_M_	0 • (2)	<u>x</u> _
$\beta_1^* + 2.5 \text{ w/o } Y_2O_3$	7.62	0.82	5.74	$\nabla$	98.5	0.227	5.8	1.0					•		
$\beta_{1}^{2} + 5  \text{w/o}  \gamma_{2}^{2} O_{3}^{3}$	7.56	1.63	6.78	V	96.8	0.146	3.7	2.3		0.5					
$\beta_{2}^{1} + 2.5 \text{ w/o } Y_{2}^{2}O_{2}^{2}$	13.62	0.79	9.60	$\nabla$	98.5	0.486	12.7	1.0		0.9					
$\beta_{2}^{4} + 5  \text{w/o}  Y_{2}^{2} O_{3}^{3}$ 1078	14.14	1.64	10.75	Ÿ	96.6	0.386	10.0	2.3		0.5			•		
1078	1.04	9.61	16.23	Ă	90.6	0.500	0	1.0	0.8	1.0		7.			
1079	3.93	4.22	10.73	Δ	92.8	0	0	3.3	3.9			7.6			
1080	11.50	5.13	19.6	♦	90	0.500	13.0	3.3	10						
1081	23.90	5.31	28.3	X	90	1.00	27.3		10						
1085	6.00	3.06	9.5	$\Diamond$	95	0	0	5	10						
1086	32.03	3.81	25.33	Ď	95	1.0	27.3	5							
1087	60.27	4.22	46.73	<b>D</b>	95	2.0	60	5							
1088	4.82	3.37	8.88	Δ	95	0	0	4	•						
1089	3.70	3.31	8.06	Δ	95	0	0	3	1 2						
1090	3.03	2.97	7.30	Δ	95	0	0	2	3						
1091	27.65	1.35	18.78	lacktriangle	96.7	0	0	0.7	3						
1103	4.77	2.98	10.60	ò	93.4	.165	4.2	0.7	6	1.1			1.5	_	_
1104	43.13	1.95	13.64	Ď	97.5	1.0	27.3	2.5	O					0	.6
1105	27.52	1.38	19.43	á	93.1	0.82	22.0	1.7			5.2				
1111	2.79	4.32	10.99	Δ	93.7	0.05	1.3	4.9	1.4		3.2				
1112	2.28	2.92	7.47	0	85.9	0.14	3.5	4.7	2.8					11 2	
1114	12.64	2.55	12.38	Δ	96.4	0.364	9.4	2.8	0.8					11.3	
1115	25.91	1.21	14.40	$\nabla$	93.3	0.74	20.0	1.7	0.0	5.0					
1116	3.01	1.53	9.63	ò	86.5	0.1	2.5	1.,	5.7	3.0				7.0	
1118	20.76	0.97	2.16	-	79.5	0	0		3.1	17.9			2 6	7.8	
						•	v			11.7			2.6		

<sup>(1)</sup> Codes were presented in Table 5.

<sup>(2)</sup> In calculating composition it was assumed that the O' phase had the same equivalency aluminum substitution as the  $\beta$ ' phase.

Test data abstracted from the appendices are summarized in Table 7. Oxidation data are expressed as parabolic rate constants, even though reference to the primary oxidation data shows that oxidation was not necessarily parabolic. Such numbers were generally calculated from the total weight gain per unit area for the accumulated exposure time. This was done simply to give a number to plot that expresses in a qualitative way the oxidation rate relative to our program goals. The numbers become quantitative as they became small (on the order of 0.05  $mg^2cm^{-4}hr^{-1}$ ).

#### 2. Graphical Presentation of Property Data

#### Equivalent Percent Representation

It is desirable to have a method for displaying composition points for the quinery system on a single plane, so that property data can be plotted as functions of composition. This can be accomplished by plotting the cation equivalent percentage on triangular coordinate paper. This in effect projects all composition points onto a plane. The composition of each point can then be specified by labeling it with either a number corresponding to the equivalent percent oxygen, or a symbol corresponding to the equilibrium phase assemblage. The latter method has been adopted in the following plots, since this reserves the use of numbers for presenting property data.

To illustrate the plotting of compositions in this way, consider compositions in the compatibility tetrahedra  $\beta$  -YAG-YS $_2$  (i.e., Si  $_{3-x}$  Al  $_{x}$   $_{x}$  N<sub>4-x</sub> - Y<sub>3</sub>Al  $_{5}$ 0<sub>12</sub>- Y<sub>2</sub>Si  $_{2}$ 0<sub>7</sub>). Projections of these phases are shown in Fig. 10. The point A could represent for instance:

1. a binary mixture of  $\beta$ ' (x=1) and  $Y_3A1_50_{12}$ 2. a binary mixture of  $\beta$ ' (x=1.5) and  $Y_2Si_20_7$ , or 3. ternary mixtures of  $\beta$  (1 < x < 1.5),  $Y_3A1_50_{12}$  and  $Y_2Si_20_7$ .

An appropriate symbol placed at point  $\underline{A}$  would then serve to identify the given phase assemblage (symbols were given in Table 6).

An advantage of this type of representation is that it can be used to project every composition in the quinery system directly onto the phase diagram for the ternary system  ${\rm A1}_2{\rm O}_3$  -  ${\rm Y}_2{\rm O}_3$  -  ${\rm SiO}_2$ , i.e., the system which contains its oxidation products. This is a help in understanding the oxidation behavior of bodies as functions of composition.

TABLE 7

SUMMARY OF PROPERTY DATA FOR Y-Si-Al-O-N SAMPLES

				0xidati	on Data	F)	lexural Stre	ngth Data			
		Apparent	Bulk	Test	Rate	Test		Number			1370°C, 69MPa
ļ	Designation	Porosity	Density	Temp.	Constant	Temp.	Range	of	Mean	Value	Creep Rate
	(Appendix B)	(%)	(g/cc)	<u>(°c)</u>	$mg^2cm^{-4}hr^{-1}$	(°c)	(ksi)	Tests	(ksi)	MPa	(hr <sup>-1</sup> )
	β'1+25%Y <sub>2</sub> 0 <sub>3</sub>	< 1	3.04	1400	0.3	25	63-8	5	67	464	
						1370	42-65	2	54	373	
	β'1+5%Y <sub>2</sub> O <sub>3</sub>	< 1	3.20	1400	0.1	1370	42-05	-	54	3/3	
	β'2+2.5%Y <sub>2</sub> 0 <sub>3</sub>	< 1	3.26	1400	0.5						
	β'2+5%Y <sub>2</sub> 03 <sup>23</sup>		3.26	1400	0.6	25	37-85	14	59	407	6x10-5
ł				1300	0.003	1370	47-51	3	49	338	OXIO_2
1		•		1000	0.0000		51	3	47	330	
İ	1078	< 1	3.51	1400	0.15						
				1000	0.10						
	1079	4	3.25	1400	0.23						
				1000	0.0005						
	1080	0	3.36	1400	16						
				1000	0.0000	•					
	1081	0	3.24	1400	17						
1.5	1085	< 1	3.14	1400	17						
33	1086	1	3.11	1400	0.3				•		
	1088	< 1	3.23	1400	0.99						
				1400*	0.78						
	1089	1	2.71	1400	0.47						
				1400*	0.39						
	1090	1	2.96	1400	0.37						
				1400*	0.31						
	1091			1400	0.005						
	1103	< 1	3.25	1400	2.3						
				1300	0.03						
	1104	< 1	3.12	1400	0.033						
				1300	0.003						
	1105	< 1	3.21	1400	0.010	25	38-58	5	10		3
				1370	0.005	25**	34-60	8	48	331	$2.4 \times 10^{-3}$
				1300	0.001	25***	28-43	5	48	331	
				1000	0.0000	25	20=43	3	34	235	
	1111	0	3.40	1400	0.6						
	1112	0	3.07	1400	0.2						
	1114	0	3.19	1400	0.8						
	1115	0	3.09	1400	0.016	25	34-48	,			
			<b>-</b>	1000	0.0000	25***	34-48 25-35	4	40	276	
	1116	< 1	3.25	1400	3.0	23000	23 <b>-</b> 35	3	29	200	
				- · • •	3.0						

\*Following heat treatment in nitrogen at 1400°C for 64 hrs.

\*\*Following heat treatment in air at 1300°C for 100 hrs.

\*\*\*Following heat treatment in air at 1370°C for 100 hrs.

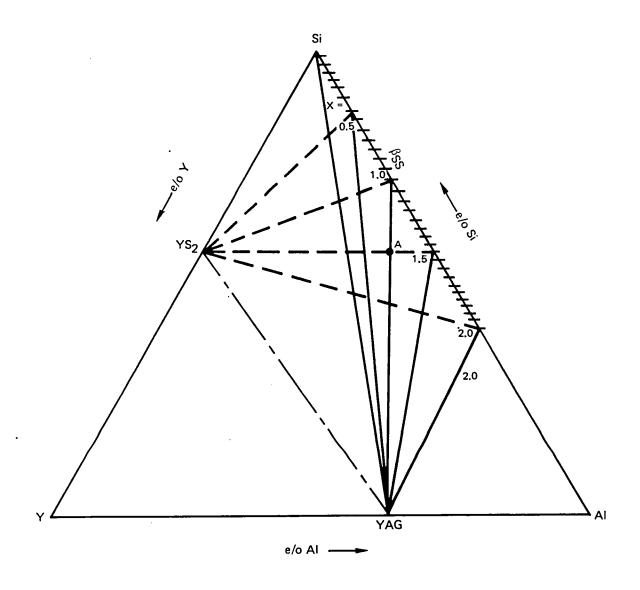


Figure 10 Planar Projection of the  $\beta'$  (0-60)-YAG-YS<sub>2</sub> Tetrahedron

#### 3. Densification

 ${\rm Si_{3-x}Al_{x}0_{x}N_{4-x}}$  bodies could not be sintered to greater than about 85 percent of theoretical density by firing to  $1750^{\circ}$ C, although  $\beta$ '-X phase bodies could be readily sintered to over 98 percent of theoretical density as a result of the presence of a liquid phase at the sintering temperature (Ref. 12). Liquid formation can also be promoted by the addmixture of  $Y_20_3$  and  $Y_20_3$ -Si0 $_2$  to Si3 $^{\rm N}_4$ and  $\beta$ ' compositions. However, in the absence of  $Al_20_3$  (or other fluxing oxides such as Be0 and  $Ce0_2$ ) the liquid which forms is not effective in promoting densification unless external pressure is applied to the system (i.e., hot pressing). In terms of the ternary projection of quinery compositions discussed in the previous section, it can be stated that compositions which project onto the Y-Si join cannot be sintered to high density, nor can compositions which project onto the A1-Si join, with the exception of compositions in the B'-X phase field. Between the two bounding regions there is a wide range of compositions which can be readily sintered to near theoretical density, as can be seen from table 7. The compositional limits within which densification can be effected at temperatures of  $1750^{\circ}$ C and below are about the 1 e/o aluminum line (in the absence of X phase) and 1 e/o yttrium line.

The effectiveness of  $Y_2 ^0 _3$ -Al $_2 ^0 _3$  addmixtures in sintering  $Si_3 ^N _4$  based bodies is certainly related to the formation of relatively low melting liquids which readily wet-out the  $\beta$  phase. How low in temperature liquid forms in parts of the quinery system can be judged from the fact that compositions lying on the  $\beta$ '-YS $_2$  plane (compositions 1080 and 1081, Appendix C and Table 7) sintered to high density at 1500°C. Compositions in the tetrahedron  $Si_3 ^N _4$ -YAG-YS $_2$  (1088, 1089 and 1090) appeared to consist of  $\beta$ ' and liquid (glass at room temperature) at temperatures at least as low as  $1400 ^{\circ} C$ .

## 4. Oxidation of Y-Si-A1-O-N Samples

### a. 1400°C Oxidation

The phase diagram for the system  $Al_2O_3-Y_2O_3-SiO_2$  was shown as Fig. 5. The  $1400^{\circ}\text{C}$  isothermal section drawn from Fig. 5 is presented in Fig. 11. The  $1400^{\circ}\text{C}$  oxidation rate data from Table 7 are plotted as described earlier, using the composition symbols given in Table 6, on the relevant portion of the  $Al_2O_3-Y_2O_3-SiO_2$  1400°C isothermal section in Fig. 12. Also plotted on Fig. 12 are oxidation rate data for two phase  $\beta$ '-X phase bodies from Ref. 12 which project onto the Si-Al join, and data for Y-Si-Al-O-N samples reported by Arias (Ref. 43) and Dutta (Ref. 44). It can be seen that, except for compositions which project onto the Si-Al or Si-Y joins, all compositions will have oxide scales at  $1400^{\circ}\text{C}$  which contain some liquid. The equilibrium phase

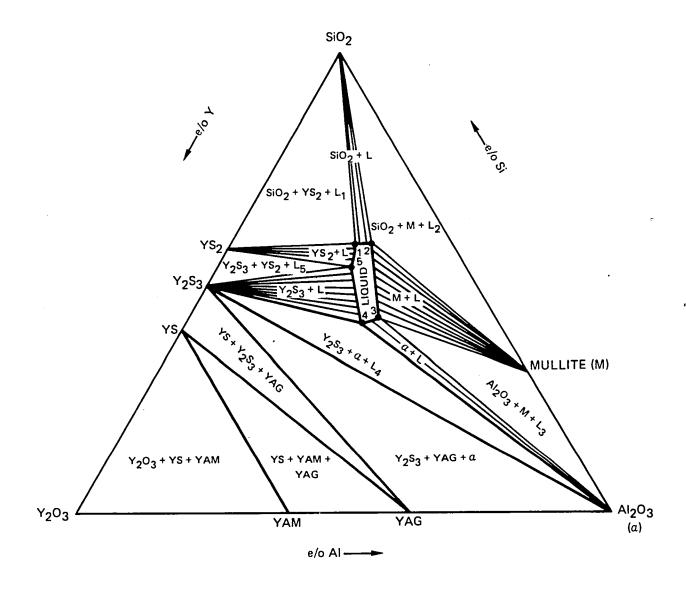


Figure 11 1400°C Isothermal Section of the Al<sub>2</sub>O<sub>3</sub>-Y<sub>2</sub>O<sub>3</sub>-SiO<sub>2</sub> System

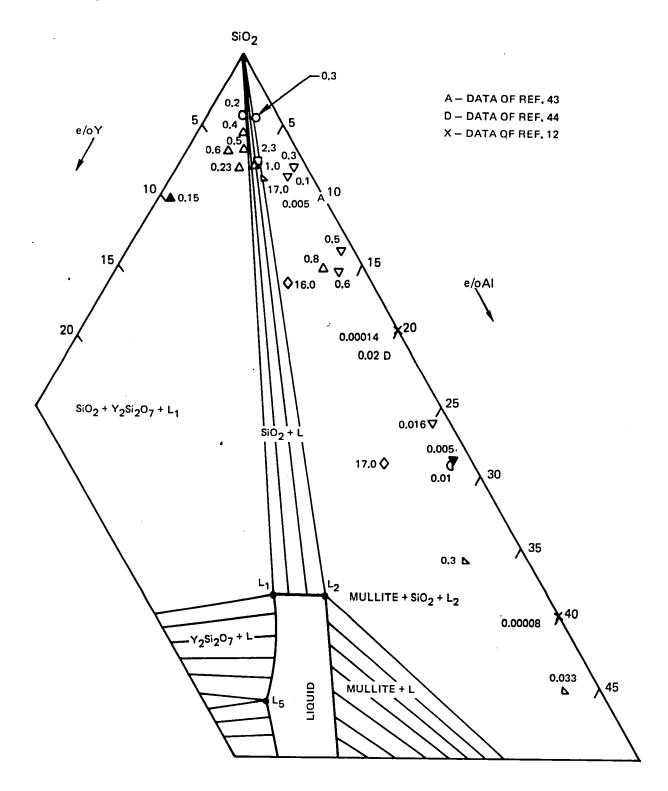


Figure 12 1400°C Oxidation Rate Constants (mg<sup>2</sup> cm<sup>-4</sup> hr<sup>-1</sup>) for Samples in the Y-Si-Al-O-N System, Plotted on the 1400°C Isothermal Section of the Al<sub>2</sub>O<sub>3</sub>-Y<sub>2</sub>O<sub>3</sub>-SiO<sub>2</sub>

Phase Diagram (Symbols Defined in Table 6)

compositions of scales can be readily calculated. A simple way to do this is to express all relevant compositions in terms of a constant valence (e.g.,  $\pm$  4). Such formulations for SiO<sub>2</sub>, mullite, yttrium disilicate, and the liquids  $L_1$  and  $L_2$ , along with their molecular weights and densities\* (needed to convert mole percent into volume percent) are presented in Table 8. Coefficients for chemical equations of the type:

$$Y_xA1_ySi_zO_2 = aSiO_2 + b(Y_{.571}Si_{.571}O_2) + c(Y_{.240}A1_{.305}Si_{.590}O_2)$$

are then determined to calculate mole percent of the phases. The calculated phase compositions for scales for some representation compositions, and their oxidation rate constants, are presented in Table 9. Micrographs of polished sections through the scales of these same samples are shown in Figs. 13 and 14. In general, the data plotted in Fig. 12 indicate that high concentrations of cristobalite and liquid in the scales result in high oxidation rates. By far, the overriding effect is that of liquid concentration. Thus composition 1080.2, which by calculation, and by rough visual assessment of Fig. 13-B, has a scale consisting of about 29 percent liquid, 68 percent cristobalite and 3 percent mullite (only cristobalite was detected by XRD) oxidized extremely rapidly. The scale was virtually nonprotective, despite the great thickness (250  $\mu$ ) that built up in only 1.24 hours of exposure, attesting to the rapid diffusion of oxygen through the liquid.

Samples having scales with lower calculated liquid concentrations, but relatively higher cristabalite concentrations, exhibited intermediate rates (but still excessive in terms of performance requirements), with the rate decreasing with decreasing liquid and cristobalite concentrations, as seen from the data for samples 1078.1 (Fig. 13A) and llll.1 (Fig. 14B). Another effect apparently unrelated to scale composition per se is evident from the micrograph of sample 1078.1. This shows the generation of subscale porosity apparently associated with diffusional processes (grain boundary?) and internal oxidation. It will be seen later that this composition also oxidizes badly at  $1000^{\circ}\text{C}$  as well).

In contrast to the above examples, sample 1105.1, Fig. 14A, with a low liquid, intermediate (but still major) cristobalite, and a substantial mullite concentration (about 30 percent) in the scale exhibited a low oxidation rate. The scale was protective, and built to a thickness of about 25  $\mu$  in 96 hours of exposure. This is generally seen to be the case: the only sintered samples studied which had nominally acceptable oxidation rates at  $1400^{\circ}\text{C}$  (on the order of 0.01  $\text{mg}^{2}\text{cm}^{-4}\text{hr}^{-1}$ ) were those samples which exhibited scales with high concentrations of mullite (about 30 percent or

<sup>\*</sup>Densities of the liquids were estimated by extrapolation of the density data of Bacon (Ref. 45) for yttrium bearing glasses.

TABLE 8

SOME EQUIVALENT FORMULATIONS, MOLECULAR WEIGHTS, AND DENSITIES FOR SOME OXIDE PHASES

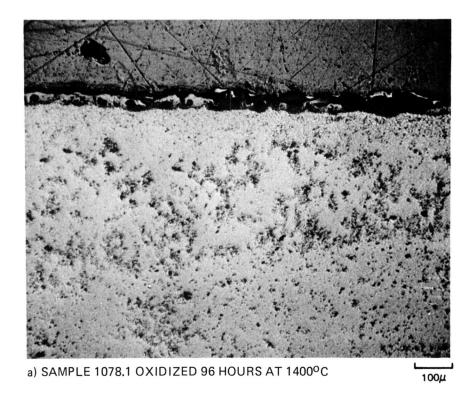
Conventional Formula	Equivalent Formula	Molecular Weight	Density (g/cc)
sio <sub>2</sub>	S10 <sub>2</sub>	60.1	2.3
A1 <sub>6</sub> Si <sub>2</sub> O <sub>13</sub>	A1 <sub>.923</sub> Si <sub>.308</sub> O <sub>2</sub>	65.5	3.15
Y <sub>2</sub> Si <sub>2</sub> O <sub>7</sub>	Y.571 Si.571 O2	98.8	4.15
Liquid L <sub>1</sub>	Y.240 Al.306 Si.59 O2	76.6	∿ 3.8
Liquid L <sub>2</sub>	Y.200 A1.347 Si.590 O2	75.7	∿ 3.8

TABLE 9

CALCULATED VOLUME PERCENT PHASES IN 1400°C

OXIDE SCALES ON SOME Y-Si-Al-O-N SAMPLES

Composition Number		uivalent ercent Y	± 	Equivalent Formulation	SiO <sub>2</sub>	Volume Po	A1 .923 Si .308 O 2	cale L1	<u>L2</u>	Oxidation Rate Constant (mg <sup>2</sup> cm <sup>-4</sup> hr <sup>-1</sup> )
1078	1.04	- 9.61	16.23	Y.128 <sup>A1</sup> .014 <sup>S1</sup> .894 <sup>O2</sup>	77.0	19.1	0	3.9	0	0.15
1080	11.50	5.13	19.60	Y.068 <sup>A1</sup> .153 <sup>Si</sup> .834 <sup>O</sup> 2	68.2	0	3.3	0	28.5	16.0
1105	27.52	1.38	19.43	Y.018 <sup>A1</sup> .367 <sup>Si</sup> .771 <sup>O</sup> 2	60.4	0	32.1	0	7.5	0.01
1111	2.79	4.32	10.99	Y.058A1.037Si.930 <sup>0</sup> 2	89.4	4.0	0	6.6	0	0.6



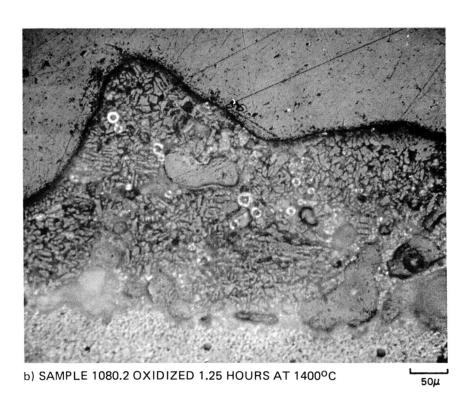
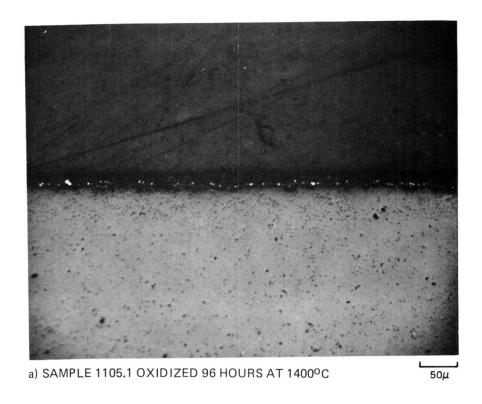


Figure 13 Oxide Scales on Samples 1078.1 and 1080.2



b) SAMPLE 1111.1 OXIDIZED 45 HOURS AT 1400°C

Figure 14 Oxide Scales on Samples 1105 and 1111.1

greater) which tend to form a dense matt of accicular crystallites. (For  $\beta$  -X phase samples which exhibit the lowest oxidation rates, only mullite was detected in the scale by XRD.)

Hot pressed  $\rm Si_3N_4$  bodies containing  $\rm Y_2O_3$  as the sintering aid exhibit very low oxidation rates (Refs. 40, 46). The crystalline oxidation products are cristobalite and  $\rm Y_2Si_2O_7$  (Ref. 47). By way of analogy with the mullite forming bodies in the Y-Si-Al-O-N system, it could be anticipated that bodies in the same system with scales bearing on the order of 30 volume percent  $\rm Y_2Si_2O_7$  crystals would also exhibit oxidation rates on the order of 0.01 mg<sup>2</sup>cm<sup>-4</sup>hr<sup>-1</sup> at  $\rm 1400^{o}C$ . One can then inquire what cation equivalent percents would yield such a concentration of  $\rm Y_2Si_2O_7$  crystal, given the constraint that on the order of 1 e/o Al is necessary to permit pressureless sintering. Calculations shown that this would require an equivalent formulation of the oxide scale composition to be about  $\rm Y_{.182Al_{.013}Si_{.854}O_2$ . This would be the oxidation product, for instance, at a body that was about 71.5 percent  $\rm Si_{.2.943Al_{.057}O_{.057}O_{.057}N_{3.943} + 28.5$  m/o  $\rm Y_2Si_2O_7$ , or, for another instance, 71.6 m/o  $\rm Si_3N_4 + 27.4$   $\rm Y_2Si_2O_7 + 0.9$   $\rm Y_3Al_5O_{12}$ . Such bodies would require about 33 w/o  $\rm Y_2O_3$  in their formulations. Bodies with such high yttrium concentrations were not investigated.

Returning to consideration of the mullite forming bodies, examination of the mechanical test data presented in Table 7 shows that samples of compositions with low 1400°C oxidation rate constants (compositions 1105 and 1115) nonetheless suffer significant degradation in room temperature flexural strength following 100 hrs exposure to air at 1370°C, although there was no degradation after similar exposure at 1300°C. Examination of the micrographs of the polished sections shown in Figs. 13 and 14 shows that the scale-body interface does not advance smoothly during 1400°C oxidation, but generates a rather pitted surface which is probably responsible for the loss in strength.

## b. 1300°C Oxidation

The  $1300^{\circ}\text{C}$  parabolic oxidation rate constants for some Y-Si-Al-O-N samples are plotted vs composition in Fig. 15. At this temperature (which is below the minimum solidus temperature in the  $Y_2O_3$ -Al $_2O_3$ -SiO $_2$  system) oxidation rates are acceptionally low (particularly for the samples which would have a substantial fraction of mullite in their scales) and as mentioned earlier, the samples tested showed no degradation in room temperature flexural strength after 100 hrs of air exposure.

## c. 1000°C Oxidation

Most of the samples subjected to oxidation at  $1000^{\circ}\text{C}$  showed little or no weight gain or other evidence of sample degradation. The exception was

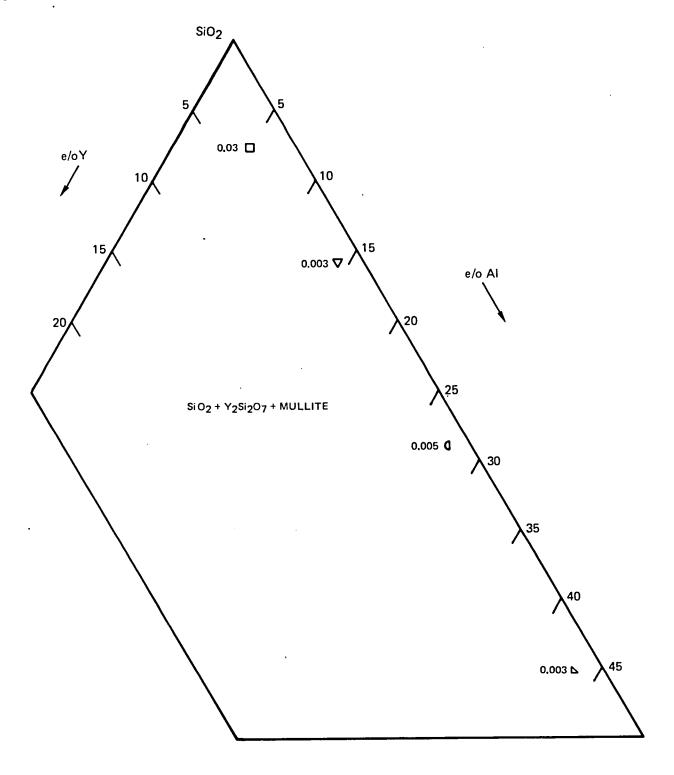


Figure 15 1300°C Oxidation Rate Constants (mg<sup>2</sup> cm<sup>-4</sup> hr<sup>-1</sup>) for Samples in the Y-Si-Al-O-N Systems (Symbols Defined in Table 6)

composition 1078, lying in the four-phase field  $\mathrm{Si_3N_4}$ -YAG-YS<sub>2</sub>-H which oxidized nearly as fast at  $1000^{\circ}\mathrm{C}$  as at  $1400^{\circ}\mathrm{C}$ . (As shown earlier, this sample showed evidence of several sub-scale degradation during  $1400^{\circ}\mathrm{C}$  oxidation.) Macroand micro-graphs of the polished cross section of sample 1078.2 are presented in Fig. 16 which show the extent of the attack after 65 hrs at  $1000^{\circ}\mathrm{C}$ . X-ray diffraction shows the surface region to consist of the X<sub>I</sub> modification of  $\mathrm{Y_2Si0_5}$  and  $\mathrm{B}'$ .

#### 5. Mechanical Properties

The mean room temperature and 1370°C strength values, and 1370°C 69 MPA creep data for various samples recorded in Table 7 are plotted against composition, as discussed earlier, in Fig. 17. It can be seen that the general trend is for strength (measured both at 25°C and 1370°C) to decrease, and creep rate to increase, with increasing aluminum concentration. The progressively poorer elevated temperature mechanical properties correlated with the decrease in solidus temperature with increasing aluminum concentration mentioned earlier. While various types of recognizable flaws served to initiate fracture at 25°C, distinct flaws were generally not resolved on the fracture surfaces of samples tested at 1370°C, suggesting that the grain boundaries themselves were strength controlling at this temperature.

The  $\beta'-Y_2O_3$  formulations (entires 1 through 4 of tables 6 and 7) were studied in some detail in Ref. 21. Flaws that were identified from SEM and microprobe analyses of the fracture surfaces of the samples included:

- 1) Metallic inclusions, exemplified by Fig. 18. These are silicontransition metal particles resulting from the impurity-initiated decomposition  ${\rm Si}_3{\rm N}_4$  during firing.
- 2) Small irregular voids, exemplified by Fig. 19. These may be the sites of inclusions that were ejected at fracture.
- 3) Large voids, probably resulting from the inclusion of foreign matter in the sample.

Fractures of composition 1105 also initiated irregular voids near the surface, exemplified in Fig. 20, or at edges, although at substantially lower stress levels than the  $\beta\,'-Y_2O_3$  samples. Micrographs at HF etched and unetched polished surfaces of sample 1105.1, Fig. 21, show a coarse grain structure with a relatively high volume fraction at etched material, probably glass, in grain boundaries.



4500μ



200μ

Figure 16 Polished Cross Section of Sample 1078.2 Oxidized at 1000°C for 65 Hours

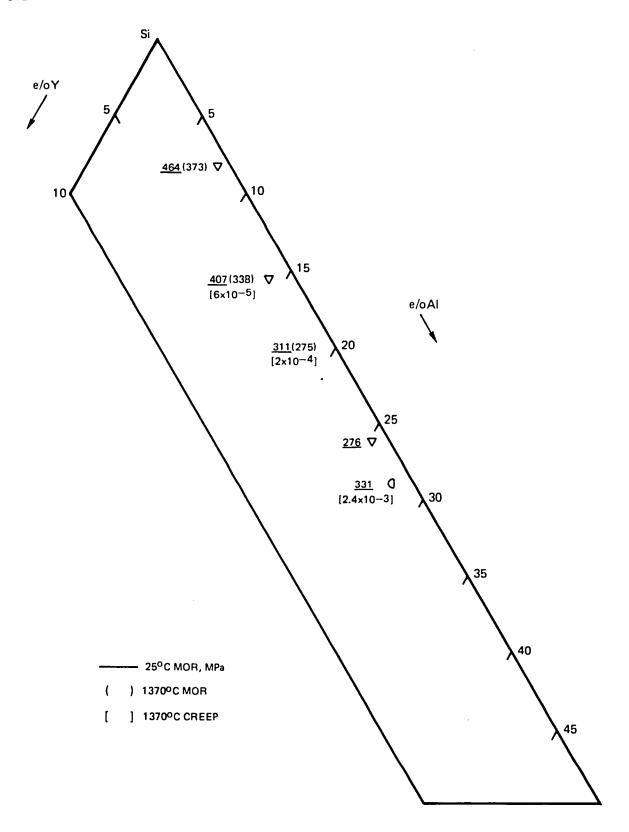
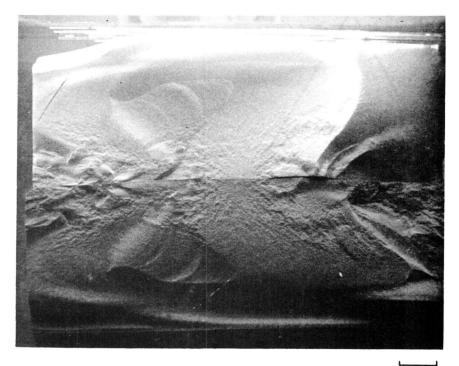


Figure 17 Mechanical Property Data vs. Composition for Some YSiALON Samples (Symbols Defined in Table 6)



700µ

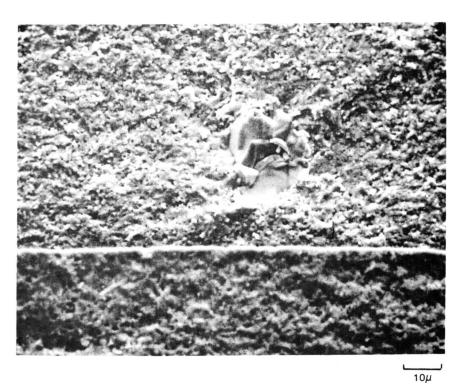
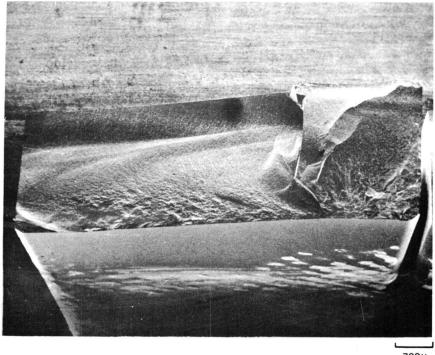


Figure 18 Fracture Initiated at Metailic Inclusion in  $\beta'$ 1 + 2.5 w/o Y<sub>2</sub>O<sub>3</sub> Sample ( $\sigma$  = 386 MPa)



700μ

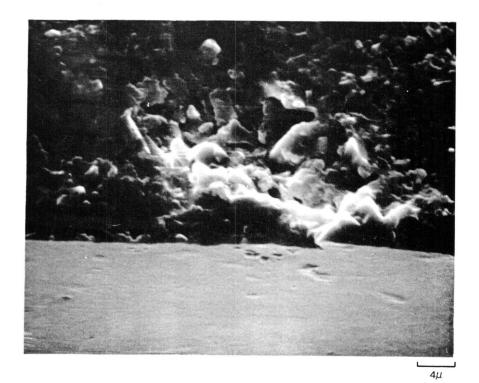


Figure 19 Fracture Initiated at Subsurface Void in  $\beta'$ 2 + 5 w/o Y<sub>2</sub>O<sub>3</sub> Sample ( $\sigma$  = 510 MPa)

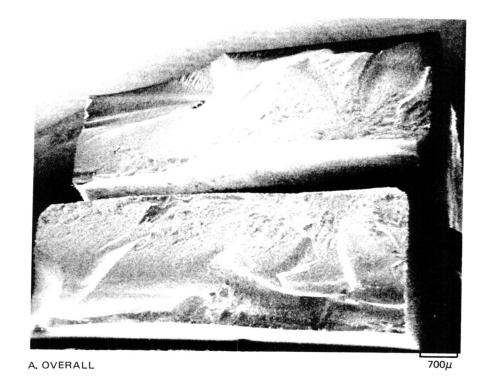




Figure 20 Sample 1105.10 Fracture Surface ( $\sigma$  = 341 MPa)

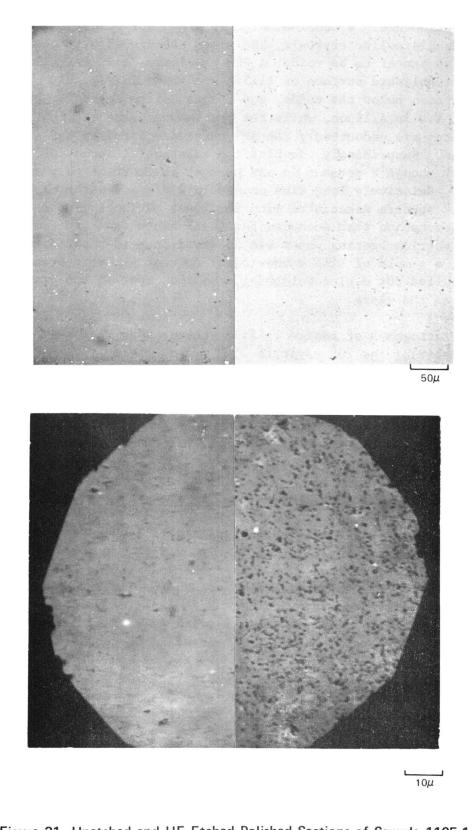


Figure 21 Unetched and HF Etched Polished Sections of Sample 1105.1

51

78-09-136-7

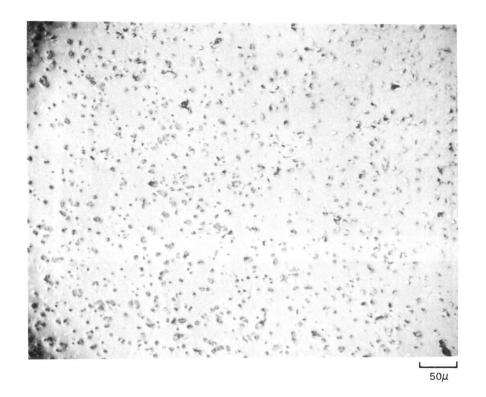
The samples with the lowest strength - those of composition 1115 - exhibited a particularly coarse grained microstructure shown in Fig. 22. This shows a three-phase structure: a continuous phase presumably consisting of  $\beta$ ' crystals discontinuous plate-like crystals, and areas associated with the plate-like crystals which appear to be void. A SEM photograph and EDAX silicon and aluminum maps from the polished surface of 1115.2 are shown in Fig. 23. It can be seen that the surfaces below the voids, and the lath-like crystals, are high in aluminum and low in silicon, while the continuous phase is rich in both Al and The plates are undoubtedly the  $2 H^{\delta}$  phase disclosed by XRD to be present in the sample. Surprisingly, despite the substantial amount of Y in the sample, there was not enough Y present in any part of the surface to permit making an yttrium map. Relatively long time counts in different parts of the surface produced weak spectra associated with the lower surfaces of the void areas. A reasonable assumption that accounts for these facts (and the fact that no crystalline yttrium bearing phase was observed in powder obtained by crushing and grinding a sample of 1115 composition) is that the void areas contained glass that pulled out during polishing, and that most of the yttrium in the samples was present in the glass.

A SEM fractograph of sample 1115.2, presented as Fig. 24 more clearly shows plate-like habit of the  $2{\rm H}^{\delta}$  crystals.

From the above observations it may be concluded that the lower room temperature strength of the high aluminum samples is also related, albeit indirectly, to the lower solidius temperature and higher liquid concentration at sintering temperature in this part of the system, which leads to coarsening of the microstructures.

## 6. Summary of Properties of Y-Si-Al-O-N Samples

Five 3-phase compatibility tetrahedra and 11 4-phase tetrahedra exist in the quinery system wherein one phase is either  $\mathrm{Si}_3\mathrm{N}_4$  or a  $\mathrm{Si}_{3-\mathrm{X}}\mathrm{Al}_{\mathrm{X}}^0\mathrm{N}_{4-\mathrm{X}}$  solid solution. Compositions from all of the 3-phase (or bounding 2-phase planes) and two of the 4-phase tetrahedra ( $\mathrm{Si}_3\mathrm{N}_4$  -  $\mathrm{YAG}$ - $\mathrm{YS}_2$ -H, and  $\mathrm{Si}_3\mathrm{N}_4$  -  $\mathrm{YAG}$ - $\mathrm{ZH}^\delta$ -M) where screened for oxidation and mechanical properties. The only sinterable bodies with low parabolic oxidation rate constants at 1400°C (on the order of 0.01  $\mathrm{mg}^2\mathrm{cm}^{-4}\mathrm{hr}^{-1}$ ) were compositions whose oxides fell in the primary phase field of mullite and produced scales at 1400°C that were about 30 volume percent or more mullite. Unfortunately, these compositions fall in regions of the quinery system where the solidus temperature is low (perhaps below 1500°C) so that 1370°C mechanical properties are poor. No sinterable composition was found that exhibited oxidation and mechanical properties that would be acceptable for high stress applications in oxidizing atmospheres at 1370°C (2500°F). However, bodies were produced which exhibit properties



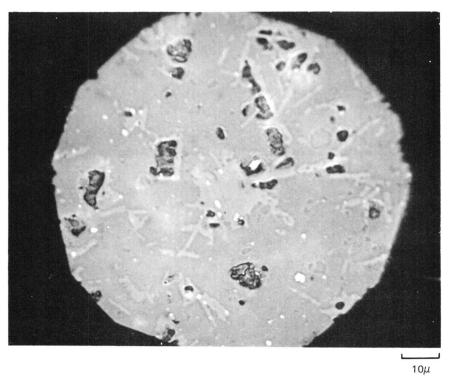


Figure 22 Relief Polished Section of Sample 1115.2

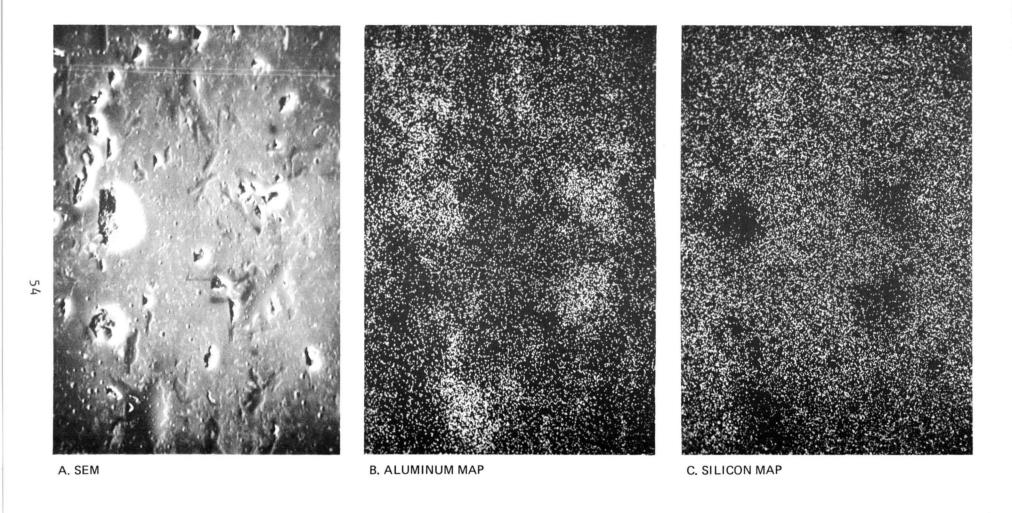
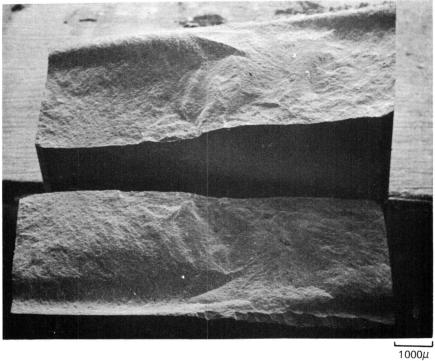


Figure 23 SEM and EDAX Element Maps of Sample of Composition 1115



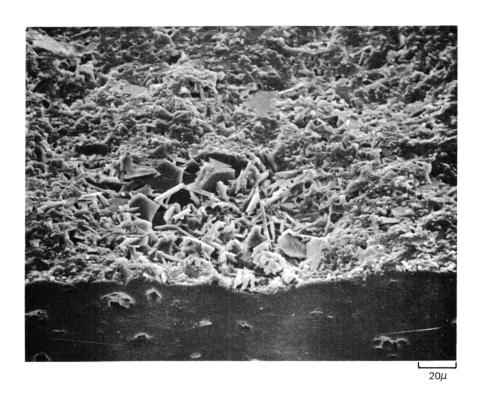


Figure 24 Fracture Surface of Sample 1115.3  $(\sigma = 327 \text{ MPa})$ 

attractive for stressed application at temperature of  $1300^{\circ}$ C and below. One formulation, (96.6 m/o Si $_{2.614}^{A1}$ .386 $^{0}$ .386 $^{N}$ 3.614 + 2.3 m/o Y $_{3}^{A1}$ 50 $_{12}$  + 1.0 m/o Si $_{3}^{A1}$ 6 $^{N}$ 10) and process, produced samples which exhibited the following properties:

Property	<u>Value</u>
1300°C oxidation rate	$0.003 \text{ mg}^2 \text{cm}^{-4} \text{hr}^{-1}$
25°C MOR (mean)	518 MPa
1370°C MOR (mean)	340 MPa_
1370°C, 69 MPa creep	$6 \times 10^{-5} hr^{-1}$

D. System Y-Si-Be-O-N

#### Background

## a) System BeO- $Y_2O_3$ -SiO<sub>2</sub>

Prior work on some binary subsystems includes that or Morgan and Hummel on the BeO-SiO<sub>2</sub> system (Ref. 48), Bondar and Galakov on the  $Y_2O_3$ -SiO<sub>2</sub> system (Ref. 38), Olds and Otto (Ref. 49), and Bartran (Ref. 50) on the system BeO- $Y_2O_3$ . Olds and Otto show no compounds between BeO and  $Y_2O_3$ , but indicate extensive solution of BeO in  $Y_2O_3$  about  $1000^{\circ}$ C. Bartram presented a tentative phase diagram which showed three incongruently melting compounds with molor ratios BeO: $Y_2O_3$  of 1:1, 4:1\* and 9:1, and indicated the possibility of metastability. Other investigators (Refs. 51, 52 and 53) had prepared the 1:1 compound from the melt, but could not prepare it in the solid state. Harris and Yakel (Ref. 53) determined its crystal structure and speculated that the atomic environment was more nearly representative of liquids than of equilibrium solids. They reported that the compound tended to dissociate into the component oxides when heated (to unspecified temperature).

The only reference located on work in the ternary system is that of Bartram (Ref. 54) who discovered single crystals, later determined to be of composition  $Y_2Si_2Be_2O_7$ , growing on the surfaces of  $BeO-Y_2O_3$  samples fired in a furnace insulated with silica brick. He later heated a pellet of this composition prepared from the component oxides in air of  $1350^{\circ}C$  and found it to consist of  $Y_2SiBe_2O_7$  along the traces of  $Y_2O_3$ ,  $Y_2BeO_4$  and cristobalite. Bartram determined the structure of  $Y_2SiBe_2O_7$ , and showed it ot isomorphous with minerals of the melilite family.

\*Latter shown to be Y2SiBeO7 (Bartram, personal communication).

# b) The system $Si_3N_4-Si_0_2-Y_2O_3-YN$

This was discussed in Section IV.B.1.e.

c) The system  $BeO-SiO_2 - Si_3N_4 - B_3N_2$ 

This system was investigated by Huseby et al, (Ref. 55). His diagram is reproduced as Fig. 25.

#### 2. Experimental Investigation

# a) System $BeO-Y_2O_3-SiO_2$

Given the demonstrated existence of the melilite phase, it was questioned whether other silicate or aluminate analogs exist in this system which could constitute possible fully oxidized matrix phases for  $\mathrm{Si}_3\mathrm{N}_4$  bodies. Compositions prepared, heat treatments given, and phases observed by XRD are presented in Table 10. Note that the compositions numbers 1, 2, 3 and 4 correspond to possible isomorphs to the melilite,  $\alpha$  wollastonite, apatite, and YAM phases respectively, which also have nitrogen substituted analogs in the  $\mathrm{Si}_3\mathrm{N}_4$  -  $\mathrm{SiO}_2$  -  $\mathrm{Y}_2\mathrm{O}_3$  system (Ref. 20). Pellets fired at  $1400^{\circ}\mathrm{C}$  exhibited little shrinkage, (less than 5 percent) or other evidence of liquid formation.

The only ternary phase that was positively identified was the melilite phase  ${\rm Y_2Be_2SiO_7}$  previously reported by Bartram. Our XRD data for this phase are compared with those of Bartram in Table 11. The pattern of another phase was also observed in a number of the samples. The XRD data for this phase are presented in Table 12 and are compared with the data of Lange for the YAM isomorph  ${\rm Y_4Si_2O_7N_2}$  (J phase) and that of Bartram for  ${\rm Y_2BeO_4}$ .

The observed x-ray diffraction pattern is seen to bear a close resemblance to both the J and  $Y_2BeO_4$  phases. Since all mixtures containing the new phase were multiphase, it is not certain that the diffraction data presented are complete or accurate because of possible overlap with peaks from other phases. Although at first reading the data for the new phase appear more closely to approximate that of J phase, if one eliminates the peaks attributed to  $Y_2Be_2SiO_7$  from the pattern it is seen to be very similar to the  $Y_2BeO_4$  pattern, although with a somewhat expanded lattice which could be anticipated if some solubility toward  $Y_2SiO_5$  existed. More significant perhaps, is the fact that the ambiguous 4-phase data for compositions 5, 7, and 11 can be readily interpreted in

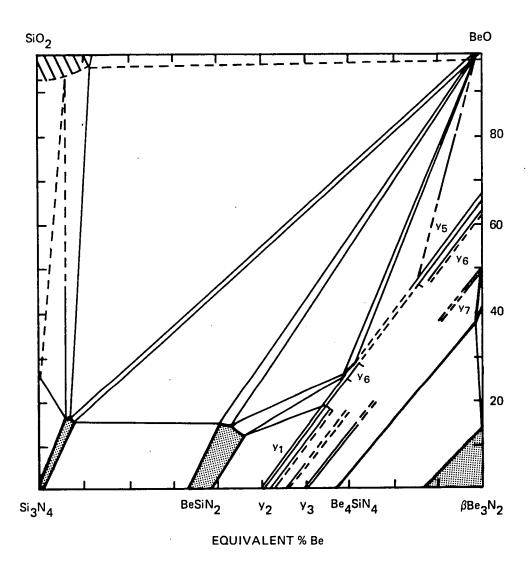


Figure 25 System  $Si_3N_4$ - $SiO_2$ -Be $_3N_2$ -BeO at 1760°C, After Huseby et. al. (ref. 53)

Table 10 Phase Observed After Heat Treatment of Compositions in the System  ${\rm BeO-Y}_2{\rm O}_3{\rm -SiO}_2$ 

Comp. Number				Heat Treatment*			I	hases obse	erved by x-1	ay diffract	ion
	BeO	Y <sub>2</sub> O <sub>3</sub>	S10 <sub>2</sub>		BeO	<sup>Y</sup> 2 <sup>0</sup> 3	$^{\text{SiO}}_{2}$	<sup>Y</sup> 2 <sup>Si0</sup> 5	<sup>Y</sup> 2 <sup>Si</sup> 2 <sup>O</sup> 7	Y2B2S107	Y <sub>2</sub> Be0 <sub>4</sub>
1	50.0	25	25	. A					m	s	
2	33.3	33.3	33.3	<b>A</b>				m	w	S	
3	9	45.5	45.5	A				s		m	
4	25	50	25	A		s		m			s
5	33.3	50	16.7	A		s		w		m	s
6	40	20	40	Α			m			s	3
7	65	25	10	A	w	m					
7a	65	25	10	В	w					m	S
<b>7</b> b	65	25	10	C		m				S	s
8	70	7.5			W	s				s	S
J	70	7.5	22.5	A	W		m			s	
9	50	10	40	Α	W		m			s	•
10	33.3	33.3	33.3	Α		w				•	
10a	33.3	33.3	33.3	В		w		s	m	s s	w w
11	25	5.0								J	•
	25	50	25	Α		s		m		m	s
lla	25	50	25	B C		s		m			
TTP	25	50	25 at 1400°C	С		S		m		m m	s
R: heat	ed for 60	) hours a	at 1400°C			_		ш		m	s

B; heat treatment A samples crushed to powder recompacted and heated an additional 45 hours at 1200°C. C; heat treatment B samples crushed to powder recompacted and heated an additional 100 hours at 1000°C.

TABLE 11

X-RAY DIFFRACTION PATTERNS FOR Y2Be2SiO7

mi t	1-	Bartram <sup>(1)</sup>						
This	I/I <sub>1</sub>	dA	1/11	hk1				
dA	-/ -1	<b></b>	- <b>.</b> - <b>L</b>					
5.12	2	5.19	25	110				
4.74	17	4.80	35	001				
3.63	2	3.67	8	200				
3.48	20	3.51	16	111				
3.25	20	3.28	100	210				
2.88	30	2.906	35	201				
2.69	100	2.702	95	211				
2.567	5	2.590	18	220				
2.376	5	2.391	16	002				
2.301	13	2.315	50	310				
2.261	8	2.276	35	221				
2.159	8	2.172	12	301				
2.071	13	2.083	30	311				
2.016	2	2.029	8	320				
1.989	3	2.000	< 5	202				
1.919	30	1.929	25	212				
1.819	7 .	1.829	8	400				
1.765	11	1.774	20	410				
1.748	1	1.757	< 5	222				
1.715	11	1.725	60	330				
1.700	9	1.707	35	302				
1.653	28	1.662	25	312				
1.627	5	1.627	18	420				
1.614	15	1.621	35	331				
1.585	3	1.592	14	003				
1.424	11	1.431	10	510				
1.391	4	1.396	4	332				
1.364	1	1.371	6	511				
1.350	3	1.357	18	223, 422				
1.300	10	1.307	16	313				

<sup>(1)</sup> ASTM powder data file #22-110

TABLE 12 COMPARISON OF X-RAY DIFFRACTION DATA OF SOME Y-Si-Be-O-N PHASES

$\frac{Y_4Si_2O_7N_2}{}$ (1)			rved tern	Y <sub>2</sub> BeO <sub>4</sub> (2)				
d	I/Io	d	1/10	d	I	hk1		
7.3	15	7.4	10	7.21	14	101		
				5.22	12	200		
				4.621	7	201		
4.57	15	4.53	23	4.484	71	102		
4.35	10	(4.33)	*					
3.22	20	3.20	27	3.152	24	103		
3.07	100	3:00	100	2.927	100	210		
2.93	15	(2.90)	*					
2.87	33	2.87	30	2.844	36	302		
				2.807	16	21.1		
2.84	50	2.85	60	2.791	36	203		
		2.82	50	2.776	34	112		
		2.58	10	2.605	10	400		
2.53	15	2.53	23(b)	2.477	6	004		
2.51	10	·		2.411	10	013,311		
2.27	10	2.26	7	2.237	9	204		
				2.219	15	312		
2.05	10	2.01	20	2.095	14	410		
				1.980	17	313		

<sup>(1)</sup> After Lange(2) After Bartram

 $Y_2Be_2SiO_7$  line

terms of high and low temperature compatibility relationships involving the  $Y_2BeO_4$  phase, which has already been demonstrated to be unstable at low temperatures (Ref. 51). It is less straight-forward to interpret the data if one assigns the new diffraction pattern to a YAM phase having the composition  $Y_4BeSiO_7$ . In light of these observations, tentative phase relations for the system are suggested to be as shown in Fig. 26. The lower temperature of stability of the  $Y_2BeO_4$  phase is not specified.

#### b) The Quinary Systems

Composition, fabrication, and test data for Y-Si-Be-O-N samples are found distributed in Appendicies B, C, and D. Compositions, recalculated from Appendix B in terms of equivalent percent Be, Y, and O, and phase compositions calculated on the basis of the XRD data, are presented in Table 13. Compatibility relations inferred from the data of Table 13, in conjunction with the already established phase diagrams for the  $\mathrm{Si_3N_4-Be_3N_2-BeO-SiO_2}$  and  $\mathrm{Si_3N_4-SiO_2-Y_2O_3}$  systems, permit the establishment of several of the compatibility tetrahedra in the quinery system. These are displayed in the model photographed in Fig. 27. These compatibility tetrahedra are listed in Table 14.

#### .3. Properties of Test Samples

#### a) <u>Oxidation</u>

Oxidation data for Y-Si-Be-O-N bodies are plotted as functions of compositions using the previously described techniques in Fig. 28. These oxidation rate data may be compared with those in the analogous region of Y-Al-Si diagram, Fig. 12. (Note the difference in scale between Figs. 12 and 28). Oxidation rates in the Be system are seen to be an order of magnitude lower than in the Al system. The absence of liquid in the oxide scales of the Be samples is undoubtedly an important factor in the lower rates. Note that sample 1126 (point falling in the  $SiO_2 + Y_2Si_2O_7 + Y_2Be_2SiO_7$  field) meets the program goals as far as oxidation rate is concerned.

#### b) Mechanical Properties

Mechanical properties were measured for samples of composition 1126. The mean room temperature MOR was 434 MPa (63 ksi). The creep sample exhibits a high creep rate of 5.7 x  $10^{-3}$  hr<sup>-1</sup> at  $1370^{\circ}$ C and 69 MPa stress, and failed after eight hours. A second creep sample was heated in air at  $1370^{\circ}$ C for 200 hours prior to testing. This sample failed minutes after the 69 MPa load was applied.

#### c) Summary of Y-Si-Be-O-N Data

Only a small number of samples in the Y-Si-Be-O-N system have been investigated, and large regions of the system remain to be explored. Samples of composition 0.5 e/o Be, 3.0 e/o Y and 4.4 e/o 0 exhibited a low oxidation rate, but a high creep rate at  $1370^{\circ}$ C make this composition unacceptable for high stress applications.

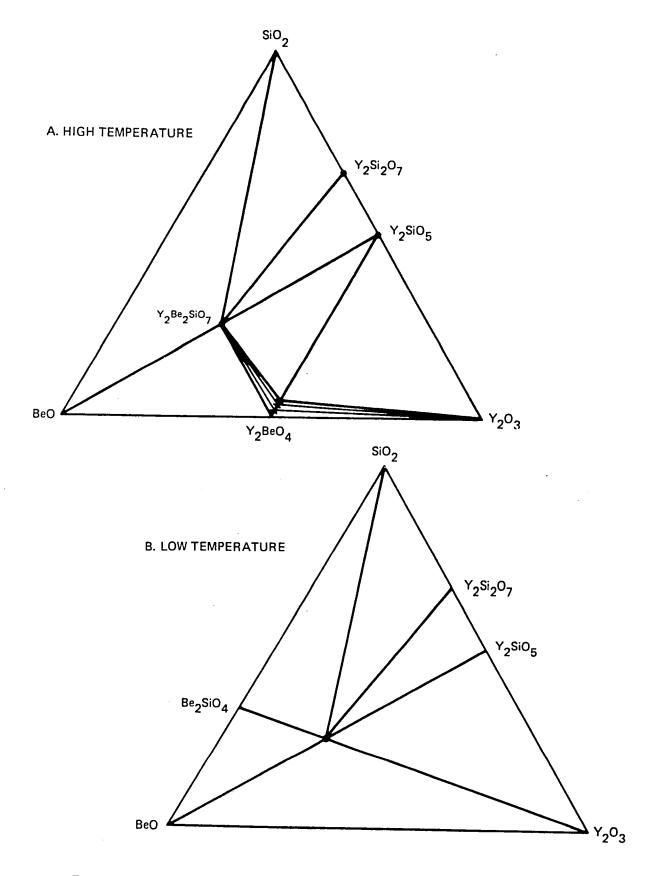


Figure 26 Solid Phase Compatibility Relations in the BeO-Y<sub>2</sub>O<sub>3</sub>-SiO<sub>2</sub> System

TABLE 13

PHASE COMPOSITIONS OF SAMPLES IN THE Y-Si-Be-O-N SYSTEM

						Ca	Other			
Composition Designation (Appendix B)	Equiva: Be	lent Pe	ercent 0	XRD Observations	<u>Symbol</u>	βSi <sub>3-</sub> : _m/o	x <sup>Be</sup> x <sup>0</sup> 2x <sup>N</sup> 4-2x <u>x</u>	12 <sup>31</sup> 3-y <sup>10</sup>	y <sup>0</sup> 3+2y <sup>N</sup> 4-2y y	m/o Phase
(Аррения 27	_			sβ + wM <sup>(1)</sup>	o	85.0	0	15.0	1.9	0.0
1083	4.7	7.3	16.7		•	97.8	0.303	2.2	2.0	0.0
1084	5.9	1.2	13.0	sβ	0	95.0	0	5.0	2.0	0.0
1113	1.7	2.4	5.8	sβ + wM		94.1	0.095	0.0		5.9 Y2Si2O7
1117	1.5	2.9	9.5	sβ + w YS <sub>2</sub>	Λ	95.0	0	4.3	0.69	0.7 Y5Si3012N
1126	0.5	3.0	4.4			50	0	50	2.0	0.0
Si3N4-Y2Be2Si07	15.4	23.1	53.8	sβ + s M	o	30	·	86.7	0.692	$13.7  Y_5  ^{Si}  _3  ^{O}  _{12}  ^{N}$
Si3N4-"Y4BeSi09'		40	60	s M + w 2.86 Å (H Phase?)						

64

<sup>(</sup>L) M = melilite phase,  $Y_2Si_{3-y}Be_y^0_{3+2y}^{0}_{4-2y}$ 

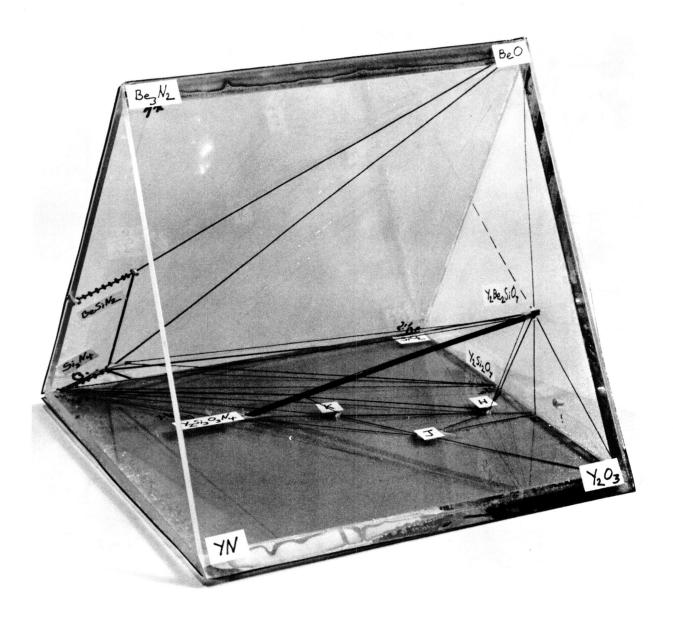


Figure 27 Model of the Y-Si-Be-O-N System

TABLE 14

#### PARTIAL LIST OF COMPATIBILITY TETRAHEDRA IN THE Y-Si-Be-O-N SYSTEM

3 Phase

4 Phase

1) 
$$\beta_{0}^{*} - M^{**}_{0 \le y \le .7} - K$$

2)

$$\beta_0 - M_y = .7 - K - H$$

3) 
$$\beta_0 - M_{.7} \le y \le 2^{-H}$$

4)

$$\beta_0 - M_y = 2^{-H - YS_2}$$

5) 
$$\beta_0 \le x \le .8 \quad y = 2 \quad YS_2$$

6)  $\beta_0 \le x \le .8 - \text{Si}_2 N_2^0 - YS_2$ 

7)

$$\beta_{x} = .8 - Si_{2}N_{2}0 - YS_{2} - Si_{2}$$

8)

$$\beta_{x} = .8 - YS_2 - SiO_2 - M_y = 2$$

9)

$$\beta_{x = .8} - SiO_2 - M_{y = 2} - BeO$$

\* 
$$\beta = Si_{3-x} Be_{x}^{0} 0_{2x}^{N} 4-2x$$

<sup>\*\*</sup>  $M = Y_2Si_3 - y^Be_y^0_{3+2y}^{N_4-2y}$  (melilite ss)

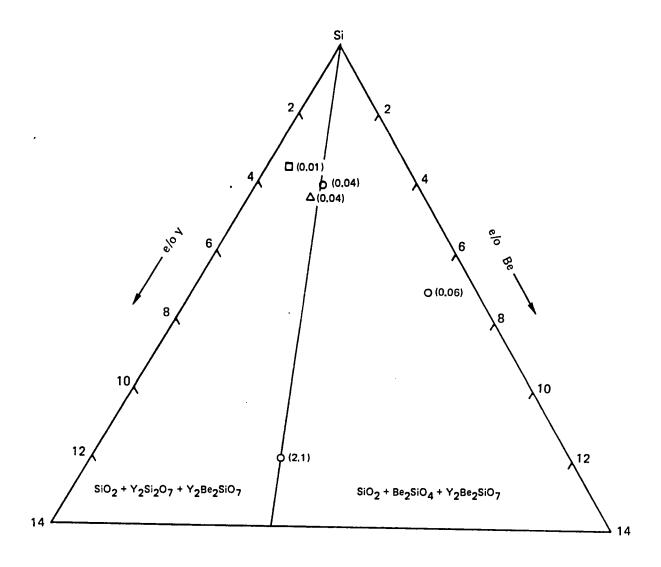


Figure 28 Oxidation Rate Constants (mg<sup>2</sup> cm<sup>-4</sup> hr<sup>-1</sup>) vs. Composition for Y-Si-Be-O-N Sample

#### E. R-Si-O-N Systems

### 1. The System Si<sub>3</sub>N<sub>4</sub>-Si<sub>0</sub>2-Y<sub>2</sub>0<sub>3</sub>

The solid phase compatibility relationships based on the experimental work of Rae et al. (20) Wills et al. (39) and Lange et al. (40) were presented in Fig. 6. In the course of the present work some liquid-solid equilibria data were obtained, as these are relevant to the sintering of Si<sub>3</sub>N<sub>4</sub> using Y<sub>2</sub>O<sub>3</sub> as the sintering aid. These data are presented in Appendix E, and are interpreted in terms of the phase diagram for the join Y<sub>2</sub>SiO<sub>5</sub>-Y<sub>2</sub>Si<sub>3</sub>O<sub>3</sub>N<sub>4</sub> in Fig. 29. Note that all compositions from about 10 m/o to 50 m/o Y<sub>2</sub>Si<sub>3</sub>O<sub>3</sub>N<sub>4</sub> are liquid at 1750°C. Compositions along the join cooled from above the solidus retain some glass at room temperature, as seen in Figs. 4 through 8 of Appendix E.

Understanding of the solid-liquid-phase relationships along the  $Y_2Si_3O_3N_4$  join helps to resolve some of the conflicts in the published phase diagrams for the system  $Si_3N_4-SiO_2-Y_2O_3$  (Ref. 20, 39, 40). Firstly, the failure of Wills et al. (Ref. 39) to find the compound  $YSiO_2N$  (K phase) is understandable in terms of the incongruencey of K phase and tendency of the liquids to glass formation.

The composition of H phase was given as  $Y_{10}Si_2O_{23}N_4$  by Lange et al. (Ref. 40) and as  $Y_5Si_3O_{12}N$ , by Wills et al. and initially by Jack (Ref. 56) who noted the similarity of the diffraction pattern of H phase to that of  $Y_4Ca(SiO_4)_3O$  which has the apatite structure. Later work by Jack's group (Ref. 20) has assigned the homogeniety range of the H phase to a region near the composition ( $Y_4Si$ ) ( $Si_3O_{11}N$ )N, or  $Y_4Si_4O_{11}N_2$ . Regarding the composition of H phase, our work confirms the assignment of Wills et al., and the earlier assignment of Jack, and shows essentially single phase material at the composition  $Y_5Si_3O_{12}N$ . This does not of course rule out the possibility that solid solutions occur between the latter composition and the  $Y_4Si_4O_{11}N_2$  composition. As is shown by Fig. 6 of Appendix E, compositions richer in the component  $Y_2Si_3O_3N_4$  than H phase can cool from temperatures above  $1700^{\circ}C$  to H phase plus glass. On the basis of x-ray data only such compositions would be assumed to be H phase.

# 2. The System Si<sub>3</sub>N<sub>4</sub>-Si<sub>02</sub>-Nd<sub>2</sub>O<sub>3</sub>

It was assumed that phases isomorphous to those in the  $\mathrm{Si}_3\mathrm{N}_4-\mathrm{Si0}_2-\mathrm{Y}_2\mathrm{O}_3$  system would form in this system, and that analogous liquid-solid relationships would also exist. The generally lower liquidus temperatures in the  $\mathrm{Nd}_2\mathrm{O}_3-\mathrm{SiO}_2$  system compared to those in the  $\mathrm{Y}_2\mathrm{O}_3-\mathrm{SiO}_2$  system suggested that ternary liquids in the  $\mathrm{Nd}_2\mathrm{O}_3-\mathrm{SiO}_2-\mathrm{Si}_3\mathrm{N}_4$  (or liquids involving mixtures of  $\mathrm{Nd}_2\mathrm{O}_3$  and  $\mathrm{Y}_2\mathrm{O}_3$ ) would be more reactive than the analogous  $\mathrm{Y}_2\mathrm{O}_3$  liquids. Two compositions along the  $\mathrm{Nd}_2\mathrm{Si}_2\mathrm{O}_7-\mathrm{Nd}_2\mathrm{Si}_3\mathrm{O}_3\mathrm{N}_4$  join were investigated, namely

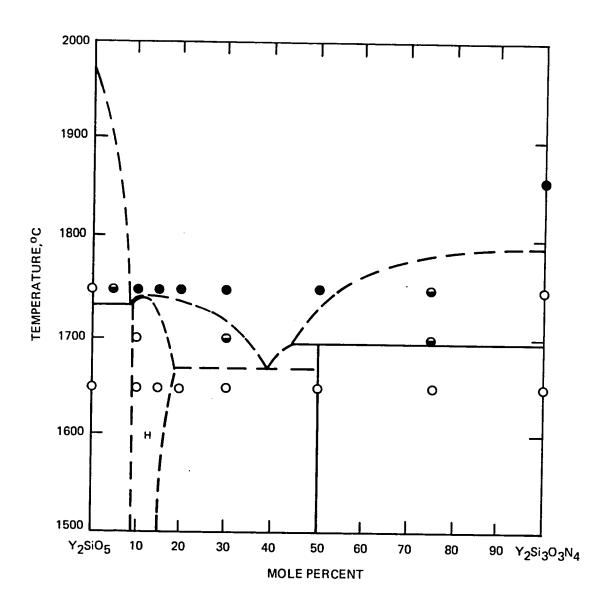


Figure 29 Phase Diagram for the System  $Y_2SiO_5 - Y_2Si_3O_3N_4$ 

20 and 50 m/o  ${\rm Nd_2Si_30_3N_4}$ . Data for these compositions is included in Appendix E. Both the H and K phases were found to occur in the system, and their XRD data are included in Appendix E.

As was the case with compositions in the yttria system, so in the neodymia system large areas of matrix liquid failed to crystallize on cooling from  $1750^{\circ}$ C, and remained as glass at room temperature. Composition  $NdSiO_2N$  cooled from 1750 and  $1800^{\circ}$ C contained no primary crystals, but exhibited dendritic structures. The x-ray pattern showed poorly crystallized H phase and  $Nd_2Si_3O_3N_4$  which is probably, but not necessarily, indicative of incongruency of the K phase in this system also.

### 3. Reaction of Ternary Liquids with Si3N4

Samples consisting of different ratios of KBI  $\rm Si_3N_4$  and various ternary mixtures that are known from the above studies to be liquid at  $1750^{\rm o}{\rm C}$  were pelletized, and fired at  $1750^{\rm o}{\rm C}$  for 1 hour. These were then tested for apparent porosity, specific gravity, and bulk density according to the ASTM C373-5 water immersion test. A portion of each pellet was crushed for x-ray analysis and a portion used for metallographic examination. Results are presented in Table 15. Metallographs of sections of samples 1054 and 1055 are shown in Fig. 30.

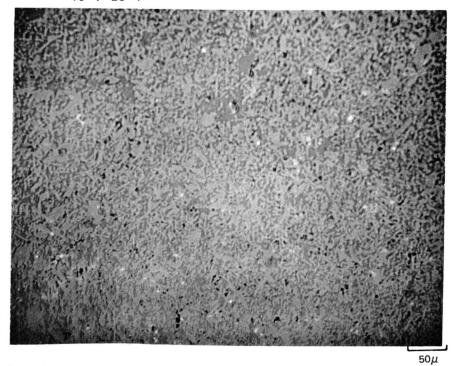
It can be seen that in all cases of 50 w/o mixtures, the bodies fired to high density. Metallography indicates that microstructures consisted of a high volume percent matrix (largely H phase) and discontinuous  $\rm Si_3N_4$  crystals. The 25 percent mixtures however were all quite porous (about 25 percent) although a substantial amount of liquid was present at the firing temperature. This suggested that the liquids do not wet-out the solid phase effectively.

# 4. Sintering and Property Data for some R-Si-O-N Bodies

Composition, fabrication, and property data for some R-Si-O-N bodies, abstracted from appendices B, C, and D are presented in Table 16. Note that compositions prepared using the amorphous  $\mathrm{Si_3N_4}(\mathrm{SN402})$  in general sintered to higher density than did compositions prepared with the KBI powder. However, even using the SN402, a composition 94 m/o  $\mathrm{Si_3N_4}$  6 m/o  $\mathrm{Y_2Si_2O_7}$  sintered to only about 80 percent of theoretical density, although SN402 mixed with 6 m/o  $\mathrm{Ce_2Si_2O_7}$ , or  $\mathrm{GdSmSi_2O_7}$ , sintered to near theoretical density. The latter bodies, although well sintered, exhibited very high oxidation rates that would prohibit their use in air of  $\mathrm{1370^{\circ}C}$ .

Sample	Compos	sition (W/O)	Apparent Porosity	Bulk Density	Specific Gravity	Phases observed by XRD					
Number	$\frac{\text{Si}_3\text{N}_4}{}$	Other	(%)	g/cc	g/cc g/cc 		H	J ——	K	<u>M</u>	
1050	50	50 NdSiO <sub>2</sub> N	1.0	3.57	3.61	m	s			s	
1051	75	25 NdSiO <sub>2</sub> N	27	2.32	3.16	s			m		
1054	50	50 "Y <sub>10</sub> Si <sub>7</sub> O <sub>23</sub> N	1.8	3.93	4.24	m	s				
1055	<b>7</b> 5	25 "Y <sub>10</sub> Si <sub>7</sub> O <sub>23</sub> N <sub>4</sub>	" 26	2.35	3.18	s	m				
1057	50	50 "Nd <sub>10</sub> Si <sub>7</sub> O <sub>23</sub> N	4" 2	4.41	4.53	w	s				
1058	75	25 "Nd <sub>10</sub> Si <sub>7</sub> O <sub>23</sub> N	<b>4"</b> 20	2.63	3.29	m			s		

A. 50 w/o  $Y_{10}Si_7O_{23}N_4$  (SAMPLE 1054)



B.  $25 \text{ w/o} \text{Y}_{10} \text{Si}_7 \text{O}_{23} \text{N}_4$  ( SAMPLE 1055 )

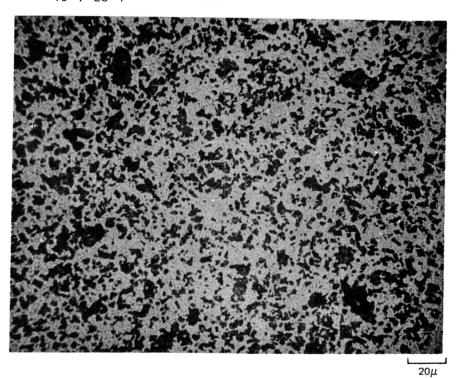


Figure 30 Etched Sections of Samples Containing Si<sub>3</sub>N<sub>4</sub> and Y<sub>10</sub>Si<sub>7</sub>O<sub>23</sub>N<sub>4</sub> in Different Proportions fired to 1750°C

77-12-44-5

TABLE 16
Fabrication and Property Data for Some R-Si -O-N Bodies

		Calcula	ted Phase Composition (	m/o)		ring itions	Apparent Porosity	Bulk Density	Specific Gravity	Oxidat Rate Cor	~
	Sample Number	Si <sub>3</sub> N <sub>4</sub>	R <sub>2</sub> Si <sub>2</sub> O <sub>7</sub>	Si <sub>2</sub> N <sub>2</sub> O	Temp (°C)	Time (hrs)	(%)	(g/cc)	(g/cc)	(mg <sup>2</sup> cm <sup>-</sup> 1370°C	
	995	90 кві	10 Y <sub>2</sub> Si <sub>2</sub> O <sub>7</sub>	0	1720	1	35	2.18	3.40		
	996	90 KBI	10 Y <sub>2</sub> Si <sub>2</sub> O <sub>7</sub>	0	1760	1	37	2.37	3.40		
	1116A	63 KBI 31 SN402	6 Y <sub>2</sub> Si <sub>2</sub> O <sub>7</sub>	0	1760	2	29	2.32	3.25		
	1116в	94 SN402	6 Y <sub>2</sub> Si <sub>2</sub> O <sub>7</sub>	0	1760	2	14	2.64	3.08		
73	1119	94 SN402	6 Ce <sub>2</sub> Si <sub>2</sub> O <sub>7</sub>	0	1760	2	0	3.44	3.44		0.3
ω	1129	94 KBI	6 Ce <sub>2</sub> Si <sub>2</sub> O <sub>7</sub>	0	1760	2	<1	3.44	3.44	0.75	
	1130	94 Starck	6 Ce <sub>2</sub> Si <sub>2</sub> O <sub>7</sub>	0	1760	2	<1	3.05	3.05	0.50	
	969	90 KBI	5.5 Sm <sub>2</sub> Si <sub>2</sub> O <sub>7</sub>	4.5	1780	4	38	2.15	3.31		
	1003	90 KBI	5.5 (Gd,Sm,Nd) <sub>2</sub> Si <sub>2</sub> O <sub>7</sub>	4.5	1760	3	14	2.93	3.39		
	1135	94 SN402	6 (Gd,Sm) <sub>2</sub> SiO <sub>7</sub>	0	1700	5	<1	3.29	3.29	1.6	
	1136	94 SN402	6 Nd <sub>2</sub> Si <sub>2</sub> O <sub>7</sub>	0	1700	5	<1	3.36	3.36	1.2	

## 5. Sintering and Property Data for the System Y-Ce-Si-O-N

### a) Sample Formulation and Fabrication

The facts that Ce-Si-O-N bodies sinter to full density, but exhibit poor oxidation behavior, whereas Y-Si-O-N bodies do not sinter to high density, but hot pressed bodies of the nonsinterable compositions can exhibit outstanding properties, lead to the speculation that bodies in the combined Y-Ce-Si-O-N system could afford a viable compromise between sinterability and performance. This question was addressed in the following experiments:

Compositions within the basic formulation  $aSi_3N_4 + bY_{2-x}Ce_xO_3 + cSiO_2$  were prepared using SN402 Si<sub>3</sub>N<sub>4</sub>, calcined yttrium and cercium oxylates, (Y<sub>2</sub>O<sub>3</sub> and CeO<sub>2</sub>) and Apache-0.03µSiO<sub>2</sub>. In these formulations the SiO<sub>2</sub> content of the Si<sub>3</sub>N<sub>4</sub> powder was not taken into consideration, and it was assumed that the reaction given below occurred during firing, as reported by Rae et al. (Ref. 20):

$$\text{Si}_3\text{N}_4 + 12\text{CeO}_2 \rightarrow 3\text{SiO}_2 + 6\text{Ce}_2\text{O}_3 + 2\text{N}_2.$$

(This procedure was simply for valence bookkeeping.) Compositions prepared are listed in Table 17. The powder batches were processed as described in section IIIA2b, without binder, cold pressed, and fired as described in Sections IIIA3, and IIIA5. No calcining step prior to firing was used. Firing conditions, and property data on the test bars, are listed in Table 18. Samples for mechanical testing were finished on the tensile face, and for oxidation testing were finished on all surface, to 600 grit SiC paper.

# b) <u>Densification</u>

The average bulk densities of samples from a given batch are plotted as a function of composition in Fig. 31. Composition is plotted on a log scale in order to space out the points. Also plotted is the theoretical density of one series of compositions. This was calculated by assuming that the samples consisted of a mechanical mixture of the phase  $Si_3N_4$  having a density of 3.20 g/cc, and the phase  $Y_{2-x}Ce_xSi_2O_7$ . The density of the latter phase was calculated by assuming the density of  $Y_2Si_2O_7$  to be 4.15 g/cc and further assuming the lattice parameter of the phase did not change with substitution of Ce for Y. The theoretical density is seen to be essentially constant for all x < 0.1. There appears to be a systematic deviation from the theoretical value of the experimentally determined density of sintered samples for which .02 < x < 1.33, but in general this deviation is less than about 1.5 percent. This could be indicative of erroneous assumptions regarding the lattice parameters and/or phase constitution of the bodies. (As will be seen later, in fact fired bodies

Composition Number		Coeff	(l icients	)	Weight Percent of Constituents					
	Α ·	В	С	x	Si <sub>3</sub> N <sub>4</sub>	Y <sub>2</sub> 0 <sub>3</sub>	CeO <sub>2</sub>	SiO <sub>2</sub>		
1134	83.93	5.36	10.71	0.3328	85.92	7.34	2.24	4.49		
1137	**	**	11	0.1664	86.16	8.11	1.12	4.61		
1138	**	11	**	0.0832	86.28	8.49	0.56	4.67		
1139	**	**	11	0.0416	86.340	8.683	0.281	4.695		
1140	"	11	**	0.0208	86.370	8.780	0.141	4.709		
1141	tt	11	11	0.0126	86.383	8.829	0.071	4.716		
1142	n	11	11	0.0052	86.393	8.852	0.035	4 <b>.</b> 720		
1145	90.46	4.77	4.77	0.010	90.301	7.611	0.058	2.030		
1146	**	11	**	0.005	90.307	7.631	0.029	2.033		

<sup>(1)</sup> In the formula A  $Si_3N_4$  +  $BY_{2-x}$   $Ce_x$   $O_3$  + C  $SiO_2$ 

TABLE 18

FABRICATION AND TEST DATA For Y-Ce-Si-O-N SAMPLES

Sample	Firing Conditions		Bulk		Strength		1370°C, 10 ksi	Additional Testa		
Numbers	Tempo (°C)	Time (hrs)	Density (g/cc)	Temp °C	kei	Нра	Creep Rate (hr-1)	Additional leats		
1134.1							•	1400°C oxidation		
. 2	1700	5	3.29							
.3			3.29							
1137.1								1400°C exidation		
,2	1700	5	2,33	25	79	543				
.3		-	3.26							
,4		-	2.34	25	70	484				
1138.1								1400°C oxidation		
,2	1700	5	3,42	25	68	468				
.3			3.44							
.4			3.35	25	85	585				
1139.1								1400°C oxidation		
.7	1700	5	3.35	25	66	454				
.3			3,38	25	78	536				
,4				•						
1140.1								1400°C oxidation		
.2	1700	15	3,38	25	60	414		1		
.3			3.40	25	84	576				
.4					•		$3.3 \times 10^{-3}$	1000°C oxidation		
.5 .6	1700	20	3.39 3.29	1370	64	440	3.3 X 10			
.7	1700	20	3.28	1370	52	360		,		
.8			2.29							
.9				25	86	592 440				
.10 .11	1700	10		25 25	64 85	584		1370°C oxidation		
1141.1								1400°C oxidation		
.2	1700	10	3.15	1270	53	370	$4.1 \times 10^{-3}$			
.3 .4			3.18 3.13	1370 1370	50	345				
.5			3.18	•	•					
.6	1700	10	3.23	25	79	545				
.7			3.24	25	74 70	510 485				
.8			3.14	25	70	40,5				
1142.1								1400°C oxidation		
. 2	1700	10	3.03	25	68	470				
.3			2.98	25	71	490		1370°C oxidation		
.4 .5							$4.2 \times 10^{-3}$	1310 1 041-41341		
• • • • • • • • • • • • • • • • • • • •								•		
1145.1			3.24				1.5 x 10 <sup>-4*</sup>	STARRE STATE		
.2	1700	15	3.24	at		550		1400°C oxidation 1000°C oxidation		
.3			3,26 3,26	25 25	80 73	505		PROF A CHIMMITAL		
.4			3.20							
1146.1			3.19							
.2	1700	15	3.19							
.3 .4			3.18 3.17							
, ~			2.1.							

<sup>\*</sup> Average rate between 10th and 15th hour of test. Furnace failure curtailed run. See Text.

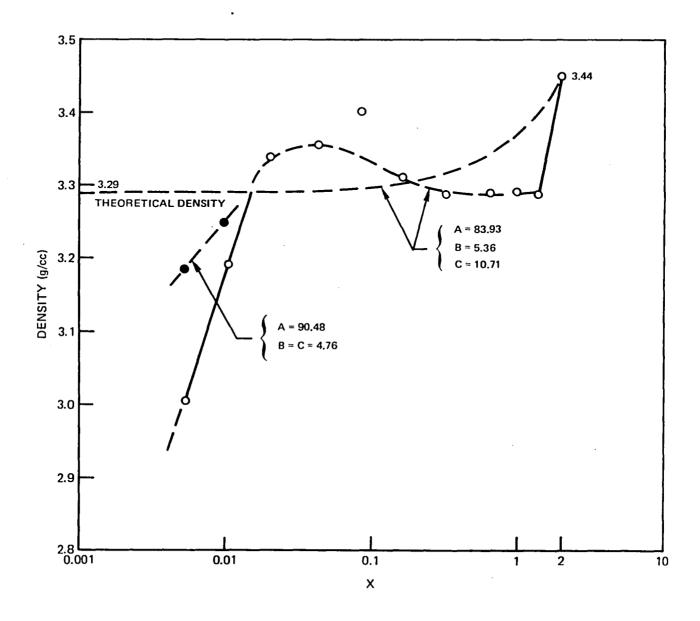


Figure 31 Density of Sintered Bodies of Compositions 94  $Si_3N_4$  6z ( $y_2$ — $_x$ Ce $_x$   $Si_2O_7$ )

contained a glass phase). However, for the values of x < 0.01 there is a significant drop in density with decreasing x. It appears that the value x=0.01 is near the lower limit of x for densely sintered bodies using the given fabrication parameters. This value of x corresponds to a solution in which one in every 200 of the rare earth type cations is cerium.

### c) Oxidation Data

The  $1400^{\circ}$ C parabolic oxidation curves for samples of the various compositions are shown in Fig. 32. Following an initial period of decreasing rate lasting up to about 6 hrs, the oxidation appears to follow parabolic kinetics. The parabolic rate constants calculated from the slopes of the curves of Fig. 32 are plotted against composition on a log-log scale in Fig. 33. Note that the oxidation rate constant decreased monotonically with the variable x for the C = 2B series of compositions for 0.01 < x < 1 (i.e. for all "theoretically dense" samples) but rose for the less dense sample 1.42 (x = .005). The minimum oxidation rate for this composition series (about  $0.03 \text{ mg}^2\text{cm}^{-4}\text{hr}^{-1}$ ) was still a factor of 3 times the target rate of 0.01. XRD from the oxidized surface showed the scales to contain the crystalline phases  $Y_2\text{Si}_2\text{O}_7$  and  $\alpha$ -cristobalite. SEM of the oxidized surface, and metallography of transverse sections through the oxidized sample 1140.1 are shown in Fig. 34, and indicate the presence also of a major glassy phase.

Samples of composition 1145 exhibited the lowest oxidation (0.015  $\mathrm{mg}^2\mathrm{cm}^{-4}$  hr<sup>-1</sup>) which approaches the target value. The scale on sample 1145.1, shown in Fig. 35, is thin, with little glass present. XRD shows the scale to contain  $\mathrm{Y}_2\mathrm{Si}_2\mathrm{O}_7$  and cristobalite, with the former phase showing a strong OkO orientation with respect to the surface.

Samples of compositions 1140 and 1145 tested at  $1000^{\circ}\text{C}$  for 75 hrs showed essentially zero weight gain, and there was no evidence of cracking.

## d) Mechanical Test Data\_

Room temperature flexural strengths for all of the Y-Ce-Si-O-N samples tested ranged from about 420 to 620 MPa (60 to 90 ksi), with an average value for 17, tests of 515 MPa (75 ksi). The average strength at 1370°C for composition 1140 and 1141 samples (4 tests) was 380 MPa (55 ksi). Those room temperature fracture surfaces that were examined by SEM were seen to initiate of processing flaws (voids), exemplified by Fig. 36. 1370°C fracture surfaces exhibited areas of slow crack growth.

The 1370°C, 69 MPa creep rates for several samples are listed in Table 18. All of the samples formulated as  $ASi_3N_4 + BY_{2-x}Ce_xO_3 + 2BSiO_2$  (i.e., 94

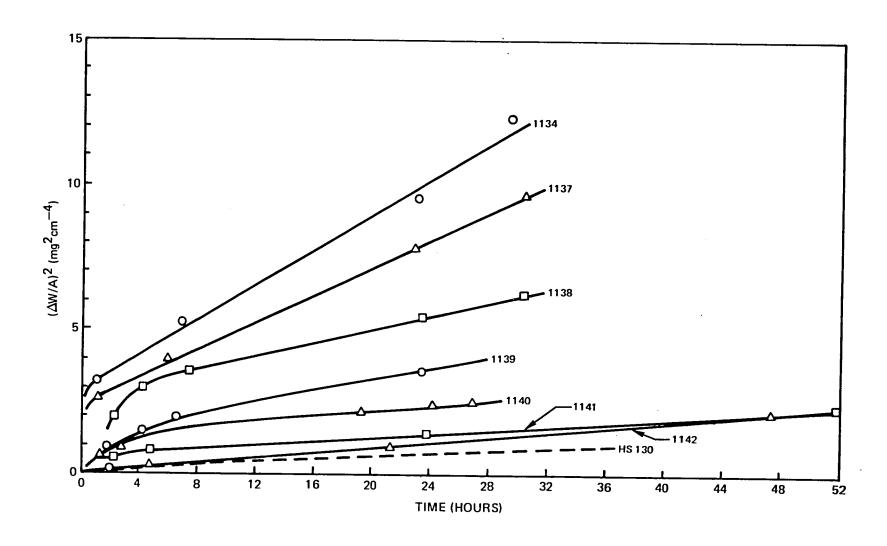


Figure 32 1400 °C Oxidation Curves for Compositions 94 m/o Si<sub>3</sub>N<sub>4</sub> + 6 m/o Y<sub>2—x</sub>Ce<sub>x</sub>Si<sub>2</sub>O<sub>7</sub>

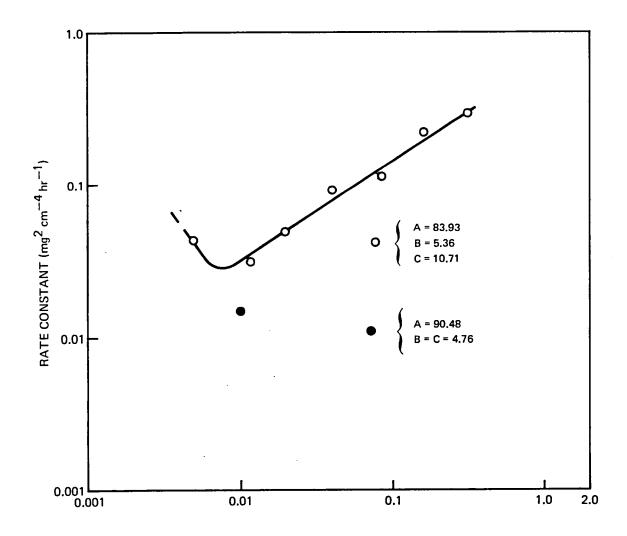
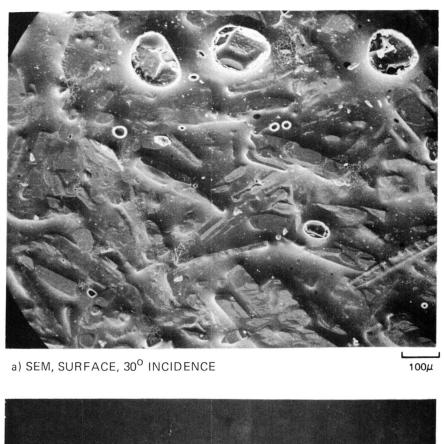
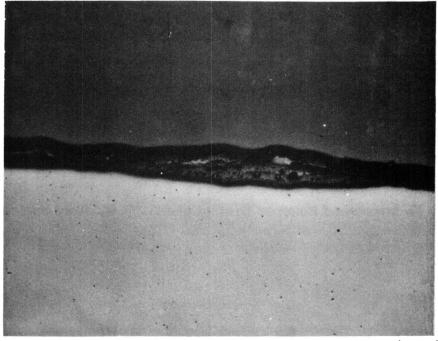


Figure 33 1400 °C Parabolic Oxidation Rate Constants for Compositions 94 m/o Si<sub>3</sub>N<sub>4</sub> 6/6 m/o Y<sub>2—x</sub>Ce<sub>x</sub>Si<sub>2</sub>O<sub>7</sub>





b) TRANSVERSE POLISHED SECTION

100μ

Figure 34 Microscopy of 1400 °C Oxidized Sample 1140:1

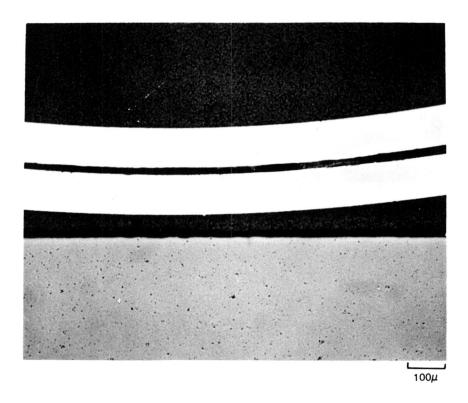
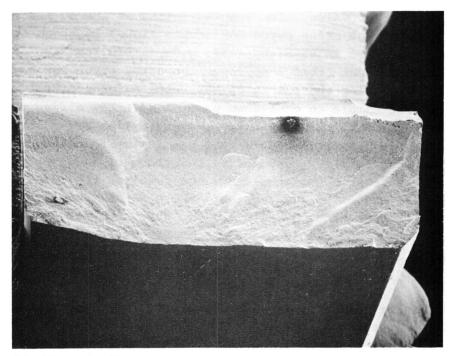


Figure 35 Tranversed Polished Section of 1400 °C Oxidized Sample 1145:2



a) OVERALL (X13)



Figure 36 Fracture Surface of Sample 1140:9

 $\mathrm{Si}_3\mathrm{N}_4+6\mathrm{Y}_{2-\mathrm{x}}\mathrm{Ce}_{\mathrm{x}}\mathrm{Si}_2\mathrm{O}_7)$  are seen to have creep rates about two orders of magnitude greater than the target value. The samples failed at the 69 MPa stress level after about 10 hrs. The creep curve for sample 1140.5 is shown in Fig. 37. Sample 1145.1 formulated as 95  $\mathrm{Si}_3\mathrm{N}_4+5\mathrm{Y}_{1.99}\mathrm{Ce}_{0.01}\mathrm{SiO}_5)$  on the other hand, exhibited quite different creep behavior. The creep curve for this sample is shown in Fig. 38. During the course of the test, a furnace failure occurred which permitted the sample to cool-down under load. The strian rate of the time of furnace shut-down was  $1.67 \times 10^{-4} \ \mathrm{hr}^{-1}$ . Information regarding sample strain during cool-down was not retrieved. After repairs were made, the sample was returned to the test facility, brought to temperature, load reapplied, and the test continued. Note that the sample did not continue the initial secondary creep curve, but instead exhibited a new primary and secondary creep behavior with a secondary creep rate of about  $5 \times 10^{-5} \ \mathrm{hr}^{-1}$ . The samples failed after about 16 hrs of the second run without entering a tertiary creep regime.

## e) Instrumental Analysis of Test Specimens

A rim of higher reflectivity around the exterior region of the fractured surface of creep specimen 1154.2 was noted under the binocular microscope, as shown in Fig. 39A. It was also noted that the external surface of the sample was discolored particularly in the region of contact with the tungsten loading bars. The fracture surface was examined in SEM, Backscatter electron microscopy (Fig. 39B) and EDS spectroscopy (Fig. 39C). No differences could be detected between the interior and exterior zones of the fracture surface using any of the above techniques. Fresh transverse fractures were made in the sample and showed the same characteristics as the original fracture surface.

One half of the specimen was waxed to a plate and thin layer of material removed to expose a clean surface of the lighter zone. An x-ray diffraction pattern was obtained from this surface. More material was then removed to expose a surface of the dark inner region, and again an XRD pattern obtained. Patterns were also obtained from the surfaces of room temperature test specimens of composition 1145, and also from powder obtained by crushing and comminuting room temperature test specimens of compositions 1145 and 1141. The low angle portions of some of the XRD patterns are reproduced in Fig. 40. In general, the patterns obtained from different samples of the powder obtained by crushing a single fired bar were quite variable in terms of the d-spacings and relative intensities of the low intensity minor phase peaks. In no case could the identity of the minor crystalline components be established with certainty. In the case of sample 1141.6 (and other C = 2B compositions that were X-rayed) the minor phase reflections were very faint, and it is assumed that the grain boundary material is largely glass, which could account for the high creep rate at 1370°C. What minor phase reflections were visible were around 3.07 and 2.89 Å, as exemplified by Fig. 40, Trace A.

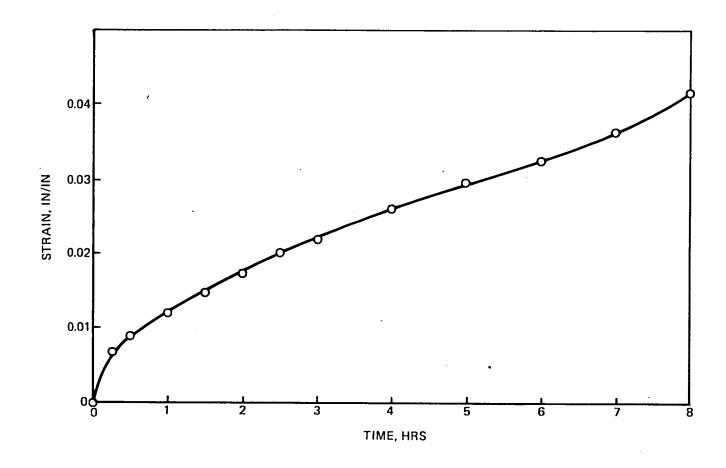


Figure 37 1370 °C, 69 MPG Creep Curve for Sample 1140:5

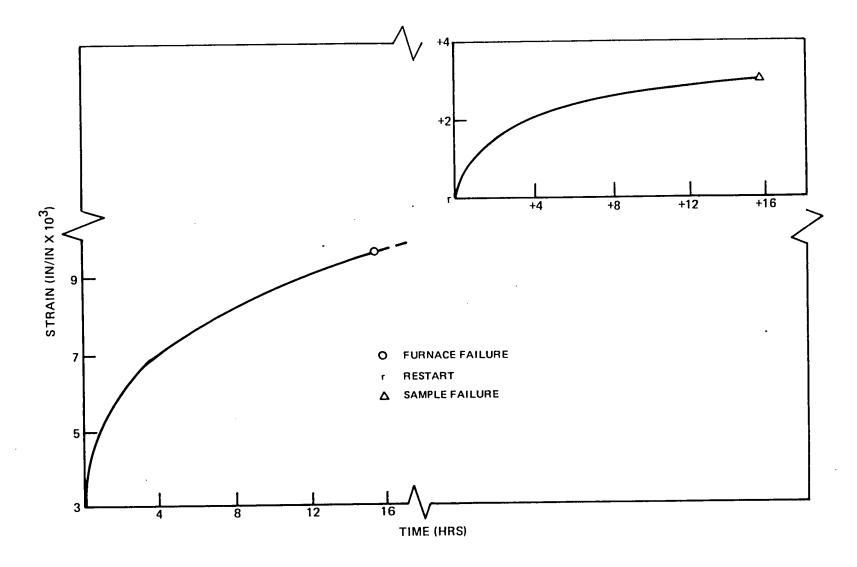
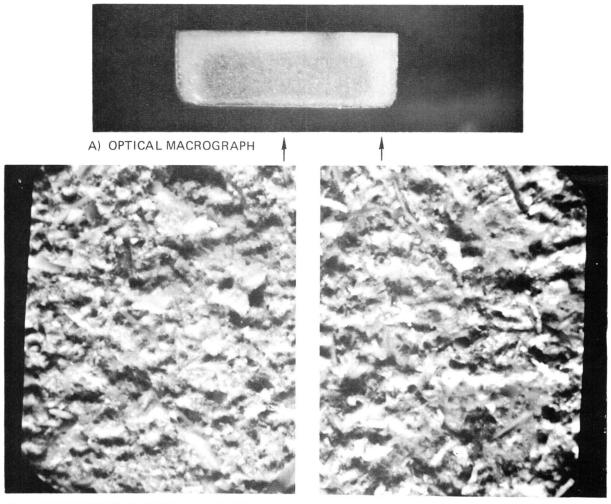
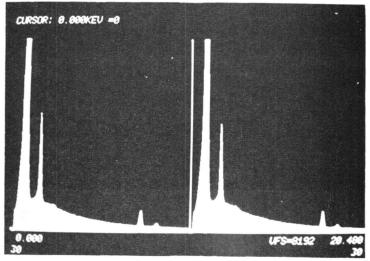


Figure 38 1370°C, 69 MPa Creep Curve for Sample 1145:1



B) BACKSCATERED ELECTRON MICROGRAPHS



C) EDS SPECTRA, INSIDE AND OUTSIDE

Figure 39 Rim Around Creep Specimen 1145:2

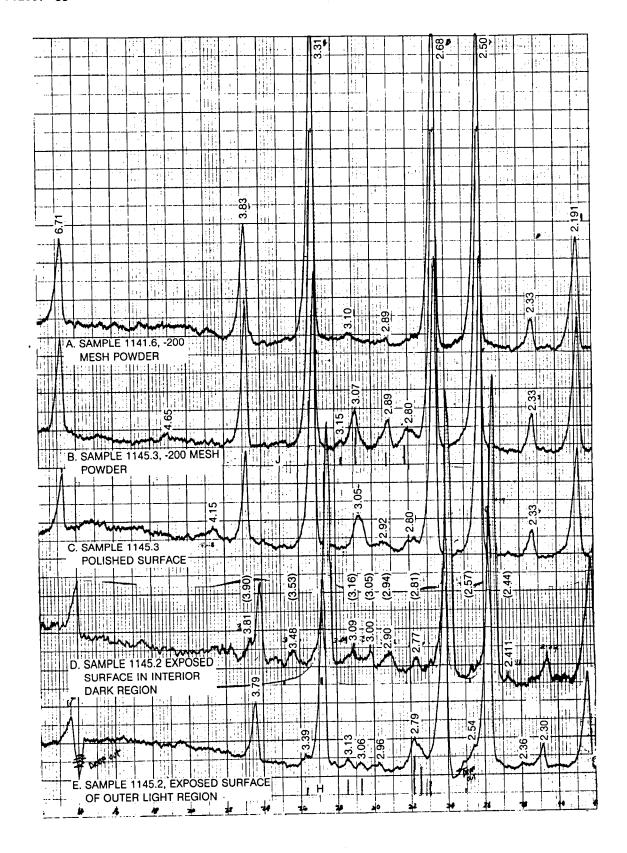


Figure 40 XRD Patterns for Some As-Fired and Heat Treated Y-Ce-Si-O-N Sample

For sample 1145.3 (C = B) which under equilibrium conditions should consist of major  $\mathrm{Si_3N_4}$ , minor H phase, and trace  $\mathrm{Y_2Si_2O_7}$ , intensities of minor phase peaks were substantially greater than for C = 2B samples. The strongest of these were centered around 3.05 to 3.07, 2.89 to 2.92, and 2.77 to 2.80 Å, with quite irreproducible relative intensities. Traces B and C are representative of the variability. Conceivable such variability could result from a non-uniform mixture of several of the  $\mathrm{Y_2Si_2O_7}$  modifications, perhaps with the additional presence of J phase. It also could perhaps result from the presence of poorly crystallized and/or highly stressed quench crystals. Whatever the true nature of the crystalline component in the grain boundaries, it is clearly a metastable assemblage.

After the  $1370^{\circ}\text{C}$  heat treatment in argon (i.e., the creep test), quite different XRD patterns were observed from surfaces exposed by grinding into the dark inner region, and the light outer region of the sample, as shown in Traces D and E, respectively. The outer region now clearly contains the (presumably) equilibrium H and  $Y_2\text{Si}_2\text{O}_7$  phases.

The patterns from these surfaces apparently are displaced. What appears to be a large lattice shift for the  $\mathrm{Si}_3\mathrm{N}_4$  is probably the spurious result of sample misalignment. When the sample was crushed to powder and the XRD pattern obtained, lattice parameters were within experimental error of those of the non-heat treated samples. The shifts are too great reasonably to be attributed to elastic strain in the heat treated sample, which would require a compressive stress on the order of 2000 MPa. Shifting Pattern D the indicated amount results in  $\underline{d}$  values for the minor phase reflections shown in parenthesis above the original values. Taking into account the shifted  $\underline{d}$  values, trace D indicates a new phase superimposed on the variable minor phase pattern in the original samples. This appears to be the  $\mathrm{X}_2$ - $\mathrm{Y}_2\mathrm{SiO}_5$  phase, although again such an identification cannot be considered as highly reliable.

While the creep sample showed evidence of phase transformations in the grain boundaries during 1370°C heating under stress in argon atmosphere, the samples which were heated in air at 1370°C for 100 hours gave no evidence of grain boundary transformation (save for minimal surface oxidation). The above results lead one to suspect that the metastability of non-equilibrium grain boundary phases, at least in some  $\text{Si}_3\text{N}_4\text{-R}_2\text{O}_3\text{-SiO}_2$  bodies can be radically altered by environment and/or stress. Further development of  $\text{S}_3\text{N}_4$  bodies from derivations of the  $\text{Si}_3\text{N}_4\text{-Y}_2\text{O}_3\text{-SiO}_2$  system must pay particular attention to optimizing grain boundary composition and heat treatments to insure long term stability under anticipated stress and environment.

#### REFERENCES

- 1. Jack, K. H. and W. I. Wilson: Ceramics Based on the SiAlON and Related Systems, Nature Phys. Sci., pp. 238, 8028, 1972.
- 2. Oyama, Y. and O. Kamigaito: Solid Solutility of Some Oxides in  $Si_3N_4$ . Japan J. Appl. Phys. <u>10</u>, p. 1637, 1972.
- Morgan, P. E. D.: Paper 4-52-74, American Ceramic Society Meeting, April 1974. Also, J. Amer. Ceramic Society, pp. 53, 392, 1974.
- Land, P. L., et al: Paper 68-B-75, American Ceramic Society Meeting, Washington, D. C., May 6, 1975, and J. Amer. Ceramic Society, pp. 61, 56, 1978.
- 5. Gauckler, L. J., et al: Paper 67-B-75, American Ceramic Society Meeting, Washington, D. C., May 6, 1975. Also, J. American Ceramic Society, pp. 58, 346, 1975.
- 6. Layden, G. K.: Process Development for Pressureless Sintering of SiAlON Ceramic Components, Final Report, Naval Air System Command Contract N00019-75-C-0232, February 1976.
- Jack, K. H.: SiAlONs and Related Nitrogen Ceramics, J. Mat. Sci. <u>11</u>, pp. 1135-1158, 1976.
- 8. Jack, K. H.: Trans. Brit. Ceram. Soc. 7, p. 376, 1972.
- 9. Huseby, I. C., H. L. Lucas and G. Petzow: J. Amer. Ceramic Soc., <u>58</u>, p. 377, 1975.
- Gauckler, L. J., H. L. Lucas and T. Y. Tien: Mat. Res. Bull., <u>11</u>, p. 503, 1976.
- 11. Layden, G. K.: Pressureless Sintering of SiAlON Gas Turbine Components, Report Number NADC-75207-30, February 23, 1977.
- 12. Layden, G. K.: Final Report, Contract N62269-77-C-0248 for Naval Air Development Center, Warminster, PA, p. 18974, January 27, 1979.
- 13. Mazdiyasni, K. S. and C. M. Cooke: American Ceramic Society Bulletin, 53, p. 392, 1974, abstract.

#### REFERENCES (Cont'd)

- 14. Wills, R. R.: Journal of the American Ceramic Society 57, 1974.
- 15. Lange, F. F.: Final Report, Contract N00019-73-C-0208, February 26, 1974.
- 16. Rice, R. W. and W. J. McDonough: American Ceramic Society Bulletin 54, p. 406, 1975, abstract.
- 17. Terwilliger, G. R. and F. F. Lange: Pressureless Sintering of  ${\rm Si}_3{\rm N}_4$ , J. Mat. Sci.  $\underline{10}$ , pp. 1169-1174, 1975.
- 18. Gazza, G. E: Effect of Y<sub>2</sub>O<sub>3</sub> Additions on Hot-Pressed Si<sub>3</sub>N<sub>4</sub>, Bulletin of the American Ceramic Society <u>54</u>, pp. 778-781, 1975.
- 19. Venables, J. D., D. K. McNamara, and R. G. Lye: High Strength Silicon Nitride Prepared with Eutectic Flux Additions, F. L. Riley, Editor, Nitrogen Ceramics, Proceedings of the N.A.T.O. Advanced Study Institute, Canterbury, August 1976.
- 20. Rae, A.W.J.M., D. P. Thompson, and K. H. Jack: The Role of Additives in the Densification of Nitrogen Ceramics, presented the NATO Advanced Study Institute on Nitrogen Ceramics, Canberbury, England, August 1976.
- Layden, G. K.: Development of SiA10N Materials, NASA CR 135290, December 1977.
- Layden, G. K. and S. Holmquist: Ninth Quarterly Progress Report on NASA Control NAS3-19712, February 1978.
- 23. Juza, R., A. Rabenau and I. Nitschke: Z. Anorg. Allg. Chem., p. 332, 1964, 1-4.
- 24. Aramaki, S. and R. Roy: J. Am. Ceram. Soc., <u>42</u>, (12) p. 644, 1959.
- 25. Welch, J. W.: Nature. 186, (4724), p. 546, 1960.
- 26. Geller, R. F. and S. M. Lang: National Bureau of Standards. 1959, Phase Diagram for Ceramists, Fig. 361.
- 27. Butterman, W. C. and W. R. Foster: Am. Mineralogist, <u>52</u>, p. 884, 1967.
- 28. Lejus, A.: Rev. Hautes Temper. et Refract. t. 1, p. 53, 1964.

#### REFERENCES (Cont'd)

- 29. Gilles, J. C.: Corrosion Anti Corrosion. 12, (1), p. 15, 1964.
- 30. Claussen, N. E., R. Wagner, L. J. Gauckler and G. Petzow: Ceramic Bulletin, 56, #3, p. 301, 1977, abstract.
- 31. Kordyuk, R. A. and N. Y. Gullko: Dokl. Akad. Nauk. SSSR, p. 154, (5), p. 1183, 1964.
- 32. Herold, P. G. and W. J. Smothers: J. Am. Ceram. Soc., <u>37</u>, (8), p. 353, 1954.
- 33. Sorrell, C. A. and C. C. Sorrell: J. Am. Ceram. Soc., <u>60</u>, (11), p. 495, 1977.
- 34. Olds, L. E. and H. E. Otto: In Phase Diagram for Ceramists. The Am. Ceram. Soc., Fig. 311, 1964.
- 35. Toropov, I. A. et al.: <u>ibid</u>, Fig. 2344.
- 36. Noguchi, T. and M. Mizuno: ibid, Fig. 4370.
- 37. Toropov: ibid, Fig. 2388.
- 38. Bondar and Galakov: ibid, Fig.
- 39. Wills, R. R. et al.: J. Mat. Sci., 11, p. 1305, 1976.
- 40. Lange, F. F., S. C. Singhal and R. C. Kuzniki: J. Am. Ceram. Soc., <u>60</u>, p. 249, 1977.
- 41. Holmquist, S.: Report, U. S. Air Force Contract F33615-C-4155.
- 42. Tien, T. Y.: Technical Report # 014536-78-1, Contract No. AFOSR-76-3078.
- 43. Arias, A.: NASA Technical Paper, p. 1249, 1978.
- 44. Dutte, S.: NASA Technical Memorandum, p. 78950, 1977.
- 45. Bacon, J.: Final Report Contract NASW-1301 and NASW-2013, January 1971.
- 46. Weaver, G. Q. and J. W. Lucek: Am. Ceram. Soc. Bull. <u>57</u>, p. 1131, 1978.

### REFERENCES (Cont'd)

- 47. Brennan, J. J.: Third Quarterly Progress Report Contract NAS3-21375, April 1979.
- 48. Morgan, R. A. and F. A. Hummel: J. Am. Ceram. Soc., <u>32</u>, 1949.
- 49. Olds, L. E., and H. E. Otto: In Phase Diagrams for Ceramists, Am. Ceram. Soc., Fig. 311, 1964.
- 50. Bartram, S. F.: G. E. Co. Report DC 61-7-10, (see. Ref. 53).
- 51. Wier, C. E., and A. Van Valkenburg: J. Res. N.B.S., 64A, p. 103, 1959.
- 52. Potter, R. A.: Met. Div. Rept. ORNL-2988, p. 180, Oak Ridge Labs, 1965.
- 53. Harris, L. A. and H. L. Yakel: Acta Cryst. 22, p. 354, 1967.
- 54. Bartram, S. F.: Acta Cryst. 1325, p. 791, 1969.
- 55. Huseby, I. C., H. L. Lukas and G. Petzow: J. Am. Ceram. Soc. <u>58</u>, p. 377, 1975.
- 56. Jack, K. H.: J. Mat. Sci. <u>11</u>, p. 1135, 1976.

			-
			•
			-

APPENDIX A COMPOSITIONS INVESTIGATED IN THE YSLAION SYSTEM, AND TEST RESULTS

Join	<u>Mo 1</u> <u>Y</u>	lecu Sí	lar Al		ula N	Y202	Co S1 <sub>3</sub> N <sub>4</sub>	mpone 510		Aln	Firing Temp C	Conditions time hrs	Shrinkaş (7)	Phases Present	
Y4A1209 - Y4S1207N4		1			1	2	0	1	0	1	1750	1	5		Observations
Y 11 0			_			8	1	1	2	0	1750	1	5	Yam Yam <sub>ss</sub>	
Y3 <sup>N1</sup> 5 <sup>0</sup> 12 - "Y3 <sup>S15</sup> <sup>0</sup> 7 <sup>N</sup> 5"	3		5 3		0 2	3 3	0	1	5 3	0	1550 1600	1	2 15	YAG	
											1550	1	5	sYAG, w 3.03, 2.87, 2.79, 2.74	
	3	1	4	11	1	3	0	2	3	2	1500 1475	1	3		Bloated
						6	1	1	8	٥	1525 1600	1	6	≢YAG, ⊌ H	Melted
						·	•	•	•	Ü	1475 1525 1600	1	4	sYAG, w H	
	3	2	3	10	4	3	1	1	3	0	1475 1525	1 1 1	4	SYAG, w J. w H	Melted
•	6	5	5	19	5	3	0	5	0	5	1600 1475	1	2	HIAG, W J, W H	Melted
											1525 1600	1	2	sYAG, m H	Melted
						12	5	5	10	0	1475 1525	1 1	2 2	SYAG, m H, w J	Herced
	3	4	1	8	4	3	2	2	1	0	1600 1475	1 1	0		Melted
		_									1550 1600	1	4 17	all, m YAG, w a Si3N4	
	3	3	2	9	2	6	3	3	4	0	1475 1525	1	2 3	syac, m H, w J	
\$/41 0 N - V 0					,				_		1600	1			Melted
SIA1202N2 - Y203	2 .	,	6	9	6	1	1	0	2	2	1500 1600	1	0		
								3	•		1625 1700	1	15	w YAG, w, E	Bloated
							1	,	0	6	1575 1625 1650	1	-2	sYAG, m 2.81, ξ, c sYAG, tr β' tr ε	
											1700 1725	1 1 1	5 4	SYAG	
	1 (	6 1	12	15	12	1	0	6	a	12	1625	1	-3	mYAG syag, w β¹, tr ξ, ε	Melted '
	6 (	6 1	12	21	12	3	O	6	0	12	1625	1	-2	sYAG. w Y2Si303N4, tr c	
S1A1402N4 - Y203	2 :	3 1	.2	9	12	1	0	3	0	12	1725	1	1		
											1750	2	1	SYAC, S C	
						1	1	0	2	8	1725 1750	1 2	8 10	syac, s c	
SIA1202N2 - Y2SI303N4	2 6	5	6	9	10	1	2,	0	2	2	1600	2	1		
						1	1	2	•	,	1625 1700	1	1 17	s YAC, m β'	
						•	•	2	0	6	1600 1625 1700	2 1 1	1 1 16	s YAG, m ß'	
SIA1202N2 - Y4S1207N2	8 13	3 1	8	32	22	12	13	0	20	14	1425 1550	7	15	s YAG, many v.w.	
											1650 1700	1 2	15		Bloated
						4	1	10	0	18	1425 1550	7 1	7 0	s YAG, m 2.8, many v.w.	2102124
											1700	1			Foamed
SIA1402N4 - Y2S1303N4	2 6	1	2	9 ;	16	1	1	3	0	12	1650 1700	1 1	0 1	YAG, m Y <sub>2</sub> 51 <sub>3</sub> 0 <sub>3</sub> N <sub>4</sub> , m c	
						1	2	0	2	8	1650 1700	1	10	YAG, m Y <sub>2</sub> 51 <sub>3</sub> 0 <sub>3</sub> N <sub>4</sub> , m c 1 c, E	
Y251303N4 - AIN	2 3			3	8	1	1	0	0	4	1650 1750	1	0	Y <sub>2</sub> S1 <sub>3</sub> 0 <sub>3</sub> N <sub>4</sub> , AlN	
	2 3		2	3	6	1	1	0	0	2	1650 1750	i 1	U	Y <sub>2</sub> Si <sub>3</sub> O <sub>3</sub> N <sub>4</sub> , AlN	
S1A1202N2 - Y2S102N2	2 5	. ,	6 1	0	8	3	5	0	7	4	1600	1	2	-23-3-4,	White
						4	3	11	2	20	1650 1600	1 1	15 -3		Black Crey
		_		_							1650	1	12		Black
si <sub>3</sub> <sup>A1</sup> 6 <sup>0</sup> 12 <sup>N</sup> 2 - Y2 <sup>0</sup> 3	2 6	1.	2 2	7	4	1	1	3	6	0	1325 1400	1	17		
						1	Ō	6	4	4	1325 1400 1500	1 1 1	3 11		Bloated
S1A1402N4 - Y4A1209	4 6	21	6 2	1 2	4	2	0	6	1	24	1650	1	-2		Bloated
4-2-4 4-2-9	. "	•		- 4							1700	i	0	<b>≡YAG,</b> m ⊢, w 2.81	
						2	2	0	Ś	16	1650 1700	1 1	7	sYAC, m c,w 2.81	
										7.					
SIA1402N4 - Y3A15012	6 9	41	, 4	2 3	· · ·	i	0	9	5	36	1650 1700	1	0 -1	8 YAG, m C	
						6	3	0	11	24	1650 1700	1	2 4	s YAG, m + j ω ℓ	

•			
			-
			`

## APPENDIX B

BATCH COMPOSITIONS FOR VARIOUS TEST SAMPLES

TABLE 5
BATCH COMPOSITIONS OF TEST BARS

Batch	Weight Percent of Components									
Number	Si <sub>3</sub> N <sub>4</sub>	sio <sub>2</sub>	Y <sub>2</sub> 0 <sub>3</sub>	A1203	AlN					
1078	67.98	5.82	25.00	1.19	-					
1079	79.94	2.82	12.12	5.12	-					
1080	65.24	7.45	14.00	9.49	3.82					
1081	52.06	7.43	13.97	18.92	7.61					

Table 7
Batch Compositions

# Batch Constituents (w/o)

Comp.	Molecular Composition	Si <sub>3</sub> N <sub>1</sub>	<u>Al<sub>2</sub>03</u>	Aln	Be <sub>2</sub> SiO <sub>4</sub>	<u>Y<sub>2</sub>03</u>
1082	Si <sub>5.42</sub> Al.29 <sup>Be</sup> .29 <sup>0</sup> .87 <sup>N</sup> 7.13	88.74	3.73	1.5	6.03	0
1083	Si <sub>2.71</sub> Be <sub>.29</sub> 0.58 <sup>N</sup> 3.42 <sup>+.15Y</sup> 203	71.46	0	0	9.36	19.18
1084	Si <sub>2.66</sub> Be <sub>.34</sub> 0.68 <sup>N</sup> 3.32 <sup>+.022Y</sup> 203	81.84	0	0	13.16	5.00

TABLE 2

COMPOSITIONS STUDIED

	Compositi	on (mole per	cent)	Cor	stituent	s (Weight	Percent	
Batch No.	β or β'	Y3 <sup>A1</sup> 5 <sup>0</sup> 12	Y <sub>2</sub> Si <sub>2</sub> 0 <sub>7</sub>	Si3N4	Si0 <sub>2</sub>	$\frac{\text{A1}_2^0_3}{}$	AlN	$\frac{^{4}2^{0}3}{}$
1085	95 Si <sub>3</sub> N <sub>4</sub>	5	. 0	81.78	0	7.82	0	10.40
, 1086	95 Si <sub>2</sub> A10N <sub>3</sub>	5	0	54.24	0	27.49	7.92	10.34
1087	95 SiA1 <sub>2</sub> 0 <sub>2</sub> N <sub>2</sub>	5	0	26.98	.0	46.96	15.77	10.29
1088	95 Si <sub>3</sub> N <sub>4</sub>	4	1	83.04	0.75	6.36	0	9.85
1089	95 Si <sub>3</sub> N <sub>4</sub>	3	2	84.25	1.53	4.87	0	9.35
1090	95 Si <sub>3</sub> N <sub>4</sub>	2	3	85.60	2.33	3.30	0	8.77
1091	95 Si <sub>3</sub> N <sub>4</sub> + 2.5 SiAl <sub>4</sub>	0 <sub>2</sub> N <sub>4</sub> + 2.5 Y	2 <sup>Si</sup> 3 <sup>0</sup> 3 <sup>N</sup> 4	62.82	0	22.83	10.56	3.79

TABLE 1

COMPOSITIONS OF TEST SAMPLES INVESTIGATED

			Weight	Percent	of Const	ituents	1)				
Composition Number	Intended Phase Composition (m/o)	KBI KBI	3N <sub>4</sub> SN402	<u>\$10</u> 2	<u>A1203</u>	<u>A1N</u>	Y <sub>2</sub> 0 <sub>3</sub>	<u>Other</u>	Grindin Al <sub>2</sub> 0 <sub>3</sub>	g Media <u>Si<sub>3</sub>N</u> 4	Phases Observed by x-rays
1103	95 Si <sub>3</sub> N <sub>4</sub> , 2.5 Y <sub>3</sub> Al <sub>5</sub> O <sub>12</sub> , 2.5 Y <sub>2</sub> Si <sub>2</sub> O <sub>7</sub>	82.85	0	2.02	6.36	0	8.77	0	x		
1104	97.5 Si <sub>2</sub> A10N <sub>3</sub> , 2.5 Y <sub>3</sub> A1 <sub>5</sub> 0 <sub>12</sub>	58.36	0	1.52	24.73	9.99	5.40	0	x		s β°
1105	95 S1 <sub>2</sub> A10N <sub>3</sub> , 2.5 S1A1 <sub>4</sub> 0 <sub>2</sub> N <sub>4</sub> , 2.5 Y <sub>2</sub> S1 <sub>3</sub> 0 <sub>3</sub> N <sub>4</sub>	62.42	0	1.61	20.21	12.00	3.77	0	×		s β', w 15R
1105A	95 S1 <sub>2</sub> A10N <sub>3</sub> , 2.5 S1A1 <sub>4</sub> 0 <sub>2</sub> N <sub>4</sub> , 2.5 Y <sub>2</sub> S1 <sub>3</sub> <sup>2</sup> 0 <sub>3</sub> N <sub>4</sub>	63.25	0	0.39	20.45	12.14	3.81	0		x	s β', w 15R
1111	95 Si <sub>3</sub> N <sub>4</sub> , 2Y <sub>3</sub> Al <sub>5</sub> 0 <sub>12</sub> , 3 Y <sub>2</sub> Si <sub>2</sub> 0 <sub>7</sub>	81.62	0	2.29	3.65	0	12.44	0	x		
1112	$\begin{array}{cccccccccccccccccccccccccccccccccccc$	87.69	0	3.57	4.33	0	4.41	0	x		
1113	95 Si <sub>3</sub> N <sub>4</sub> , 5 Y <sub>2</sub> Be <sub>2</sub> Si0 <sub>7</sub>	87.03	0	1.78	0	0	0	11.19 Y <sub>2</sub> Be <sub>2</sub>	S10 <sub>7</sub>	×	s β' w Y <sub>2</sub> Be <sub>2</sub> S10 <sub>7</sub>
1114	95 Si <sub>2.5</sub> Al <sub>.5</sub> 0.5 <sup>N</sup> 3.5 2.5 Y Si <sub>3</sub> 03 <sup>N</sup> <sub>4</sub> 2.5 SiAl <sub>4</sub> 0 <sub>2</sub> N <sub>4</sub>	78.33	0	1.58	12.53	3.60	3.96	0		x	
1115	$^{81.4}_{16.1}  ^{Si}_{S1A1}  ^{A1}_{40}  ^{50}_{2N} .  ^{5N}_{3.5}$ $^{2.5}_{2}  ^{Y}_{2}  ^{Si}_{30}  ^{3N}_{3}  ^{4}$	64.72	0	1.31	15.7	14.68	3.59	0		x	sβ', w 2H <sup>δ</sup>
1116	94 Si <sub>3</sub> N <sub>4</sub> , 6 Y <sub>2</sub> Si <sub>2</sub> O <sub>7</sub>	54.49	27.52	6.36	3.06	0	8.57	0	ж		
1116A	94 Si <sub>3</sub> N <sub>4</sub> , 6 Y <sub>2</sub> Si <sub>2</sub> O <sub>7</sub>	56.38	28.72	6.10	0	0	8.79	0		x	
1116в	94 Si <sub>3</sub> N <sub>4</sub> , 6 Y <sub>2</sub> Si <sub>2</sub> O <sub>7</sub>		85.76	5.50	0	0	8.74	0		x	· .
1116C	94 Si <sub>3</sub> N <sub>4</sub> , 6 Y <sub>2</sub> Si <sub>2</sub> O <sub>7</sub>		84.07	5.39	0	0	8.57	2 carbowa	x	×	
1117	85 Si <sub>3</sub> N <sub>4</sub> , 9Si <sub>2</sub> N <sub>2</sub> O,	54.34	27.69	5.88	0	0	8.47	1.45 BeO 2.17 carbo	owax	x	
1118	Y <sub>2</sub> Si <sub>3</sub> O <sub>7</sub> , 4Y <sub>2</sub> Be <sub>2</sub> SiO <sub>9</sub> 80 Si <sub>3</sub> N <sub>4</sub> , 17.5 Si <sub>3</sub> Al <sub>6</sub> N <sub>10</sub> 2.5 Y <sub>2</sub> O <sub>3</sub>	72.68		1.46	0	22.86	3.00			x	

<sup>(1)</sup> Values listed include contribution from grinding media wear, and  ${\rm Si0}_2$  impurity in  ${\rm Si}_3{\rm N}_4$ 

Table 1 (Cont)

Composition of Test Samples

		Weight Percent of Constituents								
Composition Number	Intended Phase Composition, (m/o)	Si3N4 (1)	<u>si</u> 0 <sub>2</sub>	<u>Y</u> 203	<u>Ce0</u> 2	<u>A1<sub>2</sub>0</u> <sub>3</sub>	BeO			
1119	94 Si <sub>3</sub> N <sub>4</sub> , 6 Ce <sub>2</sub> Si <sub>2</sub> O <sub>7</sub>	83.64	3.40	0	12.96	0	0			
1120	94.5 Si <sub>3</sub> N <sub>4</sub> , 4.5 Ce <sub>2</sub> Si <sub>2</sub> O <sub>7</sub> ,	83.26	2.25	0	12.90	1.59	0			
	1 Ce <sub>3</sub> Al <sub>5</sub> O <sub>12</sub>									
1122	94 Si <sub>3</sub> N <sub>4</sub> , 5 Ce <sub>2</sub> Si <sub>2</sub> O <sub>7</sub>	83.70	3.20	0	12.97	0	0.31			
	1 Ce <sub>2</sub> Be <sub>2</sub> SiO <sub>7</sub>			,						
1123	94 Si <sub>3</sub> N <sub>4</sub> , 3 Ce <sub>2</sub> SiO <sub>7</sub>	83.80	3.61	0	12.98	0	1.50			
	3 Ce <sub>2</sub> Be <sub>2</sub> SiO <sub>7</sub>									
1124	94 Si <sub>3</sub> N <sub>4</sub> , 1 Ce <sub>2</sub> Si <sub>2</sub> O <sub>7</sub>	83.91	1.51	0	13.00	0	1.57			
	5 Ce <sub>2</sub> Be <sub>2</sub> SiO <sub>7</sub>									
1125	97 Si <sub>3</sub> N <sub>4</sub> , 3 Ce <sub>4</sub> BeSiO <sub>9</sub>	86.53	0	0	13.00	0	0.47			
1126	97 Si <sub>3</sub> N <sub>4</sub> , 3 Y <sub>4</sub> BeSiO <sub>9</sub>	89.42	1.18	8.90	0	0	0.49			

(1) SN402

Table 1 (Cont)

Compositions of Test Samples

Composition	Intended Phase		Weight Pe	ercent of	Constituents	5
Number	Composition, (m/o)	$\underline{\text{Si}_{3}\text{N}_{4}}(1)$	S10 <sub>2</sub>	<u>Y203</u>	CeO <sub>2</sub>	Other
1129	94Si <sub>3</sub> N <sub>4</sub> , 6Ce <sub>2</sub> Si <sub>2</sub> O <sub>7</sub>	83.64(2)	3.40	0	12.96	
1130	94Si <sub>3</sub> N <sub>4</sub> , 6Ce <sub>2</sub> Si <sub>2</sub> O <sub>7</sub>	83.64(3)	3.40	0	12.96	
1131	94Si <sub>3</sub> N <sub>4</sub> , 6CeYSi <sub>2</sub> O <sub>7</sub>	85.99	2.92	4.39	6.70	
1132	94Si <sub>3</sub> N <sub>4</sub> , 6Ce <sub>1.33</sub> Y <sub>0.67</sub> Si <sub>2</sub> O <sub>7</sub>	84.54	3.82	2.88	8.76	
1133	94Si <sub>3</sub> N <sub>4</sub> , 6Ce <sub>0.67</sub> Y <sub>1.33</sub> Si <sub>2</sub> O <sub>7</sub>	85.46	4.27	5.83	4.44	
1134	94Si <sub>3</sub> N <sub>4</sub> , 6Ce <sub>0.33</sub> Y <sub>1.67</sub> Si <sub>2</sub> O <sub>7</sub>	85.92	4.49	7.34	2.24	
1135	94Si <sub>3</sub> N <sub>4</sub> , 6 GdSm Si <sub>2</sub> O <sub>7</sub>	82.22	4.49	0	0	13.29 GdSmO <sub>3</sub>
1136	94Si <sub>3</sub> N <sub>4</sub> , 6 NdSi <sub>2</sub> O <sub>7</sub>	82.82	4.53	0	0	12.68 Nd <sub>2</sub> 0 <sub>3</sub>
1137	94Si <sub>3</sub> N <sub>4</sub> , 6Ce <sub>0.17</sub> Y <sub>1.83</sub> O <sub>7</sub>	86.16	4.61	8.11	1.12	
1138	94Si <sub>3</sub> N <sub>4</sub> , 6Ce <sub>0.08</sub> Y <sub>1.92</sub> O <sub>7</sub>	86.28	4.67	12.98	0.86	

<sup>1)</sup>  $SN_4O_2$  unless otherwise noted

<sup>2)</sup> KBI high purity grade

<sup>3)</sup> Stark hot pressing grade

			-
			•
			•
			-
			-

# APPENDIX C

FABRICATION AND TEST DATA FOR VARIOUS TEST SAMPLES

TABLE 6
Fabrication and Test Data for SiAlON Compositions

### Average Density Data

Batch Number	Sample Numbers	Firing Co	Tomp (OC)	Apparent Porosity (40)	Specific Gravity (kgm <sup>-3</sup> x 10 <sup>-3</sup> )	Bulk Density (kgm <sup>-3</sup> x 10 <sup>-3</sup> )	Phases Observed by X-Ray Diffraction
1078 1079 1080	(1,2,3	1	1700	0.2	3.53	3.51	Surface: $s\beta$ , $s\delta$ $Y_2Si_2O_7$ . About 0.02 cm below surface: $s\beta$ , $wK$ + $w$ 3.0 Å. Crushed sample, $s\beta$ + $m$ $J_{SS}$
1078	4,5,6	2	1765	0.2	3.55	3.54	
	7,8,9	1	1790	1	3.48	3.44	
	1,2,3	1	1725	6	3.39	3.25	β' only
C 1079	4,5,6	2	1765	0	3.39	3.38	
	7,8,9	2	1770	1	3.36	3.33	
	1,2,3	2	1750	bloated,	large internal v	oids	β' only
1080	4,5,6	1	1580	0	3.56	3.56	sβ' + w αSi <sub>3</sub> N <sub>4</sub>
	7,8,9	1	1525	0	3.36	3.36	
	1,2,3	2	1710	bloated,	large internal v	oids	β' only
1081	4,5,6	2	1650	bloated,	large internal v	oids	
	7,8,9	1	1550	0	3.24	3.24	

TABLE 3

SAMPLE FIRING AND CHARACTERIZATION DATA

	Firing Con	ditions	<u>D</u>	ensity Data	<u>a</u>		
Sample <u>Number</u>	T (°C)	Time (hrs)	Apparent Porosity (%)	Bulk Density (g/cc)	Specific Gravity (g/cc)	<u>Notes</u>	Minor Phases
1085.1 1085.2)	1725	2	7 <1	3.01 3.14	2.80 3.14	refired 64 hrs at 1400°C	W40
1085.3 1085.4	1715	2	<1 <1	3.22 3.18	3.23 3.19	refired of mrs at 1400°C	YAG glass
1086.1	1725	2	1	3.11	3,13		
.2	1760	2	1 1	3.13 3.17	3.16 3.19	refired	YAG SiAl <sub>4</sub> O <sub>2</sub> N <sub>4</sub>
1087.1	1775	2				melted reduced	
.3	1725	2	17 17	2.67 2.63	3.23 3.18	bloated   refired	
1088.1	1725	2	<1 <1	3.23 3.21	3.24 3.22	refired	glass glass
1089.1	1725	2	1 1	2.70 2.71	2.72 2.73	refired	glass glass
1090.1	1725	2	1 1	2.96 2.83	2.98 . 2.91	refired	glass glass
1091.1	1725	2					Č

#### SAMPLE FABRICATION AND TEST DATA

Sample Number	Standard Pressing Conditions Yes No	Firing Co	Time (hrs)	Firing Weight Loss (%)	Apparent Porosity (%)	Bulk Density (g/cc)	Specific Gravity (g/cc)	Heat Atm	Temp (°C)	Time (hrs)	Surface <u>Finish</u>	Flexural Strength (ksi) (mPa)(avg mPa)	Additional Tests
1103.1 .2 .3 .4	<b>√</b>	1760	2		0.4 0.3 0.2 0.4	3.25 3.28 3.24 3.22	3.26 3.29 3.24 3.23				Linde A Linde A		1400° Oxidation 1300° Oxidation
1104.1 2 .3 .4	<b>√</b>	1760 <sub>.</sub>	2		0.5 0.6 0.6 0.8	3.12 3.11 3.11 3.12	3.13 3.13 3.12 3.14				Linde A Linde A		1400° Oxidation 1300° Oxidation
.5 .6 .7	<b>√</b>	1680	2		0.0 0.0 0.0 0.0	3.29 3.26 3.28 3.26	3.29 3.26 3.28 3.26					• .	
1105.1 .2 .3 .3 .4 .4	\ \ \	1760	2		0.2 0.3 0.7 0.4	3.21 3.21 3.23 3.20	3.22 3.22 3.25 3.21		·		Linde A Linde A Linde A		1400° Oxidation 1300° Oxidation Low Temperature Oxidat
.6 .7 .8 .9	\ \ \	1760	2				}	air	1300	100	240 Grit	38.4 265 48.1 332 57.7 398 44.5 307 49.5 341	
.10 .11 .12 .13 .14	\/ \/	1760	2				}				240 Grit	50.6 349 60.2 415 34.4 237 41.7 287 53.0 366	
.16 .17 .18 .19	\/\	1760	2				}	air	1370	100	240 Grit	33.5 231 27.5 190 35.9 247 28.7 198 42.8 295	
.21 .22 .23	\ <u>\</u>	1760	2		0.0 0.0 0.3	3.24 3.22 3.21	3.24 3.22 3.21				Linde A	45.2 311 52.7 363 46.4 320 331	1370° Creep

Table 2
Sample Fabrication and Test Data

	Ch 1 1	<b></b> .				mpre rabit	acton and 1	est Data						
Sample	Standard Pressing	Firing Max	Conditions	Firing	Apparent	Bulk	Specific	Heat	Treatme		Cfaaa	E1.		
Number	Conditions Yes No	Temp_	Time	Weight Loss	Porosity	Density	Gravity	Atm	Temp		Surface Finish		exural ength	Additional Tests
	Yes No	(°C)	(hrs)	(%)	(%)	(g/cc)	(g/cc)	-	(°C)	(hrs)			(mPa) (avg mPa)	
1111.1				1.54	0.0	3.40	3.40							1400° Oxidation
.2 .3	{ <	1760	2		0.7	3.37	3.35							1400 Uxidation
	)			1.48	0.0	3.37	3.37							
1112.1	) ,													1400° Oxidation
.2 .3 .4	<b>\</b>	17/0		3.78	0.0	3.07	3.07						•	1400 Oxidation
.3 1	}	1760	2	3.82	0.0	3.08 3.09	3.08							•
,				3.02	0.0	3.09	3.09							
1113.1	} ✓	1760	2											1400° Oxidation
1114.1														
.2	\\ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \	1760	2											
.3	}			1.67 1.57	0.0	3.19	3.19							1400° Oxidation
•	,			1.37	0.0	3.19	3.19							
1115.1			_	0.93							Linde A	38	259	
.2	} *	1760	2	0.80 0.70	0.0	3.09	3.09					48	327	1400° Oxidation
.4 )				1.12	0.0	3.11	3.11						327	
- \														
.5 .6	,	1760	2		0.0	3.01	3.01							1000° Oxidation
.,,		1,00	•		0.0	3.08 3.07	3.08 3.07							
_ 1	./						-10.							
.8 .9	}	1760	2		0.0	3.04	3.04		1070			35	240	
.10	)				0.0	3.17 3.11	3.17 3.11	air	1370	100	240 grit	26 25	181 171	
1115A.1	)			0.92	0	3.13	3.13				Linde A	34	235	*****
.3 (	( V											41	284	1400° Oxidation
.4	Binder	1760	2	1.12	0.0	3.15	3.15						204	
.6	)				0.3 0.0	3.19 3.17	3.20 3.17							1000° Oxidation
,					•••	3,1,	3.17							
1116.1					0.1	3.25	3.25							1400° Oxidation
.3	<b>/</b>	1760	2		0.1 0.1	3.24 3.29	3.24 3.29							
.4	}				0.0	3.25	3.25							1000° Oxidation
11164 1 1		1760	2		28.5	2 22	2 20							
1116A.1	\ \ \	1700	-		28.2	2.32 2.32	3.25 3.24							
- 1	. ,													
.3	<b>/</b>	1760	5	11.8										
•••		1,00	,	12.5										
1116B.1 .2	. 🗸	1760	2		14.4	2.64	3.08							
.2 )					14.6	2.62	3.07							
.3 }	•	1760	5	18.6	20.0	2.59	3.11							
.4 )				18.4	21.7	2.59	3.15							
.5 .6		1700	12		32.1	2.24	3.30				As Fire	d	:	1400 Oxidation
.6 .7					22.5	2 22	2 20							
• /					32.5	2.22	3.28							
1116C.1 }	/		5	12.3	10.9	2.69	3.02							
.2 }	Binder			13.0	13.5	2.67	3.09							
1117.1				5.35									1	1400° Oxidation
.2 (		1760	4	4.60	0.1	3.11	3.11				Linde A	50	342	
.2 .3 .4	Binder			4.99	0.0 0.0	3.10 3.11	3.10 3.11							
1118.1	/	17/6	,	1.25	43.3	1.87	3.30							
.2	}	1760	4	1.25 1.42	43.1 42.6	1.88 1.84	3.30 3.21							
.3				1.20	42.3	1.86	3.22							
,	•													

Table 2 (Cont)

#### Sample Firing and Test Data

	Firing Con	ditions	Apparent	Bulk	Specific	Surface			
	Max	Time	Porosity	Density	Gravity	Finish	Flexural S		
Sample Number	Temp (°C)	(erH)	(Z)	(g/cc)	(g/cc)		ks i	mP <sub>8</sub>	Additional Tests
			0	3.49	3.49				1400°C Oxidation
1119.1	1760	2	0.03	3.44	3.44	Linde A	52.6	363	
.2	1760	4	0.03	3.42	3.42	Linde A	45.7	315	
• • • • • • • • • • • • • • • • • • • •									
.1						Linde A	53.1	366 339	
1120.2	1760	2	0	3.32	3.32	Linde A	49.2	339	1400°C Oxidation
.3			0	3.32	3.32				
.1			0	3.32	3.32	As Fired			
1123.2	1760	2	0	3.32	3.32	As Fired			1400°C Oxidation
.3									
			_	2 25	3.35	As Fired			
.1	1760	2	0 0	3.35 3.35	3.35	As Fired			1400°C Oxidation
1124.2	1760	2	Ū	3.33	2.22				
••									
.1			0	3.34	3.34	As Fired			
1125.2	1760	2 .	0	3.33	3.33	As Fired			
.3									
.1			1.67	3.05	3.00	400 Grit			1370°C Creep
.2			2,61	3.06	2.98	As Fired			1400°C Oxidation 1000°C Oxidation
1126.3	1760	2		2.98	3.17	As Fired 400 Grit	68.8	474	1000 0 001001100
.4			6.12 6.00	2.98	3.16	400 (1222			
.5 .6			2.25	3.06	3.13	400 Grit	67.9	467	
.7			2.64	3.08	3.16	400 Grit	52.0	358	
						•			
1129.1			0.07	3.48	3.48	As Fired			1370°C Oxidation 1000°C Oxidation
.2	1760	2	0.03	3.39	3.39	As Fired			1000 C Oxidacion
.3									
1130.1									
.2									
.3	1760	2	0.11	3.05	3.05				1370°C Oxidation
.4			0.20	3.05	3.05				
			•	3.32	3,32	As Fired			1400°C Oxidation
1131.1	1700	15	0 0	3.32	3.32	As Fired			1370°C Oxidation
.2 .3	1700	• • •	ŏ	3.32	3.32	I. inde A	92	632	
.4								437	
.5			0	3.25	3.25	Linde A	40 52	276 359	•
.6	1700	5	0	3.25	3.25 3.25	Linde A Linde A	49	341	
.7			0	3.25	1.23	Bluce 4	• •		
1132.1			.09	3.28	3.28	As Fired			1400°C Oxidation
.2	1700	5	.13	3.28	3.28	600 grit			
.3			.15	3.30	3.30	600 grit			1370°C Oxidation
						As Fired			1400°C Oxidation
1133.1	1100	5	0.06	3.29	3.29	600 grit	62	429	
.2	1700	,	0.0	3.28	3.28	600 grit	57	393	•
.,									1100 0 11-11-
1134.1						As Fired		401	1400° Oxidation
. 2	1700	5	0	3.29	3.29	600 grit 600 grit	58 67	401 461	7
.3			0	3.29	3.29	noo grac	"	1	
1115 1						As Fired			1400° Oxidation
1135.1 .2	1700	5	0.03	3.29	3.29	600 grit			
.3	• • • • •	-	0.01	3.22	3.22	600 grit			
						Am Elmad			1400° Oxidation
1136.1			0.09	3.38	3.38	As Fired 600 grit			ATTO CATORITATION
. 2	1700	5	0.08 0.05	3,38	3,33	600 grit			
.3			0.02			••			

Table II
Fabrication and Test Data

Sample	Firing Con	ditions	Bulk	Flexural	Strength				
Numbers		Time (hrs)	Density (g/cc)	ksi	MPa	Additional Tests			
1134.1									
.2	1700	5	3.29			1400°C oxidation			
.3	2,00	,	3.29						
			3.29						
1137.1									
.2	1700	5	2.33	79	543	1400°C oxidation			
.3			3.26		2.0				
.4			2.34	70	484				
1138.1						1400°C oxidation			
.2	1700	5	3.42	68	468	1400 C Oxidation			
.3 .4			3.44						
• 4			3.35	85	585				
1139.1									
.2	1700	5	3.35			1400°C oxidation			
.3	2,00	J	3.38	66	454				
.4			3.30	78	536				
1140.1									
.2	1700	15	3.38	60	414	1400°C oxidation			
.3			3.40	84	576				
.4						1000°C oxidation			
.5			3.39			1000 C oxidation			
.6	1700	20	3.29						
.7 .8		•	3.28						
.9			2.29						
.10	1700	10		86	592				
.11	1700	10		64	440	0			
				85	584	1370°C oxidation			
1141.1			•			1400°C oxidation			
.2	1700	10	3.15			1400°C oxidation			
.3			3.18						
.4			3.13						
.5			3.18						
.6	1700	10	3.23						
.7			3.24						
.8			3.14						
1142.1						_			
.2	1700	10	2.02			1400°C oxidation			
.3	1700	10	3.03 2.98						
.4			4.70			107000			
		•				1370°C oxidation			

			-
		•	
•			•
	•		
			-
			•
			ı

### APPENDIX D

# OXIDATION DATA FOR VARIOUS TEST SAMPLES

Rate Data for Samples Oxidized in Air at Different Temperatures  $1400^{\rm O}\text{C}$ 

Sample Number	t (hr)	$\frac{\Delta W^2}{A^2}$ $(g^2 cm^{-4} \times 10^7)$	$\frac{\Delta W^2}{A^2 t}$ ( $g^2 cm^{-4} hr^{-1} \times 10^7$ )	Average Parabolic Rate Constant $(g^{2}cm^{-4} hr^{-1} \times 10^{7})$
1078.1	1.0 3.0 4.3 5.5 8.5 11.2 26.9 96.6	0.86 3.62 7.15 9.52 13.32 19.55 40.65	0.86 1.62 1.66 1.75 1.58 1.75 1.51 1.57	1.54
ب <b>1079.</b> 1	1.0 3.0 4.3 5.5 8.5 11.2 26.9 96.6	12.5 14.6 25.6 31.8 34.0 58.4 85.8 236.2	12.5 4.86 5.95 5.78 4.00 5.21 3.19 2.40	5.46
1080.1	1.25	168.0	134.0	134.0
1080.7	1.25	202.0	162.0	162.0
1081.6	1.0	166.0	166.0	166.0
HS 130*		•	•	0.33
			1000°C	
1078.2	65.0	62.9	0.97	0.97
1079.2	65.0	0.33	0.005	0.005
1080.2	26.0	0	0 ·	0

<sup>\*</sup>From data of Tripp and Graham

1.0

2.5

2.83

4.25

5.75

6.25

22.25

46.0

58.75

0.4

0.63

0.91

1.54

1.65

0.65

0.76

0.76

0.84

1.16

TABLE 4 WEIGHT GAIN OF SAMPLES HEATED IN AIR AT  $1400^{\circ}\mathrm{C}$ 

Sample #	1085.2 ΔW/A <sub>2</sub>	1085.3 ΔW/A	1086.1 ΔW/A	1086.2 ΔW/A	1088.1 ΔW/A	1088.2 ΔW/A	1089.1	1089.2	1090.1	1090.2
time (hrs)	(mg/cm <sup>2</sup> )		————			ΔW/ A	ΔW/A	ΔW/A 	ΔW/A 	ΔW/A
1	4.38	5.57	2.04	1.65	3.53	2.84	1.29	1.30	0.90	0.70
3	4.54	5.82	2.35	2.19	3.81	3.02	1.57	1.63	1.35	1.14
10.75	4.87	6.42	2.60	3.80	4.74	4.07	1.61	2.54	2.42	2.06
23.25	7.23	8.16	2.85	4.64	5.68	4.99	3.75	3.52	3.21	2.86
47.25	8.36	10.37	3.41	5.19	7.02	6.29	4.74	4.42	4.18	3.84
59.25	9.17	10.52	3.60	5.40	7.66	6.81	5.28	4.82	4.70	4.28
Sample # Time (hrs)	1084.1 W/A	1091. _W/A								

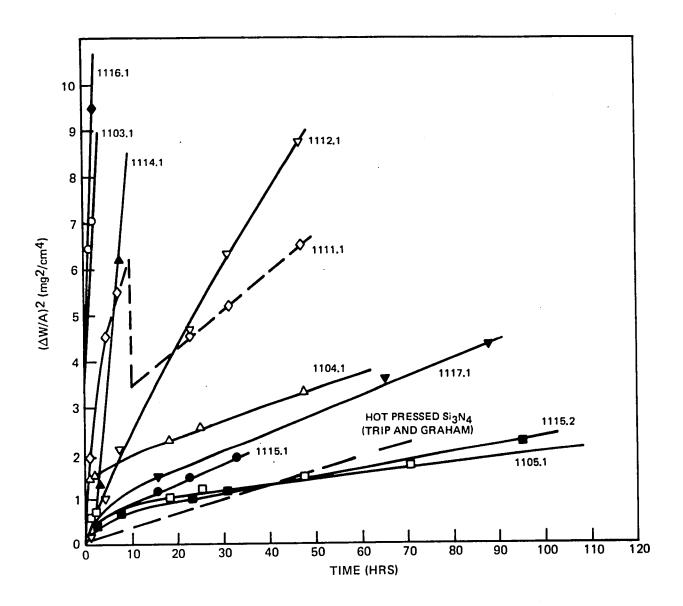


Fig. 5 Weight Gain in Air at 1400°C for Some Compositions in the YSiAION and YSiBeON Systems

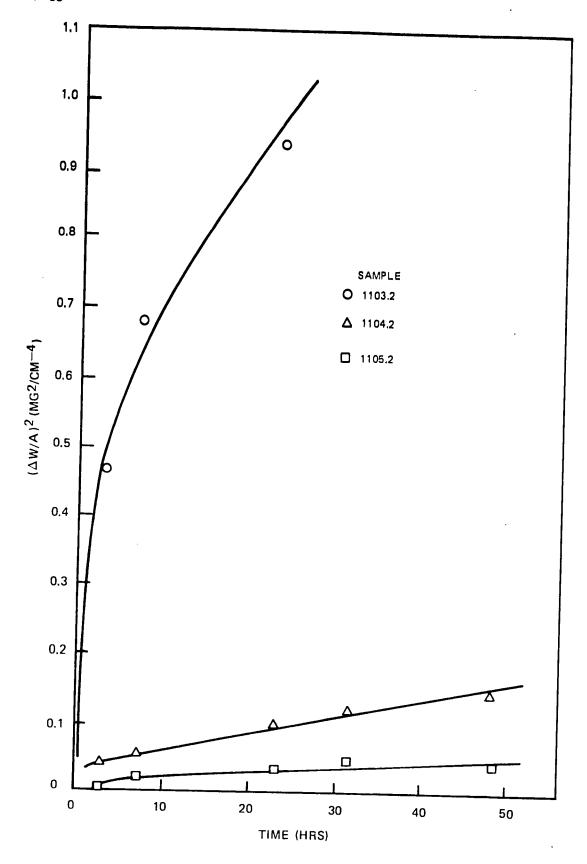


Fig. 6 Weight Gain in Air at 1300°C for Compositions in the System YSiALON (Parabolic Plot)

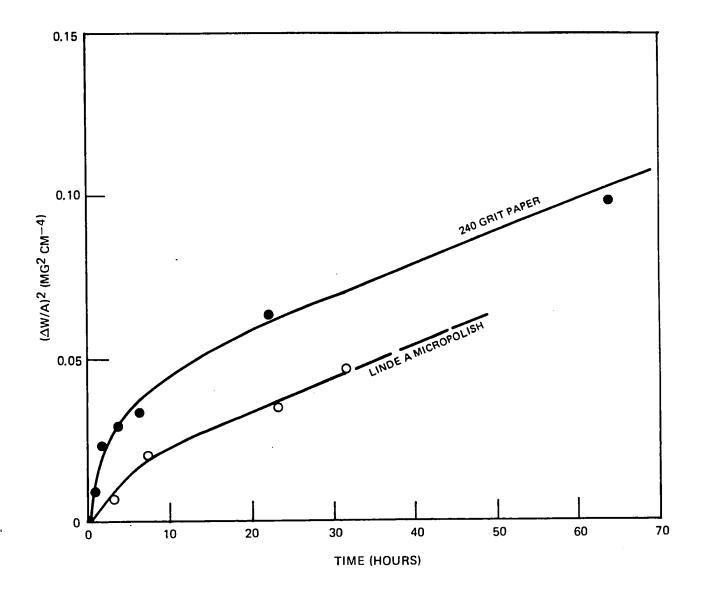


Fig. 7 Weight Gain in Air at 1300°C for 1105 Sample Having Different Surface Finishes

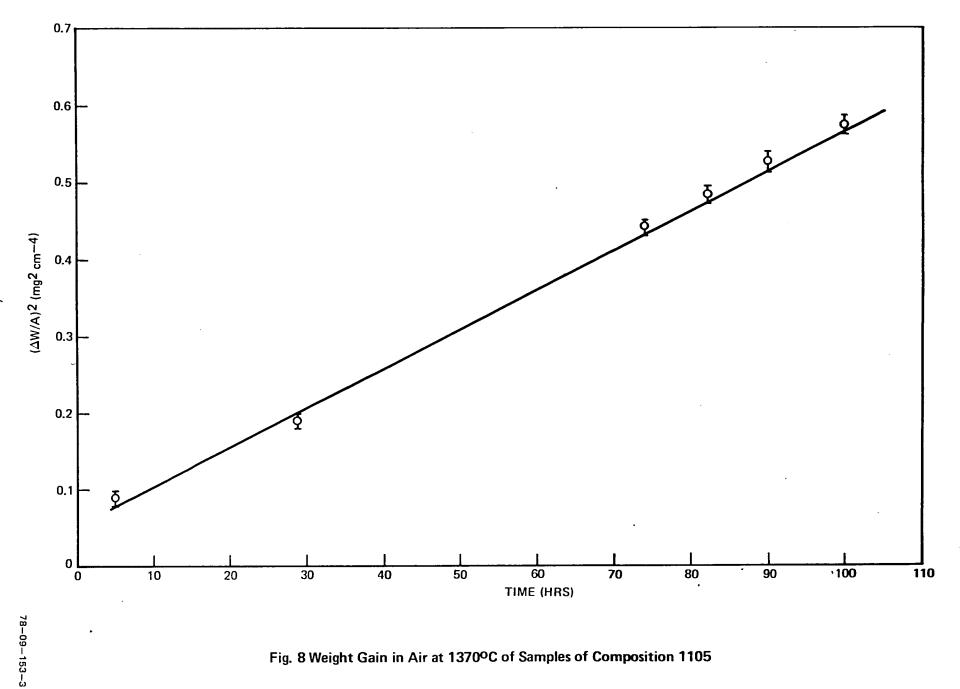


Fig. 8 Weight Gain in Air at 1370°C of Samples of Composition 1105

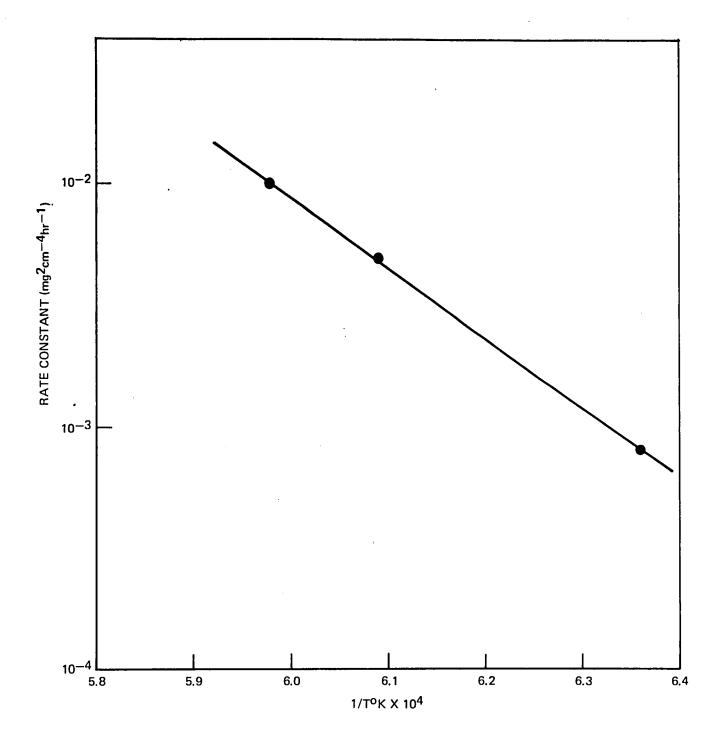


Fig. 9 Temperature Dependance of Oxidation Rate for Composition 1105 Samples

Parabolic Oxidation Rates in Air at 1400°C and Principle Components of Oxide Scales of Samples of Different Composition

TABLE 4

Composition	Oxidation Rate $\frac{\text{Constant}}{(\text{mg}^2\text{cm}^{-4}\text{hr}^{-1}) \times 10^2}$	Principle Components of Scale
1103	100	glass, cristobalite
1104	3.5	mullite
1105	1.0	mullite
1111	8.3	glass, cristobalite, Y <sub>2</sub> Si <sub>2</sub> O <sub>7</sub>
1112	15.6	glass, cristobalite Y <sub>2</sub> Si <sub>2</sub> O <sub>7</sub>
1113	4.0	glass, cristobalite
1114	100	glass, cristobalite
1115	1.6	mullite
1116	100	glass, cristobalite
1117	4.0	glass, cristobalite
HS130	2.7*	glass, cristobalite

<sup>\*</sup>Data of Tripp and Graham.

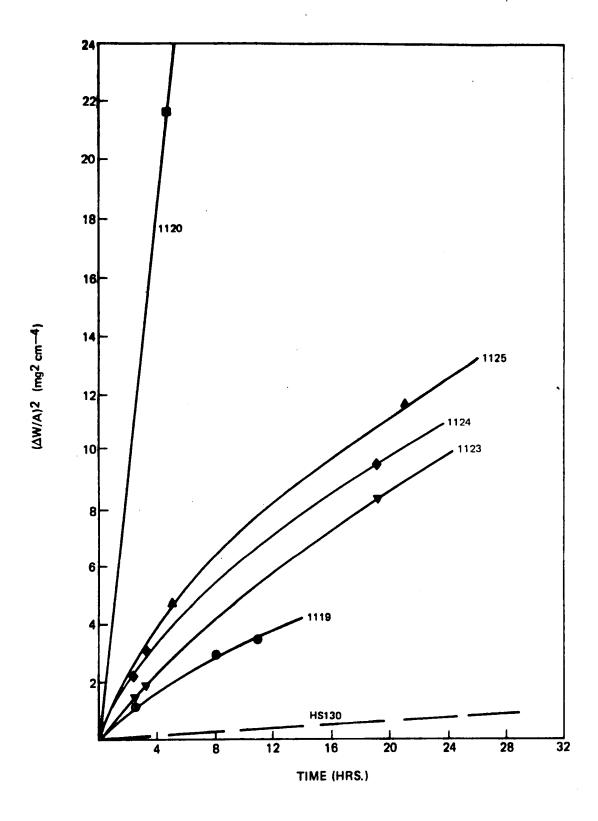


Fig. 3 1400°C Oxidation Curves for Some Compositions in the CeSiON, CeSiAION, and CeSiBeON Systems

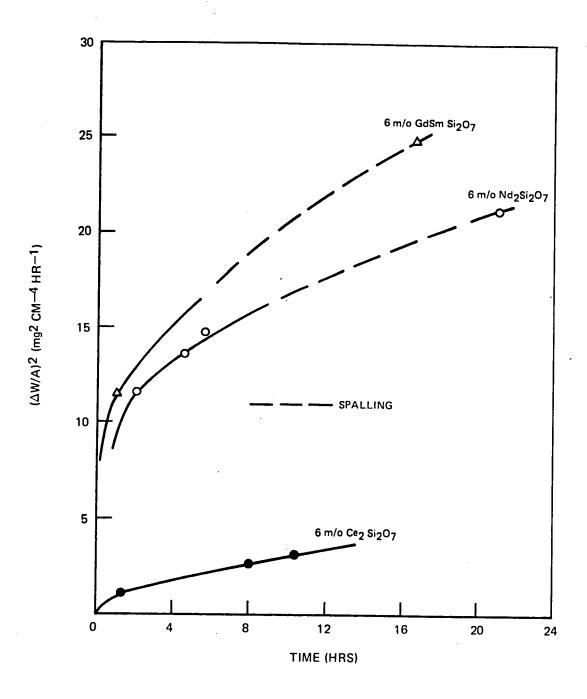


Fig. 4 1400°C Oxidation Curves for Samples 94 m/o Si<sub>3</sub>N<sub>4</sub> 6 m/o R<sub>2</sub> Si<sub>2</sub>O<sub>7</sub>

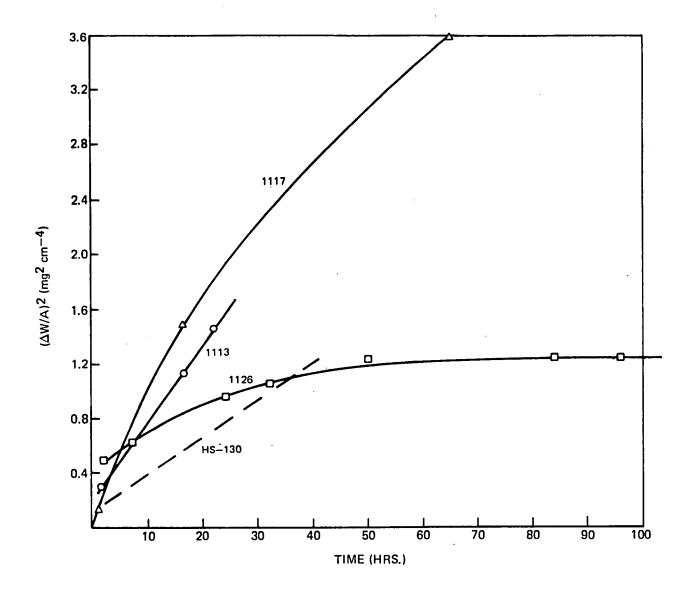


Fig. 5 1400°C Oxidation Curves for Some Compositions in the YSiBeON System

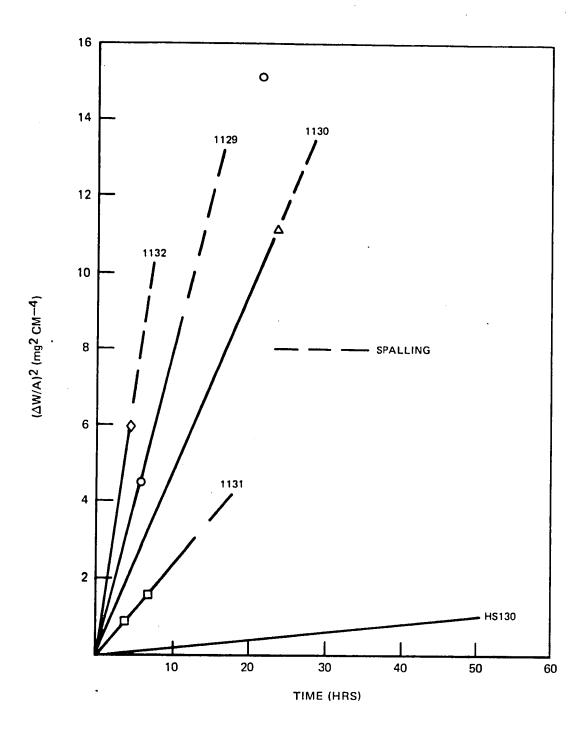


Fig. 6 1370°C Oxidation Curves for Some Compositions in the Y Ce Si on System

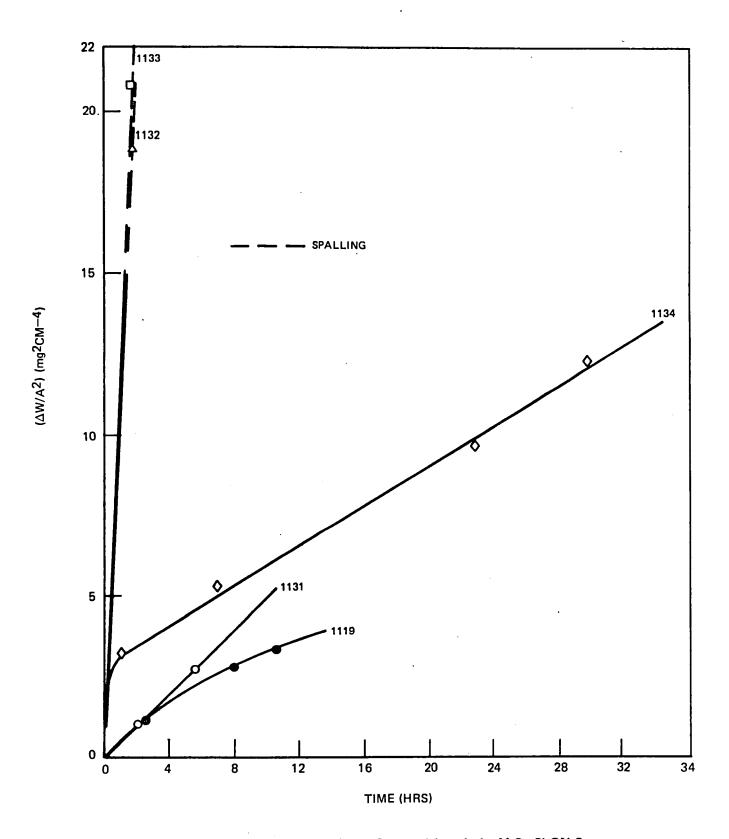


Fig. 7 1400°C Oxidation Curve for Some Compositions in its Y Ce Si ON System

# APPENDIX E - SOLID-LIQUID EQUILIBRIA IN THE Y-Si-O-N AND Nd-Si-O-N SYSTEMS

### A. <u>Y-Si-O-N System</u>

In order to obtain a better understanding of the interactions of the melts with  $\mathrm{Si}_3\mathrm{N}_4$ , the melting behavior of ternary compounds in the systems  $\mathrm{Si}_3\mathrm{N}_4$  -  $\mathrm{Si0}_2$  - $Y_2O_3$  and  $Si_3N_4-SiO_2-Nd_2O_3$  was investigated. This led to a systematic investigation of liquid-solid equilibria along the join  $Y_2SiO_5 - Y_2Si_3O_3N_4$  in the former system. In these investigations, compositions were prepared from raw materials in ten or twenty gram batches by grinding under acetone until dry, pressing into pellets, and firing to 1650° for two hours. The fired pellets were examined for evidence of melting, crushed to pass 200 mesh screen, and x-ray diffraction patterns obtained. Portions of the crushed powder were then repelletized and heated to high temperature in small boron nitride crucibles. When there was evidence of melting, the recrystallized material was cut into two pieces. One piece was examined metallographically and the other was crushed and x-rayed. Compositions investigated in this fashion, and the results, are presented in Table El. Micrographs of polished sections etched with HF of recrystallized samples in the  $\rm Y_2O_3$  -  $\rm SiO_2$  -  $\rm Si_3N_4$ system are shown in Figs. E2 through E10. From these data a tentative phase diagram for the joint  $ext{Y}_2 ext{Si0}_5 ext{-Y}_2 ext{Si}_3 ext{0}_3 ext{N}_4$  is constructed and presented in Fig. 29 in the main section of the report. Aspects of the proposed phase diagram are discussed in terms of the metallographic and x-ray data below. Composition points refer to Fig. El.

# 1. Composition Point 13 (5 m/o Y2Si303N4)

Metallography and x-ray results show this composition to be solid at  $1700^{\circ}$ C and a mixture of primary crystals of  $Y_2Si0_5$  ( $X_2$  phase) and liquid (which crystallized to H phase) at  $1750^{\circ}$ C.

# 2. Compositions Points 14-17 (10 through 30 m/o Y2SiO3N4)

Composition 14 showed no evidence of melting at  $1700^{\circ}\text{C}$ , but appeared to have been all liquid at  $1750^{\circ}\text{C}$ . The micrograph of the 10 m/o  $Y_2\text{Si}_30_3\text{N}_4$  composition cooled from  $1750^{\circ}\text{C}$  (Fig. E3) appears to be very nearly single phase with just a trace of what appears to be a eutectic microstructure at triple points between primary H phase cyrstals. Figure E4 shows large primary crystals of H phase and eutectic matrix indicating that composition 15 m/o  $Y_2\text{Si}_30_3\text{N}_4$  cooled through a two phase region. Not shown in Fig. E4 are occasional small pockets of material that, on the basis of its etching characteristics, appears to be residual glass. Figure E5 shows the 20 m/o  $Y_2\text{Si}_30_3\text{N}_4$  composition cooled from  $1750^{\circ}\text{C}$ , and again shows the presence of large primary H phase cyrstals, and, in areas of the sample, a eutectic like matrix. In other areas of the sample, however, the matrix phase appears quite different, as though residual glass has been etched to disclose

crystalline dendrites. Figure E6 shows small primary H phase crystals in a glassy matrix for the 30 m/o  $Y_2Si_3O_3N_4$  composition cooled from  $1700^{\circ}C$ . The matrix phase has taken on many colors ranging from purples and blues through yellows and brown. These colors are probably interference colors resulting from films which formed on the glass as a result of HF etch. The primary H phase crystals were the only phase detected by x-rays in this sample.

# 3. Composition Points 18 and 19 (50 and 75 m/o Y2Si3O3N4)

Composition point 18 was K phase at  $1650^{\circ}$ C, but crystallized to mostly  $Y_2Si_3O_3N_4$  and H phase when cooled from 1750, indicated either that K phase melts incongruently to  $Y_2Si_3O_3N_4$  and liquid, or perhaps that K phase melts to two liquids. As in the case of sample 1007B (20 m/o  $Y_2Si_3O_3N_4$ , Fig. E5), the matrix phase appeared to differ in different regions of the sample. There appears to be little contrast between primary crystals and matrix phases in one area, while the matrix phase shows strong interference colors in another region, as shown in Fig. E7. This sample was repolished and re-etched with the same result, and it seems resonably sure that the matrix phase did not crystallize in certain regions during cool down of the sample from  $1750^{\circ}$ C. Sample 19 (75 m/o  $Y_2Si_3O_3N_4$ ) cooled from  $1700^{\circ}$ C, shown in Fig. E8, contains both  $Y_2Si_3O_3N_4$  crystals which were growing, and small rounded crystals of K phase which were dissolving in a liquid which has cooled to a glass. Clearly then, the K phase melts incongruently at very nearly  $1700^{\circ}$ C.

# 4. <u>Composition Point 10 (J Phase)</u>

This composition point appeared single phase to x-rays and metallographic examination at all temperatures investigated. It is solid at  $1750^{\circ}$ C.

### B. Nd-Si-O-N System

The generally lower liquidus temperatures in the  $\mathrm{Nd}_2\mathrm{O}_3\mathrm{-SiO}_2$  system compared to those in the  $\mathrm{Y}_2\mathrm{O}_3\mathrm{-SiO}_2$  system suggested that ternary liquids in the  $\mathrm{Nd}_2\mathrm{O}_3$   $\mathrm{SiO}_2\mathrm{-Si}_3\mathrm{N}_4$  (or liquids involving mixtures of  $\mathrm{Nd}_2\mathrm{O}_3$  and  $\mathrm{Y}_2\mathrm{O}_3$ ) would be more reactive than the analogous  $\mathrm{Y}_2\mathrm{O}_3$  liquids. Only the composition points 16 and 18 (Fig. E1) were investigated in the neodynium system. Data for these points are presented in Table E1. E-ray data for the neodynium H and K phases are presented in Tables E2 and E3. Micrographs are shown in Figs. E10 and E11. As was the case with composition point 16 in the yttria system, so in the neodymia system, large areas of matrix liquid failed to crystallize on cooling from  $1750^{\circ}\mathrm{C}$  and remained as glass at room temperature. The amount of glass retained was considerably greater in the neodymia case than in that of yttria. Composition 18 (that of K phase) cooled from 1750 and  $1800^{\circ}\mathrm{C}$  contained no primary cyrstals, but exhibited dendritic structures. The x-ray pattern showed poorly crystallized H phase and  $\mathrm{Nd}_2\mathrm{Si}_3\mathrm{O}_3\mathrm{N}_4$  which is probably, but not necessarily, indicative of incongruency.

### APPENDIX E

### TABLE E1

### METALLOGRAPHIC AND X-RAY DATA FOR TERNARY COMPOSITIONS

Firing

	Re	itch Comp	osition (grams)	Composition	Condi	tions		To L.		- 0		
Sample	Si <sub>3</sub> N <sub>4</sub>			Point	Ţ °c	Time	Visual and/or Metallographic	Y-1	ises rav	3 U Di	DS6	erved by caction(1)
Number	3 4	Si0 <sub>2</sub>	Y203 Other	(Fig. 4)	<del>+</del> 30	(hrs)	Observation					<del></del>
30071								<u>β</u>	<u>H</u>	<u>J</u>	<u>K</u>	M Other
1007A	1.858	3.184	14.956	16	1650	2	Solid					
1007B	1.858	3.184	14.956	16	1750	2	Melted		s s		w	
1007C	1.858	3.184	14.956	16	1850	0.5	Melted				.w	- W G10
1008	1. 201	- 011								m	ы	s Y <sub>2</sub> SiO <sub>5</sub>
1008A	4.304	1.844	13.852	18	1650	2	Solid			w	s	
	4.304	1.844	13.852	18	1850	0.5	Melted			w	3	m,
1048	4.304	1.844	13.852	18	1750	0.5	Melted			w		s
1024	0 510	00								*		5
1024	2.542	1.089	16.369	11	1650	2.5	Solid			s		
1039	2.542	1.089	16.369	11	1850	0.5	Melted			s		
1005	2.542	1.089	16.369	11	. 1750	0.5	Solid			s		
1029	7.664									-		
1040			12.336	12	1650	2.5	Solid	w		s		m YSi
1046	7.664		12.336	12	1850	0.5	Melted, partially reduced			~		
1046	7.664		12.336	12	1650	2.5	Solid	m		m		s
1000	7.664		12.336	12	1750	0.5	Solid		•			S
10714	1 050						·					3
1071A	1.052	2.550	11.29	15	1750	0.5	Melted		s			
									•		•	
1072A	0.701	2.705	11.29	14	1750	0.5	Melted					
1072B	0.701	2.705	11.29	14	1650	1.0	Solid					
1072C	0.701	2.705	11.29	14	1700	1.0	Solid		_			
					·		20114		5			
1073A	0.351	2.855	11.29	13	1750	0.5	Partial Melting					V 010
							- m. viai meiving	V	,			w Y <sub>2</sub> SiO <sub>5</sub>
1074A	1.402	2.404	11.29	17	1750	1.0	Melted	9	3			Weak undecipherable
1074B	1.402	2.404	11.29	17	1700	1.0	Partial Melting	s				wom minorphotony
_							•					
1078A	5.258	0.751	11.290	19	1700	1.0	Partial Melting	I	a	1	<u>m</u> .	
1078в	5.258	0.751	11.290	19	1750	0.5	Partial Melting	v				s
1000	- (-(	- (00	_									
1033 1041	1.606	0.688	Nd <sub>2</sub> 0 <sub>3</sub> 7.706	18	1650	2.5	Solid	ъ	,(2)	J.		s(3)
1041	1.606	0.688	_1033	18	1850	0.5	Melted, dendites	17	ī			w m YSi
1041	1.606	0.688	1033	18	1750	0.5	Melted, dendites	r				w
1056	7 1.00	0 101			_			_				
	1.403	2.404	Nd <sub>2</sub> 0 <sub>3</sub> 16.825		1650	2.0	Partial Melting	s		1	a T	
1057	1.403	2.404	1056	16	1750	0.5	Melted	S		T.		

Letters s, m, w indicate: Strong, medium and weak intensity pattern. the letters  $\beta$ , H, J, K and M are:  $\beta$ 'Si $_3N_{\dot{l}_1}$ , N - apatite, N - YAM, N -  $\alpha$  wollastonite, M - N - melilite

<sup>(2)</sup> the x-ray diffraction pattern for the neodymium H phase is given in Table 4.

<sup>(3)</sup> the x-ray diffraction pattern for the neodymium K phase is given in Table 5.

TABLE E2

X-RAY DIFFRACTION PATTERN FOR
NEODYMIUM H PHASE

a(A)	<u> </u>
l. 03	10
4.21 4.04	8
3.65	. 8
3.30	20
3.17	20
2.90	100
2.80	20
2.32	5
2.19	5
2.09	8
2.00	20
1.940	10
1.901	20
1.847	8
1.816	20
1.799	20
1.783	10

TABLE E3

X-RAY DIFFRACTION PATTERN FOR NEODYMIUM K PHASE

d(A)	I/I <sub>o</sub>
4.72	15
3.66	30
3.40	5
3.20	10
2.88	100
2.36	10
2.097	20
1.975	20
1.913	20

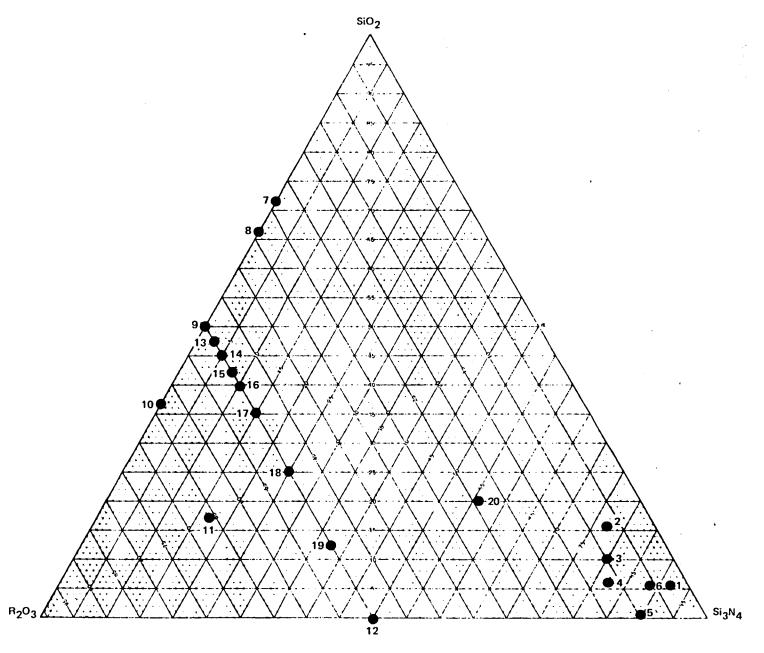


Figure E1 Compositions Investigated

### TOP OF SAMPLE

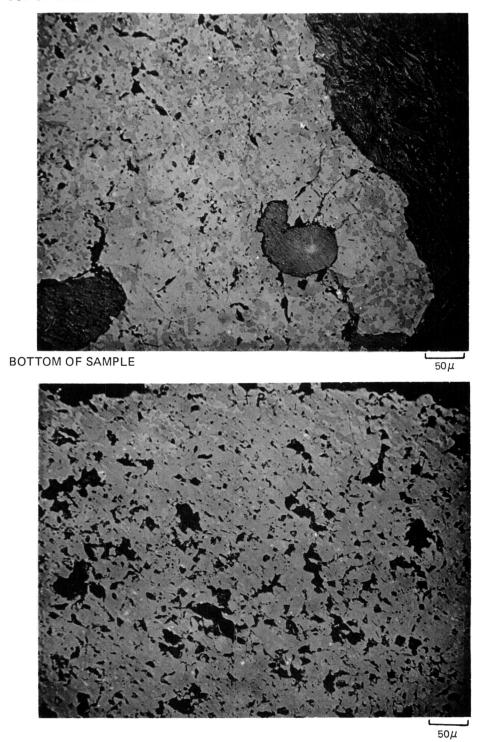


Figure E2 Etched Sections of Sample 1073A (5 m/o  $Y_2Si_3O_3N_4$  Cooled From 1750°C)

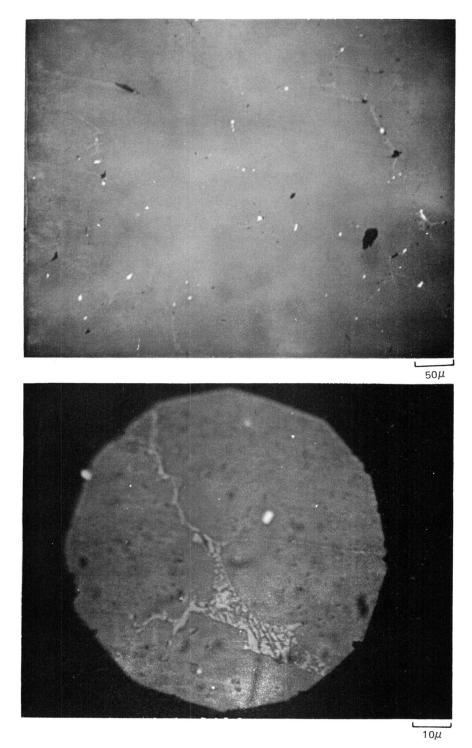


Figure E3 Etched Sections of Sample 1072A (10 m/o  $Y_2Si_3O_3N_4$  Cooled From 1750°C

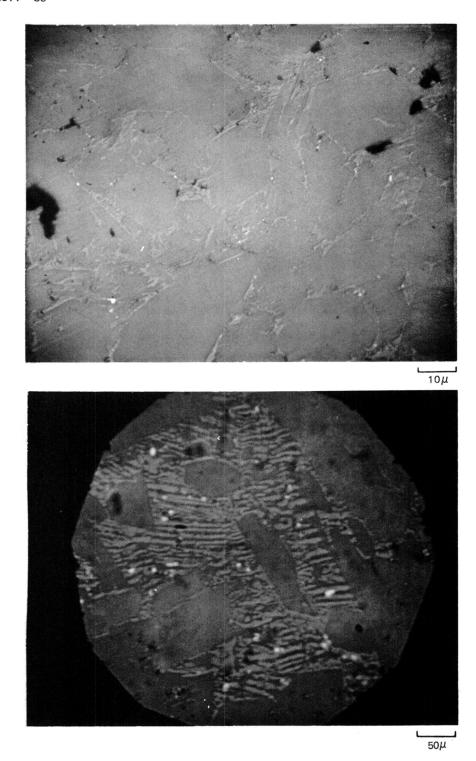


Figure E4 Etched Sections of Sample 1071 ( 15 m/o  $\,\mathrm{Y_2Si_3O_3N_4}$  Cooled From 1750°C )

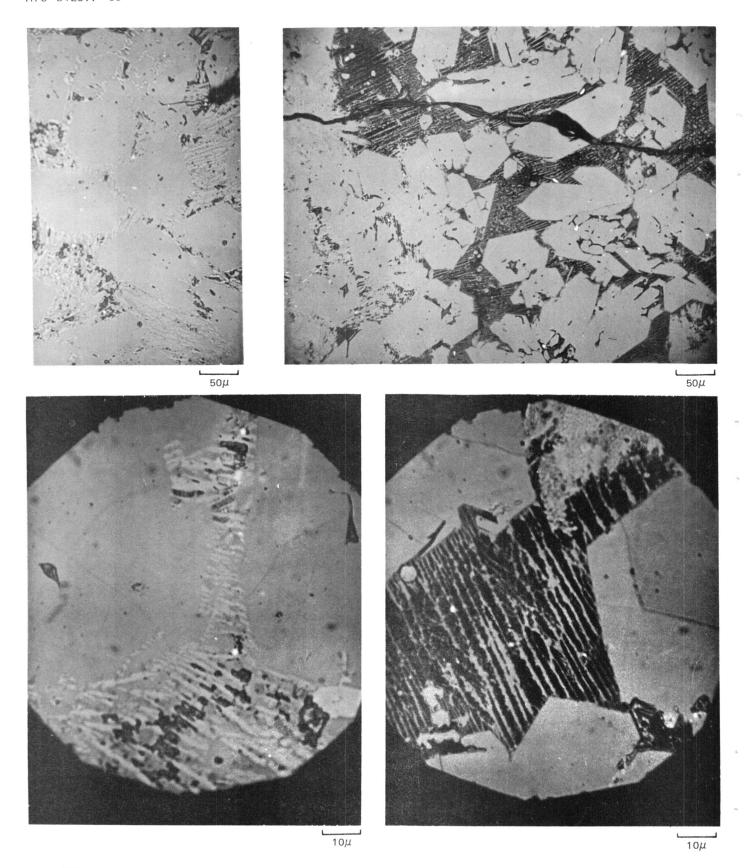


Figure E5 Etched Sections of Sample 1007B (20 m/o Y<sub>2</sub>Si<sub>3</sub>O<sub>3</sub>N<sub>4</sub> Cooled From 1750°C)

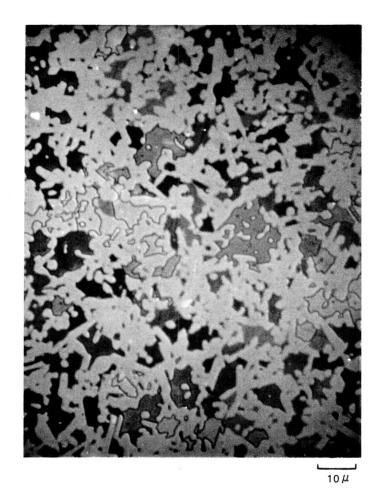
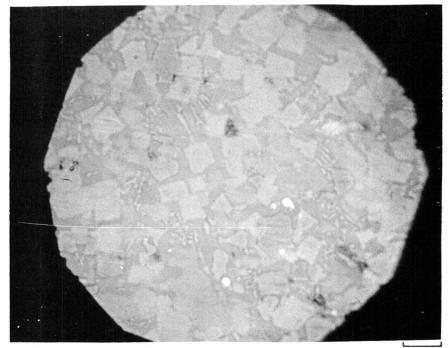


Figure E6 Etched Section of Sample 1074 B ( 30 w/o Y2Si3O3N4 Cooled From 1700°C

### A. LOW CONTRAST REGION (ALL CRYSTALLINE 2)



B. (HIGH CONTRAST REGION (GLASSY MATRIX 2)

10μ

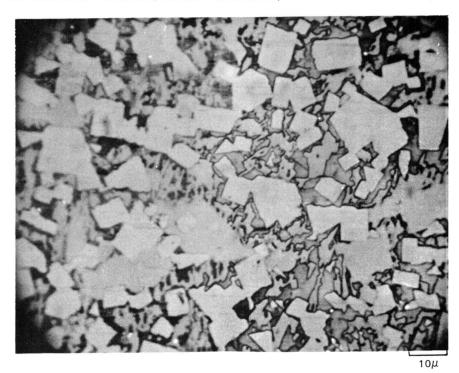


Figure E7 Etched Sections of Sample 1048 (  $50~\text{m/o}~\text{Y}_2\text{Si}_3\text{O}_3\text{N}_4$  ) Cooled From  $1750^o\text{C}$ 

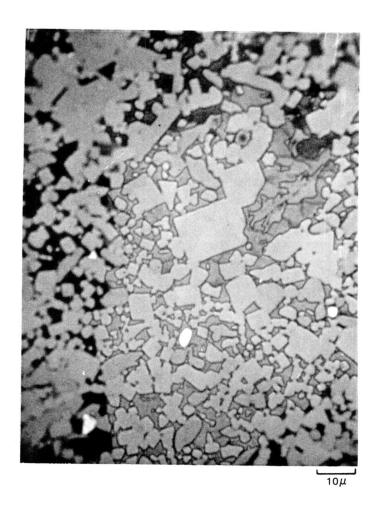


Figure E8 Etched Section of Sample 1078 A ( 75 w/o  $\,\mathrm{Y}_2\mathrm{Si}_3\mathrm{O}_3\mathrm{N}_4\,\mathrm{Cooled}$  From 1700°C

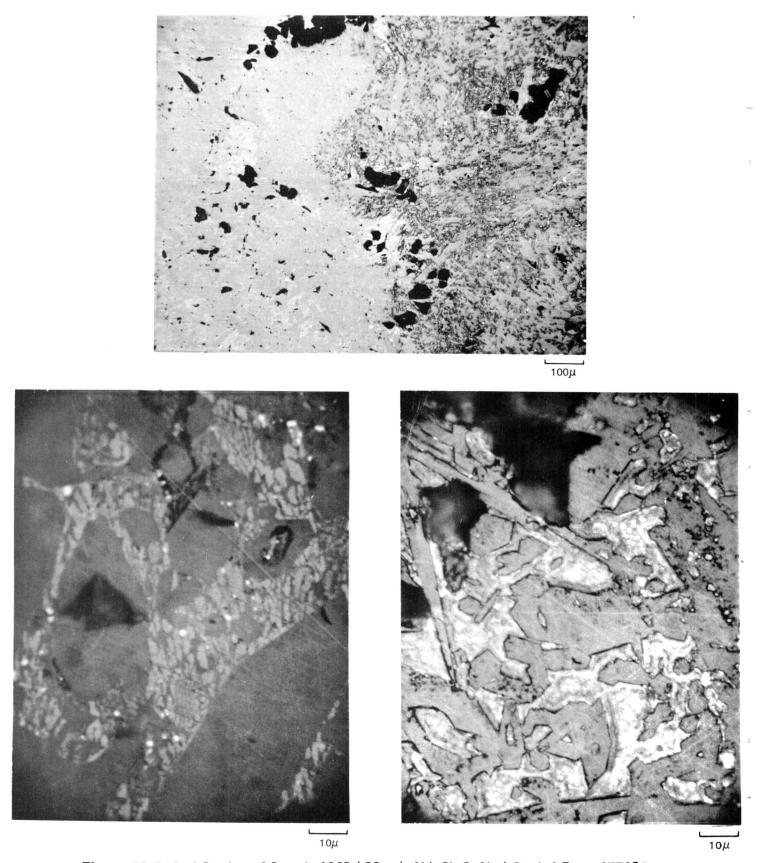
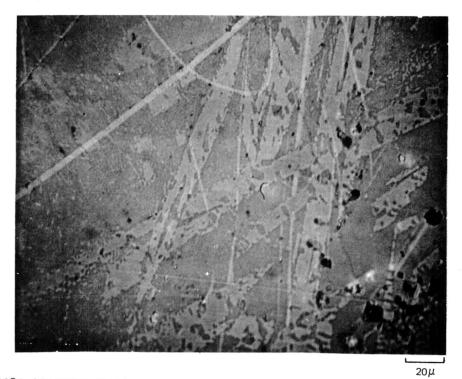


Figure E9 Etched Section of Sample 1068 ( 20 m/o  $Nd_2Si_3O_3N_4$  ) Cooled From 1750°C

# A. 1750°C (SAMPLE 1041)



B. 1800°C (SAMPLE 1047)



Figure E10 Etched Section of NdSiO<sub>2</sub>N Cooled From Different Temperatures

			2
			, a
			9

•			
· •			
· ·			
•			
: = -			
	•		
l .			
•			
•			
:			
,			
1 .			
· [			
ا سن			
. •			
· · · · · · · · · · · · · · · · · · ·			
. • 			

			•
			<b>∿</b> .
			•
			**